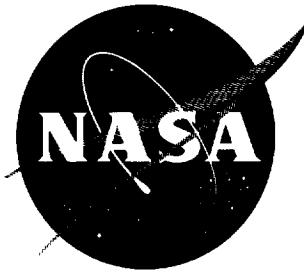


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TECHNICAL NOTE

D-1394

A TABULATION OF WIND-TUNNEL PRESSURE DATA AND SECTION
AERODYNAMIC CHARACTERISTICS AT MACH NUMBERS OF 1.61
AND 2.01 FOR TWO TRAPEZOIDAL AND THREE DELTA WINGS
HAVING DIFFERENT SURFACE SHAPES

By Emma Jean Landrum

Langley Research Center
Langley Station, Hampton, Va.

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
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SUMMARY

The pressure, section normal-force, and section pitching-moment coefficients for two wings, one flat and one warped, having the same trapezoidal planform and three wings, one flat, one cambered, and one cambered and twisted, having the same delta planform are tabulated. The trapezoidal wings had a 3-percent circular-arc thickness distribution, a taper ratio of 0.721, an aspect ratio of 1.342, and a semispan of 9.815 inches. The delta wings had an NACA 65A003 thickness distribution, an aspect ratio of 1.456, 70° of sweepback of the leading edge, and a semispan of 8.735 inches. The wings were tested at Mach numbers of 1.61 and 2.01 with fixed transition at Reynolds numbers of 3.6×10^6 and 3.1×10^6 , respectively. Angle-of-attack range was from -20° to 20°.

INTRODUCTION

Of current interest to the designer of efficient wings for supersonic aircraft is the prediction of the changes in aerodynamic characteristics of wings when they distort under variable flight loads. As part of a general investigation at low supersonic speeds of the effects of arbitrary camber and twist, two wings having the same trapezoidal planform and three wings having the same delta planform were tested at Mach numbers of 1.61 and 2.01. The purpose of this report is to present a tabulation of the pressure, section normal-force, and section pitching-moment coefficients obtained for these wings.

SYMBOLS

$b/2$	semispan
C_p	pressure coefficient
c	wing chord
c_n	section normal-force coefficient
c_m	section pitching-moment coefficient (taken about the midchord of the wing mean aerodynamic chord)
M	Mach number
m	cotangent of the leading-edge sweep angle
x, y, z	cartesian coordinate system with origin at apex of wing root
α	angle of attack of wing root, deg

Subscripts:

i	running index locating spanwise position
j	running index locating chordwise position

MODELS AND MODEL MOUNTING

Trapezoidal Wings

Two semispan trapezoidal wings, one flat and one warped, with the same planform were tested. Each wing had a 3-percent circular-arc thickness distribution, a taper ratio of 0.721, and aspect ratio of 1.342, and a semispan of 9.815 inches. A plan view of the models is shown in figure 1(a).

The ordinates of the warped wing were obtained by adding the mean line to the basic thickness distribution (ref. 1). The equation of the mean thickness plane is

$$\left(\frac{z}{c}\right)_{i,j} = 0.033454 \sin\left(\frac{3\pi}{2} \frac{y_i}{b/2}\right) \cos\left[\pi \left(\frac{x}{c_i}\right)_j\right]$$

where

$$0 \leq \frac{y_i}{b/2} \leq 1; \quad 0 \leq \left(\frac{x}{c_i}\right)_j \leq 1$$

For this warped wing, the maximum angle of the local slope relative to the chord of the unwarped wing is 6° . Figure 2(a) is a sketch of the mean thickness plane.

Each wing had seven streamwise rows of orifices located at 0.10, 0.35, 0.55, 0.67, 0.77, 0.87, and 0.97 semispan. (See fig. 1(a).) Chordwise location of the orifices was at 0.025, 0.05, 0.075, 0.10, 0.15, 0.20, 0.25, 0.30, 0.40, 0.50, 0.60, 0.70, 0.80, 0.90, and 0.95 local chord at each streamwise station. Orifices were located on both surfaces for the warped wing but only on one surface for the flat wing.

Delta Wings

Three semispan delta wings having the same planform were tested. One was flat, one was cambered, and one was cambered and twisted. Each wing had an NACA 65A003 thickness distribution, 70° of sweepback of the leading edge, an aspect ratio of 1.456, and a semispan of 8.735 inches. A plan view of the models is given in figure 1(b).

For the cambered wing, the equation of the mean thickness plane is

$$\left(\frac{z}{c}\right)_{i,j} = \frac{-2.18958 \times 10^{-3}}{c_i} \left[\frac{y_i}{m} + c_i \left(\frac{x}{c_i}\right)_j \right]^2$$

A sketch of this mean thickness plane is shown in figure 2(b). The maximum angle of the local slope relative to the chord of the uncambered wing is -6° for this wing.

The equation of the mean thickness plane for the cambered and twisted delta wing is

$$\left(\frac{z}{c}\right)_{i,j} = -3.35183 \times 10^{-4} \frac{y_i}{c_i} \left[\frac{\bar{y}_i}{m} + c_i \left(\frac{x}{c_i}\right)_j \right]^2$$

Figure 2(c) is a drawing of this mean thickness plane. Since the maximum twist and camber will be in the region of small wing chord, the maximum angle of the local slope relative to the chord of the uncambered wing is increased to -8° to insure reasonable incremental pressures and loadings.

Each wing had seven streamwise rows of orifices located at 0.10, 0.25, 0.40, 0.50, 0.60, 0.70, and 0.80 semispan. (See fig. 1(b).) The chordwise locations of the orifices on each spanwise station are given in table I in terms of a nominal location.

Model Mounting

The semispan wings were mounted horizontally in the tunnel from a turntable in a boundary-layer bypass plate which was located vertically in the test section about 10 inches from the tunnel wall. A photograph showing the cambered and twisted delta wing mounted in the tunnel at a positive angle of attack is presented in figure 3.

TESTS AND TEST PROCEDURES

The tests were conducted in the Langley 4- by 4-foot supersonic pressure tunnel at Mach numbers of 1.61 and 2.01. Transition was fixed about 1/2 inch from the wing leading edge by No. 60 carborundum grains.

Angle of attack was changed manually by rotating the turntable on which the models were mounted and was measured by a vernier scale outside the tunnel. The angle-of-attack range was from -20° to 20° . Tunnel stagnation pressure was 15 pounds per square inch absolute, corresponding to Reynolds numbers of 3.6×10^6 and 3.1×10^6 at Mach numbers 1.61 and 2.01, respectively.

PRESENTATION OF RESULTS

The pressure coefficients are presented in tables II to VI and the section normal-force and section pitching-moment coefficients are given in tables VII to XI. Tables II, III, VII, and VIII contain the data for the trapezoidal wings. Data for the delta wings are contained in tables IV to VI and IX to XI. For the delta wings, the chordwise orifice locations listed in tables IV to VI are the nominal locations corresponding to the actual locations listed in table I. No analysis of the data is made.

Langley Research Center,
National Aeronautics and Space Administration,
Langley Station, Hampton, Va., June 15, 1962.

REFERENCE

1. Abbott, Ira H., Von Doenhoff, Albert E., and Stivers, Louis S., Jr.:
Summary of Airfoil Data. NACA Rep. 824, 1945. (Supersedes NACA
WR L-560.)

TABLE I
CHORDWISE ORIFICE LOCATIONS FOR DELTA WINGS

(a) Flat wing

x/c, nominal	x/c at $y/b/2$ of:						
	0.10	0.25	0.40	0.50	0.60	0.70	0.80
0.0125	0.0125	0.0125	0.0152				
.0250	.0250	.0250	.0263	0.0250	0.0250	0.0250	
.0500	.0500	.0500	.0500	.0500	.0500	.0500	0.0500
.0750	.0750	.0750	.0750	.0750	.0750	.0750	
.1000	.1000	.1000	.1000	.1000	.1000	.1000	.1000
.1500	.1500	.1500	.1500	.1500	.1500	.1500	.1500
.2000	.2000	.2000	.2000	.2000	.2000	.2000	.2000
.2500	.2500	.2500	.2441	.2479	.2489	.2500	.2500
.3000	.3000	.3000	.2975	.2975	.3000	.3000	.3000
.4000	.4000	.3991	.4000	.4012	.3968	.3961	.3927
.5000	.4988	.5006	.5000	.5006	.5032	.5020	.5047
.6000	.5976	.6020	.5833	.6000	.6000	.6000	.6000
.7000	.7000	.7000	.7000	.7000	.7000	.7000	.7000
.7500							.7500
.8000	.8000	.8000	.8000	.8000	.8000	.8000	
.8500							.8500
.9000	.9000	.9000	.9000	.9000	.9000	.9000	

TABLE I.- Continued
CHORDWISE ORIFICE LOCATIONS FOR DELTA WINGS

(b) Cambered wing

x/c , nominal	x/c at $y/b/2$ of:									
	0.10	0.25	0.40	0.50	0.60	0.70	0.80	Upper	Lower	Upper
0.0125	0.0111	0.0111	0.0130	0.0151	0.0180					
.0250	.0236	.0290	.0255	.0267	.0320	0.0266	0.0160	0.0312	0.0320	0.0347
.0500	.0488	.0540	.0505	.0510	.0560	.0516	.0450	.0562	.0560	.0527
.0750	.0736	.0790	.0755	.0770	.0763	.0800	.0766	.0710	.0820	.0777
.1000	.0988	.1040	.1005	.1020	.1013	.1050	.1012	.1080	.1062	.1070
.1500	.1488	.1530	.1505	.1520	.1513	.1520	.1516	.1480	.1562	.1570
.2000	.2020	.2040	.2008	.2020	.2013	.2060	.2060	.1991	.2062	.2050
.2500	.2483	.2540	.2505	.2520	.2513	.2460	.2479	.2440	.2510	.2410
.3000	.2986	.3040	.3005	.3020	.2972	.3010	.2972	.2920	.3050	.2937
.4000	.3988	.4030	.3958	.3940	.3961	.3940	.3891	.3940	.3920	.3847
.5000	.4935	.5030	.4966	.5010	.4986	.4990	.4962	.4980	.5000	.4980
.6000	.5951	.5990	.6008	.5990	.6013	.5960	.6016	.6060	.6062	.6027
.7000	.7402	.7020	.7005	.7020	.7013	.7050	.7020	.7060	.7062	.7027
.7500										
.8000	.7986	.8040	.8005	.8020	.8010	.8050	.8016	.8060	.8060	.8027
.8500										
.9000	.8988	.9050	.9008	.9020	.9013	.9030	.9012	.9060	.9080	.8534
										.8960

TABLE I.—Concluded

(c) Cambered and twisted wing

TABLE II
PRESSURE COEFFICIENTS FOR FLAT TRAPEZOIDAL WING

(a) $M = 1.61$

x/c	Cp at $y/\frac{c}{2}$ of:														x/c	
	.10		.35		.55		.67		.77		.87		.97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 0$																
.0125															.0125	
.025	.081		.106		.120		.117		.112		.112		.105		.025	
.050	.081		.107		.108		.120		.109		.106		.089		.050	
.075	.069		.094		.095		.100		.095		.091		.078		.075	
.100	.064		.087		.091		.096		.098		.097		.071		.100	
.150	.053		.064		.083		.086		.083		.082		.056		.150	
.200	.048		.053		.056		.076		.071		.048		.057		.200	
.250	.056		.042		.036		.062		.059		.043		.038		.250	
.300	.034		.027		.032		.042		.038		.016		.031		.350	
.350	.018		.014		.023		.014		.009		.004		.009		.400	
.400															.450	
.450															.500	
.500	.012		.001		-.003		-.011		-.029		-.015		.008		.550	
.550															.600	
.600	-.010		-.008		-.027		-.036		-.027		-.019		-.010		.650	
.650															.700	
.700	-.035		-.036		-.040		-.054		-.057		-.055		-.049		.750	
.750															.800	
.800	-.042		-.060		-.074		-.061		-.055		-.052		-.033		.850	
.850															.900	
.900	-.063		-.078		-.078		-.074		-.062		-.045		-.038		.950	
.950	-.067		-.079		-.079											
$\alpha = 2$																
.0125	.028	.140	.039	.182	.054	.198	.046	.194	.041	.187	.037	.189	.035	.178	.025	
.025	.029	.133	.036	.181	.040	.187	.046	.191	.041	.183	.033	.181	.026	.154	.050	
.075	.021	.114	.028	.161	.023	.169	.033	.174	.026	.169	.020	.163	.028	.122	.075	
.100	.018	.109	.022	.147	.025	.162	.025	.165	.028	.167	.026	.169	.033	.102	.100	
.150	.009	.096	.007	.112	.015	.146	.016	.152	.013	.152	.012	.144	.024	.173	.150	
.200	.003	.083	.001	.100	-.002	.102	.009	.136	.002	.136	.001	.089	.020	.080	.200	
.250	.016	.113	-.004	.085	-.015	.088	-.004	.112	-.005	.144	-.004	.085	.013	.056	.300	
.300	-.027	.090	-.019	.072	-.021	.082	-.014	.092	-.018	.083	-.024	.044	.006	.046	.350	
.350	-.024	.060	-.034	.069	-.029	.070	-.034	.053	-.029	.037	-.033	.017	-.016	.011	.400	
.400															.450	
.450															.500	
.500	-.021	.077	-.048	.043	-.042	.042	-.049	.020	-.057	-.007	-.039	-.004	-.013	.003	.550	
.550															.600	
.600	-.071	.033	-.055	.036	-.067	.008	-.061	-.008	-.052	-.009	-.042	-.007	-.029	-.017	.650	
.650															.700	
.700	-.064	-.008	-.079	-.006	-.071	-.013	-.081	-.032	-.067	-.032	-.053	-.028	-.047	-.027	.750	
.750															.800	
.800	-.075	-.012	-.096	-.035	-.079	-.037	-.078	-.039	-.079	-.043	-.069	-.042	-.056	-.037	.850	
.850															.900	
.900	-.099	-.035	-.100	-.059	-.099	-.061	-.082	-.053	-.076	-.052	-.079	-.049	-.069	-.041	.950	
.950	-.099	-.037	-.100	-.063	-.101	-.069	-.090	-.067	-.082	-.072	-.072	-.064	-.064	-.046	.950	
$\alpha = 4$																
.0125	.041	.201	-.031	.275	-.017	.291	-.021	.290	-.027	.279	-.032	.284	-.035	.282	.025	
.025	-.030	.190	-.034	.264	-.031	.279	-.025	.277	-.026	.277	-.032	.272	-.025	.235	.050	
.075	.029	.171	-.041	.229	-.041	.260	-.031	.266	-.038	.256	-.044	.259	-.011	.185	.075	
.100	.028	.161	-.046	.213	-.048	.240	-.039	.251	-.039	.257	-.041	.262	-.000	.157	.100	
.150	.041	.147	-.059	.170	-.053	.192	-.046	.222	-.047	.227	-.053	.226	-.004	.115	.150	
.200	.046	.130	-.058	.145	-.066	.162	-.054	.197	-.058	.200	-.064	.143	-.009	.108	.200	
.250	.019	.169	-.050	.147	-.057	.147	-.067	.172	-.067	.165	-.041	.124	-.017	.076	.250	
.300	-.083	.143	-.075	.129	-.079	.137	-.075	.148	-.073	.128	-.060	.084	-.022	.051	.300	
.350															.350	
.400	-.073	.109	-.079	.117	-.080	.129	-.086	.100	-.065	.075	-.063	.041	-.048	.014	.400	
.450															.450	
.500	-.071	.125	-.102	.093	-.088	.083	-.082	.062	-.082	.028	-.068	.022	-.056	.001	.500	
.550															.550	
.600	-.117	.079	-.105	.083	-.106	.048	-.078	.021	-.078	.014	-.061	.014	-.076	-.014	.600	
.650															.650	
.700	-.105	.034	-.121	.035	-.102	.021	-.105	-.004	-.079	-.012	-.085	-.012	-.108	-.027	.700	
.750															.750	
.800	-.122	.036	-.132	-.002	-.102	-.013	-.105	-.016	-.108	-.024	-.096	-.026	-.140	-.041	.800	
.850															.850	
.900	-.145	.003	-.127	-.024	-.118	-.040	-.105	-.036	-.107	-.036	-.117	-.033	-.167	-.038	.900	
.950	-.140	-.003	-.130	-.031	-.123	-.048	-.110	-.051	-.109	-.052	-.117	-.032	-.182	-.044	.950	
$\alpha = 6$																
.0125	-.093	.262	-.088	.381	-.071	.417	-.076	.417	-.083	.406	-.086	.405	-.085	.406	.025	
.025	-.082	.249	-.091	.324	-.085	.358	-.082	.376	-.079	.391	-.085	.371	-.056	.316	.050	
.075	-.077	.229	-.095	.292	-.092	.337	-.089	.351	-.093	.348	-.096	.353	-.038	.247	.075	
.100	-.074	.214	-.103	.277	-.098	.311	-.093	.326	-.093	.339	-.090	.351	-.034	.202	.100	
.150	-.085	.203	-.113	.229	-.104	.274	-.098	.292	-.102	.297	-.101	.269	-.036	.150	.200	
.200	.091	.176	-.109	.213	-.114	.222	-.106	.261	-.109	.262	-.103	.184	-.046	.129	.200	
.250	.051	.222	-.104	.208	-.126	.205	-.118	.233	-.119	.214	-.106	.159	-.059	.085	.250	
.300	-.137	.200	-.115	.193	-.131	.199	-.124	.207	-.118	.172	-.092	.113	-.073	.058	.300	
.350															.350	
.400	-.115	.167	-.109	.176	-.128	.188	-.133	.152	-.099	.116	-.101	.068	-.103	.021	.400	
.450															.450	
.500	-.109	.171	-.143	.151	-.134	.125	-.114	.103	-.120	.067	-.105	.045	-.130	.011	.500	
.550															.550	
.600	-.153	.130	-.140	.131	-.132	.093	-.112	.056	-.115	.042	-.108	.038	-.171	-.006	.600	
.650															.650	
.700	-.136	.080	-.152	.075	-.133	.055	-.125	.031	-.118	.018	-.128	.002	-.210	-.027	.700	
.750															.750	
.800	-.157	.080	-.159	.033	-.130	.018	-.138	.006	-.130	.002	-.140	-.009	-.227	-.039	.800	
.850															.850	
.900	-.179	.037	-.147	.014	-.140	-.012	-.134	-.012	-.145	-.018	-.150	-.022	-.227	-.039	.900	
.950	-.173	.028	-.149	.001	-.148	-.018	-.138	-.025	-.145	-.033	-.154	-.025	-.225	-.04		

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR FLAT TRAPEZOIDAL WING

(a) $M = 1.61$ - Continued

$\frac{c}{\delta}$	Cp at $\frac{\sqrt{2}}{2} \alpha$ of :														$\frac{c}{\delta}$	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 6$																
.0125															.0125	
.025	-.137	.350	-.133	.507	-.114	.564	-.126	.562	-.134	.549	-.132	.562	-.135	.564	.025	
.050	-.126	.326	-.136	.492	-.131	.492	-.130	.489	-.130	.504	-.135	.479	-.100	.395	.050	
.075	-.112	.313	-.140	.481	-.138	.430	-.138	.455	-.140	.458	-.140	.467	-.081	.329	.075	
.100	-.108	.286	-.145	.358	-.144	.402	-.143	.420	-.142	.438	-.140	.450	-.079	.263	.100	
.150	-.118	.269	-.154	.307	-.145	.393	-.148	.373	-.151	.381	-.147	.319	-.085	.199	.150	
.200	-.127	.243	-.154	.300	-.163	.301	-.155	.339	-.158	.336	-.125	.240	-.101	.169	.200	
.250	-.074	.294	-.145	.286	-.169	.290	-.165	.312	-.167	.275	-.101	.212	-.116	.126	.250	
.300	-.192	.272	-.149	.269	-.175	.276	-.172	.279	-.162	.294	-.128	.165	-.130	.095	.300	
.350															.350	
.400	-.148	.242	-.138	.244	-.172	.252	-.176	.214	-.140	.175	-.141	.120	-.185	.056	.400	
.450															.450	
.500	-.140	.228	-.181	.222	-.172	.177	-.158	.152	-.158	.117	-.142	.092	-.214	.042	.500	
.550															.550	
.600	-.182	.192	-.172	.177	-.154	.145	-.149	.113	-.156	.091	-.155	.071	-.258	.014	.600	
.650															.650	
.700	-.162	.142	-.180	.126	-.165	.101	-.148	.079	-.153	.067	-.175	.043	-.282	-.007	.700	
.750															.750	
.800	-.185	.131	-.180	.080	-.158	.066	-.172	.049	-.155	.041	-.182	.024	-.283	-.021	.800	
.850															.850	
.900	-.210	.078	-.166	.061	-.164	.031	-.165	.028	-.179	.015	-.190	.002	-.273	-.031	.900	
.950	-.201	.066	-.167	.048	-.167	.022	-.170	.012	-.186	.001	-.195	-.004	-.270	-.044	.950	
$a = 10$																
.0125															.0125	
.025	-.189	.630	-.185	.574	-.167	.647	-.178	.664	-.184	.658	-.187	.675	-.191	.641	.025	
.050	-.175	.394	-.185	.502	-.176	.562	-.177	.587	-.176	.606	-.181	.594	-.147	.452	.050	
.075	-.161	.373	-.193	.458	-.190	.515	-.182	.527	-.185	.543	-.185	.549	-.118	.388	.075	
.100	-.150	.344	-.193	.432	-.192	.477	-.186	.495	-.187	.509	-.186	.512	-.130	.319	.100	
.150	-.159	.315	-.203	.371	-.197	.425	-.190	.440	-.194	.451	-.192	.436	-.138	.239	.150	
.200	-.165	.309	-.205	.368	-.212	.378	-.195	.411	-.201	.383	-.165	.421	-.154	.203	.200	
.250	-.108	.366	-.199	.354	-.217	.358	-.205	.376	-.209	.326	-.143	.359	-.177	.158	.250	
.300	-.237	.326	-.191	.336	-.217	.348	-.212	.335	-.202	.328	-.165	.211	-.194	.128	.300	
.350															.350	
.400	-.187	.300	-.176	.302	-.219	.297	-.218	.257	-.177	.225	-.179	.163	-.259	.087	.400	
.450															.450	
.500	-.180	.284	-.219	.274	-.211	.222	-.195	.190	-.192	.151	-.186	.128	-.293	.065	.500	
.550															.550	
.600	-.219	.245	-.212	.220	-.179	.191	-.181	.152	-.197	.133	-.188	.111	-.329	.030	.600	
.650															.650	
.700	-.195	.192	-.218	.168	-.192	.136	-.175	.116	-.196	.101	-.224	.076	-.339	.012	.700	
.750															.750	
.800	-.219	.168	-.211	.121	-.185	.101	-.192	.081	-.194	.072	-.235	.051	-.329	-.006	.800	
.850															.850	
.900	-.240	.109	-.196	.098	-.188	.066	-.198	.061	-.210	.045	-.244	.024	-.318	-.025	.900	
.950	-.234	.099	-.197	.084	-.192	.056	-.202	.038	-.220	.027	-.244	.021	-.311	-.038	.950	
$a = 12$																
.0125															.0125	
.025	-.228	.522	-.228	.656	-.221	.747	-.231	.769	-.240	.771	-.248	.779	-.250	.720	.025	
.050	-.216	.477	-.228	.594	-.221	.655	-.221	.645	-.222	.704	-.229	.684	-.198	.538	.050	
.075	-.200	.437	-.230	.536	-.232	.594	-.224	.616	-.222	.628	-.229	.621	-.150	.458	.075	
.100	-.185	.402	-.234	.504	-.232	.552	-.227	.563	-.226	.606	-.228	.585	-.148	.368	.100	
.150	-.193	.383	-.239	.457	-.242	.510	-.230	.543	-.232	.522	-.237	.494	-.168	.291	.150	
.200	-.177	.364	-.239	.442	-.253	.523	-.233	.489	-.238	.535	-.201	.593	-.209	.261	.200	
.250	-.127	.436	-.241	.437	-.253	.441	-.242	.447	-.244	.390	-.179	.319	-.234	.405	.250	
.300	-.268	.388	-.225	.402	-.254	.420	-.250	.400	-.237	.352	-.200	.267	-.263	.170	.300	
.350															.350	
.400	-.215	.369	-.206	.373	-.258	.351	-.256	.313	-.208	.277	-.211	.211	-.315	.122	.400	
.450															.450	
.500	-.207	.343	-.245	.324	-.249	.276	-.227	.247	-.225	.204	-.224	.167	-.352	.092	.500	
.550															.550	
.600	-.240	.299	-.240	.262	-.201	.242	-.223	.204	-.229	.181	-.231	.158	-.379	.064	.600	
.650															.650	
.700	-.218	.237	-.244	.214	-.214	.183	-.208	.160	-.241	.148	-.267	.116	-.387	.040	.700	
.750															.750	
.800	-.245	.203	-.230	.188	-.211	.144	-.216	.128	-.232	.113	-.285	.084	-.362	.016	.800	
.850															.850	
.900	-.263	.143	-.215	.135	-.212	.108	-.227	.103	-.236	.086	-.297	.059	-.347	-.007	.900	
.950	-.261	.133	-.213	.118	-.216	.098	-.231	.064	-.249	.064	-.293	.053	-.345	-.018	.950	
$a = 14$																
.0125															.0125	
.025	-.272	.591	-.280	.755	-.284	.845	-.292	.865	-.297	.874	-.311	.870	-.315	.790	.025	
.050	-.260	.566	-.271	.670	-.270	.743	-.268	.770	-.271	.795	-.284	.771	-.257	.614	.050	
.075	-.241	.517	-.272	.616	-.276	.681	-.266	.708	-.271	.715	-.275	.693	-.196	.519	.075	
.100	-.235	.461	-.275	.589	-.278	.661	-.269	.668	-.267	.684	-.269	.648	-.228	.428	.100	
.150	-.228	.460	-.280	.532	-.278	.591	-.268	.606	-.273	.571	-.273	.496	-.244	.341	.150	
.200	-.227	.488	-.280	.535	-.290	.563	-.268	.564	-.285	.497	-.240	.417	-.265	.297	.200	
.250	-.153	.509	-.285	.512	-.292	.514	-.278	.509	-.283	.450	-.219	.376	-.297	.249	.250	
.300	-.292	.463	-.273	.472	-.287	.477	-.285	.452	-.275	.400	-.235	.319	-.325	.212	.300	
.350															.350	
.400	-.244	.443	-.241	.444	-.291	.414	-.292	.371	-.244	.326	-.248	.255	-.369	.155	.400	
.450															.450	
.500	-.239	.403	-.274	.373	-.290	.329	-.260	.296	-.259	.252	-.262	.214	-.395	.129	.500	
.550															.550	
.600	-.267	.345	-.271	.311	-.230	.286	-.263	.250	-.267	.228	-.277	.187	-.417	.095	.6	

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR FLAT TRAPEZOIDAL WING
(a) M = 1.61 - Concluded

x/c	Cp at y/c of :														K/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 16$																
.0125	-.304	.671	-.327	.856	-.343	.942	-.348	.967	-.357	.970	-.370	.938	-.372	.850	.0125	
.050	-.293	.650	-.310	.769	-.316	.843	-.317	.870	-.321	.885	-.337	.844	-.313	.681	.050	
.075	-.283	.597	-.308	.715	-.313	.779	-.310	.805	-.316	.800	-.324	.769	-.324	.581	.075	
.100	-.263	.581	-.309	.684	-.311	.737	-.308	.759	-.309	.762	-.312	.730	-.315	.585	.100	
.150	-.251	.551	-.314	.640	-.314	.746	-.306	.684	-.310	.688	-.314	.559	-.298	.395	.150	
.200	-.251	.524	-.311	.620	-.321	.617	-.304	.630	-.314	.556	-.275	.471	-.321	.348	.200	
.250	-.176	.388	-.318	.583	-.325	.573	-.313	.586	-.317	.507	-.253	.425	-.353	.291	.250	
.300	-.304	.346	-.311	.537	-.319	.537	-.318	.510	-.311	.457	-.268	.372	-.378	.256	.300	
.350															.350	
.400	-.270	.522	-.271	.503	-.324	.470	-.326	.428	-.276	.385	-.285	.309	-.411	.197	.400	
.450															.450	
.500	-.268	.455	-.294	.422	-.331	.381	-.290	.351	-.290	.306	-.302	.262	-.429	.165	.500	
.550															.550	
.600	-.287	.391	-.298	.360	-.263	.332	-.297	.297	-.299	.277	-.312	.213	-.443	.129	.600	
.650															.650	
.700	-.268	.322	-.301	.295	-.256	.275	-.289	.244	-.316	.223	-.348	.188	-.443	.095	.700	
.750															.750	
.800	-.294	.287	-.280	.259	-.260	.226	-.274	.205	-.325	.187	-.370	.152	-.426	.061	.800	
.850															.850	
.900	-.313	.227	-.264	.218	-.263	.195	-.281	.179	-.316	.153	-.392	.121	-.414	.045	.900	
.950	-.313	.221	-.259	.197	-.266	.185	-.269	.160	-.321	.135	-.394	.117	-.413	.032	.950	
$\alpha = 18$																
.0125	-.341	.780	-.373	.958	-.396	1.035	-.405	1.049	-.410	1.043	-.424	1.023	-.428	.907	.0125	
.050	-.326	.761	-.349	.866	-.359	.934	-.364	.950	-.368	.955	-.388	.907	-.370	.740	.050	
.075	-.316	.695	-.341	.815	-.351	.867	-.351	.882	-.358	.870	-.372	.825	-.310	.638	.075	
.100	-.297	.680	-.340	.779	-.347	.821	-.346	.832	-.348	.828	-.363	.778	-.325	.541	.100	
.150	-.272	.659	-.341	.722	-.342	.757	-.341	.749	-.348	.694	-.356	.621	-.343	.447	.150	
.200	-.271	.665	-.338	.692	-.347	.685	-.338	.689	-.351	.618	-.318	.533	-.369	.400	.200	
.250	-.200	.657	-.342	.655	-.351	.638	-.342	.625	-.349	.566	-.293	.483	-.400	.344	.250	
.300	-.316	.631	-.342	.614	-.346	.600	-.348	.568	-.348	.521	-.310	.429	-.418	.306	.350	
.350															.350	
.400	-.286	.588	-.296	.572	-.351	.531	-.355	.494	-.306	.450	-.324	.373	-.437	.254	.400	
.450															.450	
.500	-.289	.512	-.309	.477	-.359	.440	-.318	.407	-.322	.363	-.344	.319	-.454	.215	.500	
.550															.550	
.600	-.306	.447	-.314	.426	-.294	.397	-.324	.356	-.333	.336	-.358	.279	-.465	.181	.600	
.650															.650	
.700	-.281	.390	-.318	.361	-.277	.341	-.326	.308	-.348	.290	-.390	.250	-.464	.145	.700	
.750															.750	
.800	-.313	.354	-.294	.322	-.280	.290	-.312	.264	-.362	.242	-.410	.206	-.452	.111	.800	
.850															.850	
.900	-.329	.291	-.276	.280	-.283	.256	-.308	.239	-.363	.215	-.425	.181	-.443	.096	.900	
.950	-.329	.286	-.270	.261	-.287	.240	-.315	.216	-.361	.189	-.429	.173	-.437	.077	.950	
$\alpha = 20$																
.0125															.0125	
.025	-.371	.892	-.420	1.053	-.442	1.113	-.449	1.120	-.453	1.109	-.463	1.084	-.464	.961	.025	
.050	-.355	.849	-.383	.961	-.405	1.016	-.407	1.027	-.410	1.024	-.426	.973	-.417	.800	.050	
.075	-.349	.805	-.379	.909	-.386	.952	-.391	.960	-.398	.938	-.406	.894	-.368	.703	.075	
.100	-.325	.785	-.373	.878	-.380	.907	-.385	.907	-.386	.895	-.398	.847	-.365	.604	.100	
.150	-.291	.763	-.376	.813	-.373	.840	-.376	.823	-.384	.768	-.386	.693	-.383	.512	.150	
.200	-.291	.761	-.368	.780	-.376	.766	-.369	.762	-.382	.693	-.358	.608	-.403	.463	.200	
.250	-.224	.753	-.369	.745	-.377	.716	-.373	.700	-.378	.639	-.327	.557	-.431	.409	.250	
.300	-.331	.725	-.372	.696	-.377	.674	-.377	.644	-.379	.594	-.339	.505	-.447	.370	.350	
.350															.350	
.400	-.309	.665	-.321	.636	-.380	.594	-.383	.557	-.336	.510	-.356	.435	-.464	.309	.400	
.450															.450	
.500	-.311	.561	-.331	.527	-.386	.490	-.346	.465	-.349	.418	-.376	.378	-.472	.268	.500	
.550															.550	
.600	-.324	.499	-.336	.479	-.325	.456	-.349	.418	-.360	.393	-.391	.334	-.482	.229	.600	
.650															.650	
.700	-.301	.448	-.341	.419	-.301	.391	-.354	.361	-.378	.338	-.420	.303	-.481	.190	.700	
.750															.750	
.800	-.330	.409	-.316	.378	-.301	.346	-.345	.323	-.389	.296	-.437	.262	-.472	.160	.800	
.850															.850	
.900	-.349	.354	-.288	.346	-.306	.318	-.336	.298	-.400	.271	-.452	.237	-.462	.143	.900	
.950	-.348	.363	-.281	.335	-.271	.311	-.341	.289	-.398	.258	-.456	.241	-.457	.135	.950	

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR FLAT TRAPEZOIDAL WING

(b) $M = 2.01$

x/c	C_p or $y/\frac{R}{2}$ of :														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 0$																
.0125	.062		.069		.089		.077		.078		.066		.073		.0125	
.050	.064		.073		.081		.073		.073		.058		.055		.050	
.075	.068		.065		.073		.065		.062		.058		.040		.075	
.100	.040		.057		.046		.044		.046		.048		.030		.100	
.150	.036		.050		.058		.063		.058		.049		.023		.150	
.200	.036		.045		.050		.054		.054		.038		.016		.200	
.250	.036		.035		.034		.045		.047		.020		.008		.250	
.300	.012		.020		.031		.035		.031		.024		.001		.300	
.350															.350	
.400	.002		.005		.016		.016		.004		.003		.004		.400	
.450															.450	
.500	.001		.013		.011		.009		.022		.017		.011		.500	
.550															.550	
.600	-.023		-.020		-.023		-.029		-.032		-.033		-.022		.600	
.650															.650	
.700	-.034		-.033		-.042		-.047		-.046		-.043		-.031		.700	
.750															.750	
.800	-.041		-.048		-.056		-.059		-.059		-.059		-.041		.800	
.850															.850	
.900	-.058		-.065		-.067		-.064		-.065		-.063		-.049		.900	
.950	-.062		-.068		-.074		-.070		-.066		-.063		-.056		.950	
$\alpha = 2$																
.0125	.022	.123	.023	.122	.043	.146	.030	.129	.029	.128	.017	.118	.012	.124	.025	
.050	.026	.117	.031	.127	.038	.137	.024	.123	.025	.123	.013	.111	.007	.102	.050	
.075	.015	.091	.019	.115	.027	.125	.017	.113	.016	.112	.013	.103	.007	.077	.075	
.100	.012	.078	.016	.107	.022	.116	.017	.117	.011	.107	.010	.107	.003	.064	.100	
.150	.004	.074	.009	.100	.012	.107	.017	.112	.009	.106	.003	.099	-.001	.052	.150	
.200	.004	.074	.004	.093	.005	.101	.006	.099	.003	.108	.004	.088	-.001	.044	.200	
.250	.013	.077	-.004	.076	-.006	.083	-.003	.091	.001	.093	.016	.066	-.010	.028	.250	
.300	-.025	.054	-.013	.062	-.013	.077	-.011	.080	-.014	.077	-.012	.058	-.015	.020	.300	
.350															.350	
.400	-.030	.039	-.025	.046	-.024	.054	-.029	.055	-.034	.042	-.025	.029	-.022	.005	.400	
.450															.450	
.500	-.031	.039	-.043	.025	-.049	.024	-.048	.025	-.053	.009	-.038	.006	-.023	-.004	.500	
.550															.550	
.600	-.052	.013	-.051	.019	-.056	.014	-.063	-.005	-.059	-.011	-.051	-.013	-.037	-.017	.600	
.650															.650	
.700	-.058	-.006	-.061	.005	-.074	-.013	-.072	-.022	-.067	-.032	-.059	-.033	-.044	-.032	.700	
.750															.750	
.800	-.064	-.014	-.078	-.016	-.081	-.029	-.078	-.040	-.077	-.047	-.071	-.049	-.059	-.046	.800	
.850															.850	
.900	-.081	-.029	-.091	-.036	-.090	-.046	-.086	-.048	-.079	-.056	-.076	-.056	-.070	-.054	.900	
.950	-.085	-.033	-.093	-.049	-.094	-.054	-.086	-.056	-.080	-.057	-.076	-.056	-.078	-.061	.950	
$\alpha = 4$																
.0125	-.012	.180	-.015	.174	-.003	.207	-.014	.191	-.013	.186	-.012	.185	-.012	.185	.025	
.050	-.012	.168	-.011	.185	-.004	.198	-.018	.184	-.017	.181	-.018	.179	-.019	.158	.050	
.075	-.022	.134	-.019	.170	-.013	.181	-.025	.177	-.024	.168	-.024	.170	-.017	.122	.075	
.100	-.022	.125	-.026	.164	-.017	.173	-.025	.183	-.028	.166	-.025	.162	-.017	.103	.100	
.150	-.015	.114	-.032	.153	-.025	.164	-.025	.169	-.032	.165	-.035	.156	-.017	.089	.150	
.200	-.028	.117	-.037	.156	-.035	.155	-.029	.158	-.039	.166	-.039	.147	-.019	.072	.200	
.250	-.015	.122	-.042	.117	-.043	.136	-.040	.145	-.039	.145	-.044	.117	-.024	.054	.250	
.300	-.062	.101	-.050	.106	-.048	.129	-.048	.132	-.048	.133	-.036	.102	-.031	.043	.300	
.350															.350	
.400	-.062	.082	-.060	.087	-.059	.100	-.063	.106	-.063	.084	-.039	.065	-.042	.016	.400	
.450															.450	
.500	-.063	.076	-.072	.068	-.078	.070	-.079	.068	-.072	.046	-.050	.035	-.040	.006	.500	
.550															.550	
.600	-.080	.052	-.084	.058	-.085	.056	-.082	.031	-.077	.018	-.061	.006	-.061	.015	.600	
.650															.650	
.700	-.082	.027	-.109	.038	-.095	.024	-.088	.010	-.082	-.003	-.073	-.013	-.077	-.030	.700	
.750															.750	
.800	-.094	.021	-.102	.019	-.102	.004	-.091	-.011	-.090	-.022	-.088	-.033	-.101	-.043	.800	
.850															.850	
.900	-.107	.007	-.114	-.009	-.106	-.016	-.095	-.023	-.095	-.035	-.095	-.041	-.122	-.052	.900	
.950	-.108	.001	-.112	-.020	-.107	-.026	-.095	-.034	-.095	-.042	-.096	-.041	-.136	-.057	.950	
$\alpha = 6$																
.0125	-.051	.242	-.056	.247	-.046	.280	-.053	.252	-.053	.250	-.059	.245	-.056	.252	.025	
.050	-.046	.219	-.050	.255	-.045	.264	-.057	.252	-.056	.246	-.059	.245	-.056	.221	.050	
.075	-.053	.190	-.056	.238	-.050	.247	-.065	.245	-.065	.231	-.063	.234	-.046	.171	.075	
.100	-.053	.175	-.063	.230	-.058	.238	-.065	.253	-.069	.231	-.063	.230	-.045	.148	.100	
.150	-.053	.158	-.068	.213	-.063	.234	-.065	.227	-.073	.246	-.071	.223	-.042	.122	.150	
.200	-.057	.166	-.073	.193	-.073	.219	-.066	.221	-.079	.225	-.076	.209	-.039	.111	.200	
.250	-.039	.170	-.077	.171	-.080	.194	-.078	.203	-.083	.206	-.072	.169	-.046	.092	.250	
.300	-.090	.154	-.080	.154	-.085	.177	-.085	.186	-.086	.189	-.061	.147	-.056	.073	.300	
.350															.350	
.400	-.090	.130	-.087	.139	-.095	.146	-.099	.151	-.094	.131	-.061	.104	-.069	.042	.400	
.450															.450	
.500	-.085	.125	-.094	.117	-.111	.114	-.110	.105	-.094	.079	-.070	.067	-.077	.026	.500	
.550															.550	
.600	-.103	.094	-.110	.106	-.116	.095	-.110	.067	-.099	.049	-.087	.029	-.103	-.004	.600	
.650															.650	
.700	-.103	.067	-.116	.086	-.115	.060	-.110	.040	-.107	.025	-.099	.009	-.131	-.023	.700	
.750															.750	
.800	-.114	.061	-.127	.056	-.121	.035	-.114	.017	-.118	.002	-.115	-.012	-.144	-.038	.800	
.850															.850	
.900	-.130	.047	-.133	.027	-.123	.008	-.114	.002	-.118	-.013	-.125	-.023	-.176	-.046	.900	
.950	-.133	.042	-.133	.016	-.122	-.004	-.123	-.011	-.118	-.020	-.125	-.023	-.184	-.050	.950	

TABLE II.- Continued
PRESSURE COEFFICIENTS FOR FLAT TRAPEZOIDAL WING
(b) $M = 2.01$ - Continued

x/c	C_p at $y/b = \frac{1}{2}$ of:														x/c	
	.10		.35		.55		.67		.77		.87		.97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 8$																
.0125	-0.089	.308	-0.093	.321	-0.086	.356	-0.087	.332	-0.084	.327	-0.098	.319	-0.086	.328	.0125	
.050	-0.089	.272	-0.088	.328	-0.088	.336	-0.088	.334	-0.088	.326	-0.098	.303	-0.075	.288	.050	
.075	-0.095	.246	-0.095	.308	-0.090	.319	-0.096	.322	-0.096	.311	-0.101	.301	-0.073	.226	.075	
.100	-0.089	.230	-0.103	.300	-0.097	.310	-0.098	.331	-0.099	.318	-0.101	.301	-0.073	.226	.100	
.150	-0.089	.205	-0.108	.269	-0.103	.305	-0.099	.301	-0.103	.321	-0.108	.295	-0.067	.170	.150	
.200	-0.090	.216	-0.111	.247	-0.110	.285	-0.099	.292	-0.107	.300	-0.108	.272	-0.067	.156	.200	
.250	-0.078	.227	-0.117	.226	-0.118	.252	-0.107	.268	-0.113	.276	-0.102	.223	-0.066	.131	.250	
.300	-0.122	.204	-0.117	.208	-0.111	.234	-0.117	.246	-0.114	.252	-0.087	.199	-0.095	.108	.300	
.350															.350	
.400	-0.123	.177	-0.122	.193	-0.123	.202	-0.126	.206	-0.116	.181	-0.095	.146	-0.117	.074	.400	
.450															.450	
.500	-0.123	.174	-0.126	.163	-0.138	.165	-0.134	.152	-0.117	.121	-0.103	.102	-0.140	.047	.500	
.550															.550	
.600	-0.131	.140	-0.141	.155	-0.144	.141	-0.132	.113	-0.122	.090	-0.117	.061	-0.170	.015	.600	
.650															.650	
.700	-0.133	.113	-0.146	.132	-0.137	.102	-0.132	.081	-0.132	.062	-0.130	.037	-0.195	-.005	.700	
.750															.750	
.800	-0.143	.106	-0.153	.098	-0.137	.074	-0.134	.055	-0.138	.037	-0.148	.012	-0.214	-.022	.800	
.850															.850	
.900	-0.157	.087	-0.160	.064	-0.140	.043	-0.133	.038	-0.147	.021	-0.159	.000	-0.210	-.032	.900	
.950	-0.159	.081	-0.161	.056	-0.140	.032	-0.138	.027	-0.143	.013	-0.159	-.004	-0.214	-.040	.950	
$a = 10$																
.0125	-0.112	.368	-0.114	.407	-0.106	.436	-0.112	.416	-0.110	.405	-0.124	.402	-0.111	.358	.0125	
.050	-0.112	.328	-0.111	.407	-0.107	.414	-0.115	.427	-0.113	.404	-0.124	.389	-0.102	.290	.050	
.075	-0.112	.300	-0.117	.379	-0.112	.397	-0.120	.407	-0.120	.396	-0.126	.382	-0.093	.260	.075	
.100	-0.112	.286	-0.123	.362	-0.116	.386	-0.120	.402	-0.124	.393	-0.126	.376	-0.098	.225	.100	
.150	-0.106	.256	-0.129	.325	-0.115	.349	-0.129	.342	-0.124	.328	-0.134	.376	-0.099	.206	.150	
.200	-0.106	.270	-0.129	.304	-0.128	.342	-0.124	.354	-0.132	.369	-0.133	.331	-0.099	.200	.200	
.250	-0.091	.281	-0.135	.277	-0.138	.305	-0.129	.322	-0.139	.336	-0.127	.283	-0.116	.175	.250	
.300	-0.129	.259	-0.135	.262	-0.140	.286	-0.137	.302	-0.139	.301	-0.112	.249	-0.128	.144	.300	
.350															.350	
.400	-0.136	.228	-0.137	.243	-0.147	.258	-0.150	.255	-0.135	.222	-0.118	.184	-0.164	.097	.400	
.450															.450	
.500	-0.135	.222	-0.139	.214	-0.163	.212	-0.155	.193	-0.135	.154	-0.124	.136	-0.190	.068	.500	
.550															.550	
.600	-0.144	.191	-0.154	.203	-0.164	.184	-0.149	.149	-0.140	.125	-0.138	.093	-0.213	.032	.600	
.650															.650	
.700	-0.145	.159	-0.158	.173	-0.157	.142	-0.150	.119	-0.146	.093	-0.154	.072	-0.229	.010	.700	
.750															.750	
.800	-0.156	.152	-0.165	.135	-0.149	.112	-0.153	.086	-0.158	.067	-0.174	.044	-0.237	-.007	.800	
.850															.850	
.900	-0.169	.134	-0.171	.104	-0.154	.076	-0.149	.067	-0.165	.047	-0.185	.024	-0.228	-.019	.900	
.950	-0.169	.129	-0.164	.093	-0.154	.063	-0.152	.048	-0.165	.036	-0.187	.020	-0.226	-.029	.950	
$a = 12$																
.0125	-0.142	.435	-0.139	.509	-0.136	.539	-0.143	.530	-0.141	.511	-0.155	.498	-0.152	.508	.0125	
.050	-0.142	.389	-0.140	.489	-0.142	.501	-0.146	.513	-0.141	.504	-0.145	.447	-0.140	.447	.050	
.075	-0.141	.366	-0.148	.456	-0.142	.490	-0.150	.500	-0.150	.488	-0.155	.490	-0.130	.375	.075	
.100	-0.140	.345	-0.152	.433	-0.146	.467	-0.150	.495	-0.153	.482	-0.155	.479	-0.118	.342	.100	
.150	-0.133	.313	-0.157	.387	-0.149	.433	-0.156	.445	-0.157	.476	-0.162	.459	-0.127	.292	.150	
.200	-0.129	.324	-0.157	.367	-0.156	.404	-0.154	.421	-0.160	.436	-0.160	.398	-0.138	.263	.200	
.250	-0.117	.345	-0.163	.339	-0.165	.366	-0.160	.385	-0.167	.396	-0.151	.336	-0.159	.223	.250	
.300	-0.154	.324	-0.166	.324	-0.168	.343	-0.167	.362	-0.171	.354	-0.137	.299	-0.175	.185	.300	
.350															.350	
.400	-0.163	.287	-0.165	.301	-0.176	.317	-0.174	.309	-0.157	.269	-0.145	.226	-0.213	.134	.400	
.450															.450	
.500	-0.156	.281	-0.162	.268	-0.189	.271	-0.179	.241	-0.159	.202	-0.152	.171	-0.235	.104	.500	
.550															.550	
.600	-0.170	.246	-0.173	.257	-0.191	.230	-0.171	.195	-0.164	.169	-0.167	.132	-0.251	.066	.600	
.650															.650	
.700	-0.164	.213	-0.180	.229	-0.179	.189	-0.171	.159	-0.171	.138	-0.185	.108	-0.242	.047	.700	
.750															.750	
.800	-0.175	.205	-0.186	.185	-0.167	.153	-0.176	.128	-0.184	.106	-0.202	.079	-0.268	.026	.800	
.850															.850	
.900	-0.186	.181	-0.190	.147	-0.174	.114	-0.176	.103	-0.191	.083	-0.218	.058	-0.253	.009	.900	
.950	-0.190	.171	-0.185	.134	-0.179	.102	-0.176	.087	-0.194	.073	-0.216	.057	-0.250	-.000	.950	
$a = 14$																
.0125	-0.175	.501	-0.171	.635	-0.165	.679	-0.167	.679	-0.167	.671	-0.178	.593	-0.160	.525	.0125	
.050	-0.171	.447	-0.170	.556	-0.167	.580	-0.172	.595	-0.168	.588	-0.181	.595	-0.151	.459	.050	
.075	-0.168	.430	-0.176	.514	-0.168	.574	-0.177	.584	-0.177	.588	-0.181	.571	-0.138	.406	.075	
.100	-0.166	.414	-0.180	.488	-0.173	.537	-0.177	.565	-0.180	.568	-0.183	.571	-0.138	.342	.100	
.150	-0.161	.375	-0.186	.447	-0.177	.499	-0.181	.516	-0.183	.543	-0.186	.524	-0.151	.242	.150	
.200	-0.152	.382	-0.186	.424	-0.183	.471	-0.180	.482	-0.187	.484	-0.183	.448	-0.168	.301	.200	
.250	-0.146	.402	-0.191	.402	-0.190	.425	-0.184	.443	-0.192	.455	-0.173	.381	-0.190	.254	.250	
.300	-0.175	.386	-0.191	.381	-0.194	.404	-0.190	.425	-0.194	.401	-0.158	.341	-0.211	.218	.300	
.350															.350	
.400	-0.185	.338	-0.192	.363	-0.201	.380	-0.200	.358	-0.180	.312	-0.168	.267	-0.241	.160	.400	
.450															.450	
.500	-0.179	.338	-0.187	.323	-0.213	.323	-0.198	.290	-0.180	.248	-0.179	.213	-0.259	.127	.500	
.550															.550	
.600	-0.187	.305	-0.198	.311	-0.212	.272	-0.193	.241	-0.185	.204	-0.191	.1				

TABLE II.- Concluded
PRESSURE COEFFICIENTS FOR FLAT TRAPEZOIDAL WING
(b) $M = 2.01$ - Concluded

x/c	C_p or $y/2$ of:														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 16$																
.0125	-+194	+572	-+190	+730	-+186	+786	-+187	+789	-+187	+800	-+185	+801	-+194	+801	.0125	
.025	-+193	+517	-+190	+639	-+188	+689	-+192	+715	-+187	+723	-+200	+727	-+185	+616	.025	
.050	-+187	+493	-+195	+589	-+190	+654	-+197	+668	-+198	+681	-+206	+699	-+175	+611	.050	
.075	-+187	+487	-+200	+556	-+195	+614	-+197	+635	-+202	+651	-+201	+657	-+169	+478	.075	
.100	-+188	+487	-+206	+517	-+199	+571	-+201	+589	-+204	+624	-+211	+588	-+186	+401	.100	
.150	-+182	+450	-+206	+490	-+205	+538	-+202	+550	-+208	+569	-+204	+504	-+201	+352	.150	
.200	-+167	+444	-+206	+490	-+205	+538	-+202	+550	-+208	+569	-+204	+434	-+225	+302	.200	
.250	-+161	+463	-+210	+469	-+210	+495	-+206	+519	-+213	+513	-+195	+434	-+225	+302	.250	
.300	-+188	+452	-+211	+453	-+211	+471	-+211	+485	-+216	+456	-+183	+394	-+243	+264	.300	
.350	-+188	+452	-+211	+453	-+211	+471	-+211	+485	-+216	+456	-+183	+394	-+243	+264	.350	
.400	-+191	+428	-+213	+425	-+218	+447	-+218	+412	-+198	+369	-+194	+318	-+270	+210	.400	
.450	-+194	+416	-+204	+395	-+228	+380	-+216	+340	-+196	+300	-+206	+264	-+281	+177	.450	
.500	-+194	+416	-+204	+395	-+228	+380	-+216	+340	-+196	+300	-+206	+264	-+281	+177	.500	
.550	-+203	+367	-+211	+376	-+231	+332	-+213	+293	-+204	+261	-+217	+221	-+287	+139	.550	
.600	-+203	+367	-+211	+376	-+231	+332	-+213	+293	-+204	+261	-+217	+221	-+287	+139	.600	
.650	-+197	+308	-+218	+316	-+218	+287	-+213	+256	-+214	+229	-+234	+195	-+295	+117	.650	
.700	-+197	+308	-+218	+316	-+218	+287	-+213	+256	-+214	+229	-+234	+195	-+295	+117	.700	
.750	-+210	+310	-+222	+264	-+202	+237	-+218	+212	-+226	+194	-+251	+161	-+296	+088	.750	
.800	-+210	+310	-+222	+264	-+202	+237	-+218	+212	-+226	+194	-+251	+161	-+296	+088	.800	
.850	-+218	+270	-+221	+231	-+202	+199	-+220	+185	-+232	+161	-+263	+134	-+284	+070	.850	
.900	-+222	+259	-+213	+215	-+206	+184	-+220	+165	-+237	+150	-+261	+126	-+281	+059	.900	
$a = 18$																
.0125	-+217	+645	-+212	+782	-+206	+874	-+207	+882	-+207	+889	-+222	+896	-+222	+896	.0125	
.025	-+214	+590	-+214	+720	-+211	+784	-+212	+809	-+207	+824	-+222	+835	-+211	+730	.025	
.050	-+211	+555	-+219	+656	-+215	+725	-+217	+748	-+217	+761	-+228	+784	-+201	+611	.050	
.075	-+210	+554	-+222	+627	-+217	+685	-+217	+704	-+221	+727	-+219	+734	-+201	+534	.075	
.100	-+209	+520	-+226	+580	-+221	+641	-+222	+654	-+225	+691	-+233	+644	-+220	+451	.100	
.150	-+185	+508	-+226	+556	-+227	+602	-+222	+617	-+229	+625	-+229	+559	-+237	+400	.150	
.200	-+181	+533	-+232	+532	-+233	+560	-+227	+581	-+234	+562	-+220	+488	-+259	+348	.200	
.250	-+207	+513	-+232	+518	-+233	+537	-+231	+537	-+234	+499	-+211	+439	-+273	+307	.250	
.300	-+216	+490	-+234	+486	-+240	+502	-+236	+462	-+221	+415	-+222	+367	-+294	+252	.300	
.350	-+216	+490	-+234	+486	-+240	+502	-+236	+462	-+221	+415	-+222	+367	-+294	+252	.350	
.400	-+215	+466	-+226	+454	-+247	+424	-+233	+385	-+218	+346	-+234	+312	-+302	+217	.400	
.450	-+215	+466	-+226	+454	-+247	+424	-+233	+385	-+218	+346	-+234	+312	-+302	+217	.450	
.500	-+222	+420	-+229	+420	-+250	+377	-+230	+336	-+223	+308	-+245	+265	-+306	+178	.500	
.600	-+222	+420	-+229	+420	-+250	+377	-+230	+336	-+223	+308	-+245	+265	-+306	+178	.600	
.650	-+215	+366	-+235	+358	-+239	+323	-+229	+295	-+234	+269	-+260	+238	-+312	+149	.650	
.700	-+215	+366	-+235	+358	-+239	+323	-+229	+295	-+234	+269	-+260	+238	-+312	+149	.700	
.750	-+226	+353	-+236	+303	-+227	+277	-+236	+253	-+245	+228	-+274	+199	-+312	+125	.750	
.800	-+226	+353	-+236	+303	-+227	+277	-+236	+253	-+245	+228	-+274	+199	-+312	+125	.800	
.850	-+236	+107	-+236	+273	-+224	+233	-+241	+224	-+252	+199	-+287	+168	-+303	+103	.850	
.900	-+239	+292	-+226	+257	-+219	+219	-+242	+199	-+257	+186	-+285	+170	-+299	+088	.900	
$a = 20$																
.0125	-+234	+731	-+229	+873	-+229	+971	-+234	+997	-+237	+1008	-+239	+936	-+245	+977	.0125	
.025	-+234	+639	-+232	+810	-+228	+888	-+234	+907	-+229	+931	-+242	+862	-+227	+789	.025	
.050	-+227	+634	-+236	+744	-+232	+808	-+236	+819	-+238	+851	-+242	+862	-+216	+617	.050	
.075	-+227	+624	-+242	+718	-+235	+774	-+236	+802	-+242	+817	-+236	+804	-+216	+597	.075	
.100	-+223	+599	-+246	+667	-+239	+722	-+245	+735	-+246	+773	-+248	+702	-+238	+506	.100	
.150	-+197	+589	-+245	+640	-+245	+683	-+245	+699	-+250	+698	-+242	+616	-+259	+448	.150	
.200	-+192	+616	-+250	+616	-+250	+635	-+249	+652	-+256	+672	-+234	+543	-+276	+393	.200	
.250	-+224	+593	-+250	+599	-+255	+616	-+255	+607	-+255	+595	-+226	+496	-+286	+348	.250	
.300	-+235	+569	-+254	+567	-+260	+569	-+260	+524	-+245	+479	-+238	+425	-+304	+292	.300	
.350	-+235	+569	-+254	+567	-+260	+569	-+260	+524	-+245	+479	-+238	+425	-+304	+292	.350	
.400	-+230	+532	-+244	+520	-+267	+486	-+253	+447	-+241	+401	-+249	+362	-+310	+260	.400	
.450	-+230	+489	-+247	+473	-+268	+436	-+247	+393	-+245	+368	-+263	+312	-+312	+217	.450	
.500	-+239	+489	-+247	+473	-+257	+374	-+251	+352	-+257	+327	-+276	+285	-+318	+193	.500	
.600	-+228	+429	-+251	+409	-+257	+374	-+251	+352	-+257	+327	-+276	+285	-+318	+193	.600	
.650	-+215	+429	-+251	+409	-+257	+374	-+251	+352	-+257	+327	-+276	+285	-+318	+193	.650	
.700	-+215	+429	-+251	+409	-+257	+374	-+251	+352	-+257	+327	-+276	+285	-+318	+193	.700	
.750	-+241	+406	-+254	+357	-+246	+327	-+257	+307	-+268	+280	-+287	+240	-+318	+159	.750	
.800	-+249	+354	-+247	+325	-+246	+285	-+264	+276	-+275	+249	-+300	+213	-+310	+135	.800	
.850	-+249	+332	-+219	+307	-+244	+272	-+263	+249	-+280	+230	-+299	+204	-+311	+122	.850	
.900	-+253	+332	-+219	+307	-+244	+272	-+263	+249	-+280	+230	-+299	+204	-+311	+122	.900	
$a = 24$																

TABLE III
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(a) $M = 1.61$

x/c	Cp at $y/\frac{c}{2}$ of:														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -20$																
.0125	.963	-.382	.920	-.395	.977	-.414	1.022	-.438	1.082	-.467	1.112	-.484	1.005	-.484	.0125	
.025	.849	-.364	.831	-.366	.880	-.388	.939	-.400	1.009	-.410	1.021	-.438	.859	-.451	.025	
.050	.782	-.352	.767	-.367	.826	-.365	.895	-.382	.946	-.401	.933	-.430	.751	-.400	.050	
.075	.741	-.335	.712	-.355	.786	-.361	.854	-.381	.906	-.388	.887	-.420	.643	-.412	.075	
.100	.695	-.271	.645	-.349	.731	-.351	.793	-.370	.817	-.391	.779	-.409	.601	-.434	.100	
.150	.650	-.261	.609	-.344	.693	-.355	.743	-.366	.744	-.373	.707	-.382	.541	-.458	.200	
.200	.600	-.217	.586	-.345	.656	-.351	.691	-.362	.691	-.379	.647	-.361	.468	-.467	.300	
.250	.560	-.217	.539	-.344	.627	-.362	.649	-.374	.649	-.390	.620	-.366	.455	-.477	.350	
.300	.548	-.289	.566	-.351	.627	-.362	.649	-.374	.649	-.390	.620	-.366	.455	-.477	.350	
.350	.546	-.272	.530	-.280	.575	-.359	.562	-.380	.574	-.342	.550	-.386	.420	-.486	.400	
.400	.546	-.272	.530	-.280	.575	-.359	.562	-.380	.574	-.342	.550	-.386	.420	-.486	.400	
.450	.468	-.274	.489	-.293	.506	-.372	.534	-.347	.524	-.345	.482	-.407	.380	-.490	.500	
.500	.434	-.296	.472	-.311	.477	-.349	.473	-.326	.472	-.354	.443	-.424	.313	-.486	.600	
.600	.434	-.296	.472	-.311	.477	-.349	.473	-.326	.472	-.354	.443	-.424	.261	-.475	.700	
.650	.442	-.278	.461	-.332	.457	-.295	.436	-.320	.428	-.373	.390	-.437	.261	-.475	.750	
.700	.472	-.313	.486	-.357	.457	-.300	.420	-.320	.382	-.375	.331	-.444	.205	-.458	.800	
.800	.511	-.347	.512	-.325	.471	-.314	.414	-.333	.364	-.379	.271	-.444	.154	-.434	.900	
.900	.536	-.357	.539	-.305	.505	-.319	.434	-.338	.378	-.380	.284	-.437	.139	-.369	.950	
$\alpha = -18$																
.0125	.868	-.335	.829	-.353	.891	-.362	.932	-.384	.998	-.416	1.046	-.444	.944	-.461	.0125	
.025	.762	-.325	.741	-.327	.793	-.348	.848	-.353	.930	-.370	.950	-.399	.791	-.418	.050	
.050	.694	-.315	.674	-.333	.731	-.327	.803	-.345	.866	-.365	.856	-.386	.673	-.364	.075	
.075	.655	-.300	.617	-.323	.687	-.322	.763	-.339	.826	-.343	.811	-.377	.576	-.367	.100	
.100	.603	-.243	.555	-.312	.628	-.317	.707	-.333	.731	-.362	.699	-.339	.524	-.396	.150	
.150	.571	-.236	.496	-.309	.588	-.323	.654	-.330	.687	-.344	.622	-.339	.463	-.434	.200	
.200	.508	-.181	.472	-.312	.555	-.319	.597	-.328	.593	-.350	.552	-.333	.387	-.456	.250	
.250	.436	-.270	.441	-.315	.509	-.333	.543	-.343	.584	-.366	.520	-.335	.378	-.466	.300	
.300	.408	-.243	.409	-.244	.458	-.327	.457	-.351	.468	-.319	.443	-.357	.332	-.484	.400	
.350	.374	-.247	.404	-.265	.405	-.337	.436	-.316	.425	-.323	.388	-.381	.302	-.488	.500	
.400	.363	-.272	.392	-.285	.396	-.314	.391	-.301	.389	-.328	.361	-.398	.245	-.483	.600	
.450	.370	-.257	.390	-.309	.384	-.272	.359	-.278	.342	-.343	.317	-.414	.197	-.468	.700	
.500	.392	-.293	.409	-.340	.370	-.281	.339	-.288	.300	-.339	.254	-.419	.138	-.436	.800	
.600	.398	-.328	.406	-.309	.374	-.293	.324	-.301	.273	-.344	.187	-.415	.086	-.400	.900	
.700	.414	-.341	.416	-.303	.375	-.296	.313	-.307	.271	-.345	.193	-.404	.057	-.375	.950	
$\alpha = -16$																
.0125	.787	-.294	.754	-.314	.821	-.317	.853	-.331	.933	-.361	.991	-.388	.897	-.408	.0125	
.025	.688	-.286	.673	-.292	.721	-.310	.713	-.313	.866	-.323	.897	-.348	.741	-.377	.050	
.050	.615	-.276	.607	-.300	.659	-.294	.731	-.306	.803	-.322	.809	-.342	.626	-.307	.075	
.075	.590	-.259	.552	-.290	.613	-.280	.693	-.309	.769	-.300	.763	-.339	.530	-.325	.100	
.100	.532	-.213	.485	-.280	.558	-.284	.645	-.303	.684	-.327	.648	-.343	.487	-.355	.150	
.150	.531	-.210	.444	-.274	.551	-.294	.601	-.301	.615	-.310	.577	-.302	.425	-.405	.200	
.200	.456	-.148	.410	-.218	.508	-.289	.545	-.291	.548	-.317	.506	-.302	.348	-.430	.250	
.250	.383	-.256	.381	-.278	.457	-.304	.488	-.314	.497	-.333	.468	-.302	.336	-.442	.300	
.300	.341	-.217	.354	-.206	.404	-.297	.404	-.323	.421	-.291	.397	-.321	.292	-.465	.350	
.350	.341	-.217	.354	-.206	.404	-.297	.404	-.323	.421	-.291	.397	-.321	.292	-.465	.350	
.400	.341	-.217	.354	-.206	.404	-.297	.404	-.323	.421	-.291	.397	-.321	.292	-.465	.350	
.450	.302	-.221	.333	-.244	.334	-.297	.359	-.283	.361	-.289	.331	-.342	.251	-.471	.500	
.500	.313	-.271	.326	-.320	.289	-.260	.258	-.258	.222	-.297	.181	-.379	.087	-.412	.600	
.600	.292	-.252	.302	-.260	.315	-.286	.302	-.266	.305	-.292	.288	-.364	.186	-.475	.650	
.650	.287	-.236	.310	-.285	.296	-.254	.271	-.247	.267	-.300	.242	-.380	.138	-.450	.700	
.700	.287	-.236	.310	-.285	.296	-.254	.271	-.247	.267	-.300	.242	-.380	.138	-.450	.750	
.750	.313	-.271	.326	-.320	.289	-.260	.258	-.258	.222	-.297	.181	-.379	.087	-.412	.800	
.800	.323	-.308	.332	-.293	.298	-.273	.238	-.271	.196	-.306	.120	-.375	.031	-.363	.900	
.900	.343	-.322	.345	-.286	.298	-.277	.243	-.277	.194	-.306	.119	-.371	.003	-.344	.950	
$\alpha = -14$																
.0125	.702	-.253	.666	-.275	.738	-.273	.766	-.285	.848	-.312	.921	-.328	.842	-.356	.025	
.025	.617	-.250	.602	-.256	.649	-.271	.692	-.273	.791	-.279	.829	-.296	.683	-.329	.050	
.050	.550	-.237	.533	-.261	.589	-.260	.648	-.267	.729	-.284	.747	-.294	.570	-.355	.075	
.100	.477	-.221	.488	-.257	.538	-.261	.613	-.274	.698	-.271	.696	-.297	.479	-.424	.100	
.150	.470	-.187	.422	-.241	.482	-.250	.572	-.272	.624	-.296	.689	-.305	.438	-.411	.150	
.200	.477	-.184	.382	-.236	.454	-.262	.535	-.269	.554	-.282	.516	-.268	.356	-.358	.200	
.250	.408	-.118	.354	-.244	.448	-.259	.493	-.265	.491	-.286	.456	-.271	.305	-.393	.250	
.300	.331	-.236	.324	-.233	.407	-.273	.437	-.290	.439	-.307	.419	-.272	.299	-.403	.300	
.350	.283	-.187	.298	-.168	.350	-.265	.351	-.294	.366	-.267	.350	-.286	.252	-.435	.350	
.400	.240	-.190	.282	-.221	.276	-.255	.306	-.257	.306	-.261	.280	-.305	.213	-.451	.450	
.450	.236	-.225	.250	-.237	.261	-.259	.245	-.228	.257	-.259	.238	-.324	.149	-.456	.500	
.500	.236	-.225	.250	-.237	.261	-.259	.245	-.228	.257	-.259	.238	-.324	.149	-.456	.600	
.600	.236	-.225	.250	-.237	.261	-.259	.245	-.228	.257	-.259	.238	-.324	.149	-.456	.650	
.700	.238	-.211	.254	-.261	.237	-.237	.220	-.221	.217	-.259	.191	-.337	.098	-.425	.750	
.750	.260	-.249	.270	-.298	.228	-.243	.195	-.236	.169	-.259	.136	-.332	.050	-.375	.800	
.800	.265	-.286	.279	-.276	.232	-.254	.177	-.244	.140	-.271	.067	-.332	-.010	-.325	.850	
.900	.280	-.300	.284	-.272	.248	-.260	.176	-.250	.129	-.271	.064	-.332	-.039	-.304	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(a) M = 1.61 - Continued

X/Y	Cp at $y = \frac{1}{2}$ of:														X/C	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = -12$																
.0125	.632	-.208	.386	-.227	.661	-.221	.683	-.229	.746	-.250	.832	-.266	.775	-.288	.0125	
.025	.531	-.212	.520	-.214	.573	-.225	.617	-.227	.701	-.225	.753	-.246	.617	-.263	.0225	
.050	.486	-.192	.472	-.223	.514	-.214	.566	-.225	.641	-.237	.680	-.246	.509	-.206	.050	
.075	.420	-.179	.415	-.216	.466	-.213	.537	-.231	.622	-.238	.637	-.254	.424	-.234	.075	
.100	.405	-.157	.359	-.203	.407	-.209	.498	-.234	.559	-.265	.526	-.269	.383	-.255	.100	
.125	.404	-.153	.312	-.194	.376	-.215	.470	-.230	.489	-.248	.456	-.235	.329	-.306	.125	
.150	.352	-.086	.298	-.201	.368	-.213	.431	-.229	.432	-.252	.398	-.240	.260	-.344	.150	
.200	.272	-.215	.263	-.177	.350	-.234	.382	-.253	.381	-.269	.362	-.237	.250	-.356	.200	
.350															.350	
.400	.231	-.154	.236	-.128	.300	-.219	.304	-.252	.312	-.235	.301	-.250	.216	-.399	.400	
.450															.450	
.500	.180	-.153	.232	-.195	.226	-.210	.254	-.221	.247	-.229	.233	-.267	.180	-.417	.500	
.550															.550	
.600	.177	-.193	.202	-.208	.206	-.229	.194	-.190	.204	-.217	.195	-.281	.109	-.431	.600	
.650															.650	
.700	.184	-.183	.199	-.235	.184	-.212	.173	-.196	.163	-.211	.144	-.287	.060	-.385	.700	
.750															.750	
.800	.211	-.220	.208	-.276	.170	-.217	.145	-.212	.123	-.219	.093	-.276	.018	-.333	.800	
.850															.850	
.900	.216	-.261	.229	-.258	.174	-.226	.122	-.218	.085	-.231	.028	-.285	.031	-.283	.900	
.950	.232	-.275	.238	-.256	.193	-.233	.117	-.221	.077	-.231	.015	-.281	.059	-.265	.950	
$a = -10$																
.0125	.540	-.164	.528	-.182	.591	-.176	.612	-.174	.656	-.196	.734	-.201	.705	-.224	.0125	
.025	.420	-.171	.426	-.170	.488	-.184	.516	-.181	.612	-.184	.673	-.198	.541	-.204	.0225	
.050	.400	-.150	.397	-.178	.454	-.175	.494	-.184	.565	-.195	.617	-.202	.450	-.160	.050	
.075	.360	-.136	.354	-.171	.411	-.172	.460	-.190	.536	-.201	.573	-.210	.375	-.182	.100	
.100	.341	-.127	.284	-.157	.340	-.162	.426	-.189	.495	-.214	.470	-.227	.337	-.201	.150	
.125	.325	-.124	.253	-.146	.305	-.175	.400	-.186	.432	-.213	.400	-.201	.287	-.253	.200	
.150	.291	-.058	.239	-.149	.301	-.174	.374	-.183	.372	-.209	.345	-.207	.212	-.291	.250	
.200	.224	-.194	.211	-.131	.285	-.188	.327	-.216	.325	-.235	.311	-.203	.208	-.298	.300	
.350															.350	
.400	.179	-.122	.175	-.088	.254	-.172	.254	-.222	.263	-.201	.253	-.208	.173	-.351	.400	
.450															.450	
.500	.132	-.120	.188	-.158	.174	-.171	.211	-.175	.205	-.188	.197	-.222	.144	-.375	.500	
.550															.550	
.600	.121	-.162	.162	-.172	.161	-.199	.149	-.157	.159	-.169	.150	-.231	.075	-.390	.600	
.650															.650	
.700	.131	-.152	.152	-.204	.143	-.184	.125	-.169	.116	-.169	.102	-.226	.031	-.342	.700	
.750															.750	
.800	.166	-.194	.154	-.248	.122	-.192	.106	-.179	.084	-.183	.053	-.222	.011	-.283	.800	
.850															.850	
.900	.172	-.236	.181	-.234	.120	-.204	.079	-.183	.045	-.189	.001	-.236	.053	-.237	.900	
.950	.188	-.249	.191	-.228	.133	-.210	.071	-.189	.036	-.189	.011	-.226	.076	-.219	.950	
$a = -8$																
.0125	.420	-.122	.448	-.136	.501	-.132	.513	-.131	.555	-.147	.630	-.143	.618	-.162	.0125	
.025	.341	-.128	.338	-.123	.376	-.143	.391	-.143	.483	-.135	.544	-.144	.447	-.150	.0225	
.050	.314	-.108	.316	-.130	.376	-.135	.401	-.146	.475	-.143	.534	-.150	.384	-.108	.050	
.075	.289	-.092	.279	-.123	.336	-.128	.378	-.148	.460	-.154	.523	-.167	.330	-.130	.100	
.100	.273	-.097	.219	-.104	.272	-.119	.344	-.155	.430	-.171	.413	-.181	.290	-.149	.150	
.125	.251	-.091	.186	-.096	.233	-.151	.311	-.143	.374	-.168	.344	-.163	.241	-.191	.200	
.150	.221	-.031	.170	-.088	.225	-.125	.302	-.138	.317	-.169	.292	-.169	.175	-.234	.250	
.200	.172	-.154	.155	-.089	.206	-.146	.258	-.170	.265	-.194	.259	-.164	.167	-.242	.300	
.350															.350	
.400	.129	-.086	.117	-.056	.198	-.125	.191	-.176	.209	-.162	.202	-.164	.134	-.295	.400	
.450															.450	
.500	.088	-.082	.134	-.114	.122	-.138	.154	-.137	.162	-.142	.150	-.175	.106	-.327	.500	
.550															.550	
.600	.064	-.137	.117	-.140	.112	-.179	.086	-.125	.114	-.127	.112	-.173	.050	-.338	.600	
.650															.650	
.700	.074	-.122	.103	-.173	.095	-.165	.076	-.144	.070	-.134	.058	-.167	.003	-.290	.700	
.750															.750	
.800	.115	-.160	.098	-.221	.073	-.173	.058	-.148	.042	-.148	.013	-.167	.037	-.240	.800	
.850															.850	
.900	.126	-.210	.122	-.212	.067	-.185	.027	-.152	.012	-.141	.033	-.173	.065	-.198	.900	
.950	.140	-.223	.139	-.207	.078	-.193	.022	-.151	.004	-.140	.039	-.167	.084	-.174	.950	
$a = -6$																
.0125	.313	-.057	.323	-.075	.345	-.067	.359	-.064	.399	-.091	.470	-.083	.495	-.099	.0125	
.025	.266	-.066	.257	-.059	.299	-.079	.347	-.078	.380	-.080	.383	-.089	.338	-.091	.0225	
.050	.200	-.040	.202	-.054	.242	-.066	.310	-.084	.362	-.087	.422	-.095	.298	-.060	.050	
.075	.212	-.044	.206	-.056	.252	-.066	.293	-.088	.366	-.103	.422	-.116	.248	-.068	.100	
.100	.205	-.052	.149	-.036	.197	-.051	.264	-.091	.330	-.122	.343	-.128	.234	-.084	.150	
.125	.179	-.044	.109	-.025	.159	-.063	.239	-.078	.304	-.118	.281	-.125	.193	-.122	.200	
.150	.161	-.006	.103	-.027	.165	-.052	.225	-.085	.250	-.113	.231	-.130	.129	-.162	.250	
.200	.117	-.085	.094	-.036	.119	-.078	.199	-.117	.198	-.147	.196	-.119	.126	-.156	.300	
.350															.350	
.400	.074	-.036	.060	-.020	.133	-.067	.125	-.123	.135	-.118	.140	-.116	.091	-.205	.400	
.450															.450	
.500	.044	-.029	.070	-.062	.073	-.096	.101	-.100	.092	-.097	.090	-.113	.060	-.239	.500	
.550															.550	
.600	.011	-.084	.071	-.094	.047	-.138	.037	-.105	.063	-.085	.063	-.113	.010	-.245	.600	
.650															.650	
.700	.013	-.083	.051	-.134	.041	-.132	.019	-.121	.012	-.105	.011	-.108	.022	-.219	.700	
.750															.750	
.800	.059	-.116	.051	-.186	.021	-.142	.014	-.124	.012	-.105	.040	-.113	.055	-.170	.800	
.850															.850	
.900	.078	-.168	.065	-.192	.021	-.158	.018	-.124	.003	-.105	.070	-.108	.083	-.143	.900	
.950	.093	-.190	.082	-.189	.029	-.167	.022	-.124	.043	-.101	.076	-.101	.091	-.121	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING
(a) $M = 1.61$ - Continued

x/c	Cp at $y/b = 0$														x/c	
	.10		.35		.55		.67		.77		.87		.97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -4$																
.0125															.0125	
.025	.247	-.013	.242	-.027	.253	-.019	.270	-.017	.295	-.036	.349	-.038	.362	-.054	.025	
.050	.215	-.023	.202	-.034	.239	-.033	.243	-.033	.310	-.027	.327	-.043	.301	-.059	.050	
.075	.172	-.012	.171	-.014	.226	-.020	.243	-.043	.289	-.043	.340	-.053	.244	-.039	.075	
.100	.160	-.011	.152	-.000	.197	-.015	.229	-.046	.289	-.057	.336	-.074	.217	-.041	.100	
.125	.153	-.021	.102	-.017	.148	-.003	.208	-.043	.265	-.078	.294	-.095	.196	-.042	.150	
.150	.140	-.010	.048	-.029	.113	-.009	.181	-.035	.253	-.072	.238	-.095	.165	-.072	.200	
.175	.109	-.026	.055	-.017	.099	-.002	.169	-.028	.209	-.065	.193	-.101	.102	-.118	.250	
.200	.096	-.024	.066	-.006	.073	-.031	.149	-.066	.163	-.105	.163	-.075	.100	-.093	.300	
.225	.074	-.024	.046	-.006	.073	-.031	.149	-.066	.163	-.105	.163	-.075	.100	-.093	.350	
.250	.028	-.004	.019	-.016	.081	-.027	.077	-.078	.107	-.078	.106	-.073	.066	-.128	.400	
.275	.012	-.011	.019	-.021	.039	-.065	.057	-.069	.048	-.061	.052	-.073	.040	-.153	.500	
.300	.012	-.047	.036	-.055	.007	-.106	.004	-.080	.035	-.052	.018	-.070	-.018	-.166	.600	
.325	-.029	-.047	.036	-.055	.007	-.106	.004	-.080	.035	-.052	.018	-.070	-.018	-.166	.650	
.350	-.029	-.047	.036	-.055	.007	-.106	.004	-.080	.035	-.052	.018	-.070	-.018	-.166	.700	
.375	-.025	-.060	.017	-.101	.011	-.110	-.024	-.087	-.020	-.078	-.015	-.069	-.036	-.156	.750	
.400	-.025	-.090	.019	-.154	-.007	-.124	-.023	-.091	-.047	-.076	-.065	-.073	-.064	-.122	.800	
.425	-.012	-.090	.019	-.154	-.007	-.124	-.023	-.091	-.047	-.076	-.065	-.073	-.064	-.122	.850	
.450	-.035	-.138	.030	-.169	-.012	-.140	-.043	-.095	-.059	-.070	-.098	-.061	-.088	-.101	.900	
.475	-.053	-.158	.043	-.174	-.006	-.146	-.056	-.097	-.065	-.068	-.089	-.057	-.093	-.082	.950	
$\alpha = -2$																
.0125															.0125	
.025	.175	.051	.161	.040	.166	.046	.180	.050	.201	.018	.236	.025	.232	.003	.025	
.050	.146	.040	.127	.052	.155	.032	.170	.039	.215	.031	.242	.016	.219	-.006	.050	
.075	.112	.039	.113	.059	.153	.051	.154	.021	.202	.010	.244	.002	.180	-.009	.075	
.100	.103	.036	.091	.078	.123	.049	.148	.021	.193	-.004	.240	-.017	.165	-.014	.100	
.125	.097	.026	.049	.086	.089	.067	.137	.028	.186	-.025	.223	.046	.153	-.025	.150	
.150	.067	.042	.019	.089	.053	.067	.123	.037	.173	.013	.183	.058	.132	-.040	.200	
.175	.057	.066	-.003	.070	.043	.062	.107	.037	.150	.008	.148	.062	.073	-.065	.250	
.200	.023	.039	-.009	.064	.019	.025	.090	.015	.110	.052	.120	.042	.078	-.033	.300	
.225	-.016	.059	-.029	.063	.016	-.027	.037	-.036	.061	-.052	.071	-.032	.046	-.042	.400	
.250	-.027	.061	-.032	.030	-.008	-.025	.006	-.047	.003	-.043	.019	-.027	.023	-.048	.500	
.275	-.067	.001	-.011	-.013	-.037	-.072	-.029	-.067	-.027	-.030	-.021	-.027	-.035	-.063	.600	
.300	-.065	-.021	-.021	-.065	-.034	-.086	-.061	-.065	-.061	-.048	-.050	-.032	-.058	-.056	.700	
.325	-.065	-.021	-.021	-.065	-.034	-.086	-.061	-.065	-.061	-.048	-.050	-.032	-.058	-.056	.750	
.350	-.025	-.059	-.015	-.119	-.039	-.100	-.059	-.071	-.091	-.048	-.050	-.026	-.069	-.045	.800	
.375	-.025	-.059	-.015	-.119	-.039	-.100	-.059	-.071	-.091	-.048	-.050	-.026	-.069	-.045	.850	
.400	-.005	-.106	-.006	-.141	-.053	-.120	-.070	-.077	-.096	-.036	-.119	-.019	-.088	-.040	.900	
.425	.011	-.123	-.007	-.152	-.041	-.121	-.079	-.079	-.097	-.036	-.115	-.016	-.092	-.023	.950	
$\alpha = 0$																
.0125															.0125	
.025	.111	.117	.090	.112	.095	.124	.104	.116	.120	.089	.141	.001	.135	.076	.025	
.050	.084	.098	.062	.130	.074	.114	.104	.107	.132	.074	.150	.072	.142	.054	.050	
.075	.065	.087	.046	.144	.075	.138	.087	.087	.126	.068	.154	.061	.117	.024	.075	
.100	.053	.080	.029	.155	.055	.134	.183	.184	.194	.126	.154	.063	.116	.014	.100	
.125	.040	.064	-.003	.154	.028	.160	.074	.082	.111	.049	.152	.009	.113	.002	.150	
.150	.020	.094	-.031	.149	-.001	.139	.063	.114	.107	.061	.122	-.010	.098	-.015	.200	
.175	.005	.108	-.046	.124	-.012	.100	.043	.097	.094	.064	.105	-.025	.048	-.043	.250	
.200	-.025	.102	-.039	.122	-.032	.084	.034	.046	.061	-.006	.076	-.006	.059	-.006	.300	
.225	-.052	.117	-.075	.125	-.043	.086	-.010	.011	.022	-.025	.030	-.016	.031	-.001	.400	
.250	-.066	.117	-.078	.082	-.056	.027	-.042	-.020	-.031	-.020	-.014	-.012	.019	-.006	.500	
.275	-.066	.117	-.078	.082	-.056	.027	-.042	-.020	-.031	-.020	-.014	-.012	.019	-.006	.550	
.300	-.052	.105	-.057	.038	-.078	-.030	-.065	-.039	-.064	-.020	-.054	-.007	-.035	-.005	.600	
.325	-.046	.052	-.015	.055	-.023	-.025	-.026	-.019	-.024	-.012	-.010	-.008	-.007	-.005	.650	
.350	-.025	.102	-.039	.122	-.032	.084	-.024	-.016	-.024	-.012	-.010	-.008	-.006	-.006	.700	
.375	-.052	.117	-.075	.125	-.043	.086	-.010	-.011	-.022	-.012	-.010	-.008	-.007	-.005	.750	
.400	-.071	.019	-.052	.088	-.074	-.084	-.094	-.092	-.123	-.020	-.111	-.001	-.084	-.011	.800	
.425	-.043	.076	-.040	.125	-.075	-.102	-.099	-.123	-.123	-.016	-.145	-.007	-.090	-.009	.900	
.450	-.028	.095	-.028	.135	-.065	-.096	-.103	-.118	-.118	-.015	-.136	-.007	-.099	-.019	.950	
$\alpha = 2$																
.0125															.0125	
.025	.050	.187	.020	.202	.030	.213	.038	.206	.049	.175	.065	.172	.061	.154	.025	
.050	.023	.157	.004	.233	.012	.205	.034	.193	.065	.181	.079	.156	.068	.127	.050	
.075	.014	.139	-.016	.235	.005	.236	.026	.180	.052	.155	.074	.140	.070	.052	.075	
.100	.002	.130	-.034	.234	-.006	.244	.019	.193	.057	.155	.083	.122	.081	.047	.100	
.125	-.015	.115	-.052	.231	-.025	.236	.008	.197	.048	.146	.081	.093	.081	.030	.150	
.150	-.022	.143	-.074	.213	-.057	.205	.003	.185	.041	.146	.070	.078	.079	.014	.200	
.175	-.045	.175	-.087	.193	-.066	.191	-.011	.164	.032	.124	.053	.032	.025	.012	.250	
.200	-.084	.174	-.102	.187	-.083	.143	-.025	.104	.010	.052	.038	.034	.039	.041	.300	
.225	-.098	.184	-.117	.193	-.088	.146	-.054	.066	-.018	.011	.002	.009	.023	.016	.400	
.250	-.120	.185	-.122	.147	-.100	.081	-.078	.016	-.066	.005	-.037	.009	.011	.011	.500	
.275	-.159	.121	-.102	.098	-.118	.014	-.108	-.018	-.091	.000	-.072	.002	-.035	.013	.600	
.300	-.135	.066	-.097	.028	-.112	-.026	-.124	-.032	-.131	-.020	-.102	.008	-.065	.017	.700	
.325	-.112	.020	-.088	-.049	-.108	-.068	-.132	-.040	-.140	-.010	-.140	.016	-.091	.013	.800	
.350	-.083	-.043	-.066	-.102	-.108	-.090	-.125	-.045	-.155	-.008	-.148	.019	-.119	.010	.900	
.																

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

x/c	Cp or $y/\frac{L}{2}$ of:														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
a = 4																
.0125	-0.007	.251	-0.036	.328	-0.025	.332	-0.023	.295	-0.010	.263	.004	.256	.001	.244	.0125	
.025	-0.011	.209	-0.052	.321	-0.042	.319	-0.027	.301	-0.001	.270	.018	.240	.017	.190	.025	
.0375	-0.013	.181	-0.072	.304	-0.050	.340	-0.034	.291	-0.010	.263	.014	.230	.032	.112	.075	
.100	-0.044	.181	-0.084	.307	-0.044	.324	-0.042	.294	-0.006	.266	.020	.217	.047	.087	.100	
.200	-0.063	.169	-0.098	.301	-0.075	.307	-0.054	.257	-0.010	.200	.021	.191	.057	.078	.150	
.300	-0.068	.200	-0.110	.278	-0.100	.267	-0.060	.241	-0.018	.206	.033	.138	.058	.055	.200	
.350	-0.087	.227	-0.123	.263	-0.113	.248	-0.070	.223	-0.031	.183	.019	.074	.014	.023	.300	
.400	-0.126	.239	-0.140	.261	-0.127	.205	-0.083	.156	-0.031	.095	.007	.063	.027	.046	.350	
.450	-0.131	.250	-0.148	.253	-0.132	.193	-0.108	.120	-0.054	.053	-0.018	.022	.008	.016	.400	
.500	-0.151	.243	-0.157	.206	-0.148	.121	-0.120	.052	-0.095	.022	-0.055	.022	-0.003	.027	.500	
.550	-0.189	.173	-0.140	.150	-0.143	.052	-0.148	.011	-0.123	.005	-0.091	.007	-0.051	.011	.600	
.600	-0.187	.111	-0.127	.070	-0.142	-0.004	-0.151	-0.016	-0.152	-0.009	-0.125	.009	-0.085	.011	.650	
.700	-0.167	-	-	-	-	-	-	-	-	-	-	-	-	-	.700	
.750	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.750	
.800	-0.148	.059	-0.113	-0.016	-0.134	-0.056	-0.171	-0.032	-0.158	-0.005	-0.156	.021	-0.123	.011	.800	
.850	-0.120	-0.012	-0.089	-0.068	-0.132	-0.076	-0.159	-0.047	-0.177	-0.004	-0.187	.020	-0.184	.007	.900	
.900	-0.094	-0.036	-0.077	-0.088	-0.125	-0.086	-0.159	-0.039	-0.178	-0.008	-0.160	.018	-0.232	.019	.950	
a = 6																
.0125	-0.061	.316	-0.087	.462	-0.079	.502	-0.075	.467	-0.065	.415	-0.053	.384	-0.057	.357	.0125	
.025	-0.080	.265	-0.102	.390	-0.092	.410	-0.078	.403	-0.057	.370	-0.040	.351	-0.033	.259	.025	
.0375	-0.077	.245	-0.119	.382	-0.100	.436	-0.079	.389	-0.063	.372	-0.041	.335	-0.010	.181	.075	
.100	-0.086	.236	-0.129	.378	-0.113	.403	-0.087	.376	-0.063	.357	-0.036	.317	-0.014	.148	.100	
.150	-0.100	.229	-0.142	.366	-0.123	.379	-0.098	.326	-0.060	.262	-0.034	.251	-0.029	.117	.150	
.200	-0.108	.260	-0.158	.342	-0.143	.326	-0.108	.304	-0.073	.262	-0.006	.184	-0.025	.060	.200	
.250	-0.120	.289	-0.162	.330	-0.157	.312	-0.115	.269	-0.082	.225	-0.011	.104	-0.010	.020	.250	
.300	-0.165	.305	-0.176	.322	-0.172	.275	-0.127	.211	-0.074	.133	-0.020	.085	-0.002	.037	.300	
.350	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.350	
.400	-0.161	.321	-0.180	.310	-0.177	.248	-0.145	.159	-0.084	.090	-0.053	.049	-0.022	.007	.400	
.450	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.450	
.500	-0.181	.302	-0.196	.269	-0.186	.168	-0.148	.087	-0.128	.043	-0.083	.025	-0.036	.006	.500	
.550	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.550	
.600	-0.218	.223	-0.176	.197	-0.167	.091	-0.173	.038	-0.150	.021	-0.120	.011	-0.080	-0.009	.600	
.650	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.650	
.700	-0.192	.159	-0.156	.103	-0.176	.027	-0.186	-0.001	-0.185	.001	-0.155	.014	-0.130	-0.004	.700	
.750	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.750	
.800	-0.181	.100	-0.141	.017	-0.161	-0.033	-0.189	-0.030	-0.202	-0.005	-0.190	.014	-0.199	.005	.800	
.850	-0.149	.016	-0.113	-0.041	-0.152	-0.070	-0.186	-0.042	-0.195	-0.006	-0.222	.017	-0.286	-0.001	.900	
.900	-0.120	-0.009	-0.099	-0.064	-0.142	-0.070	-0.176	-0.050	-0.208	-0.001	-0.228	.018	-0.328	.011	.950	
a = 8																
.0125	-0.123	.397	-0.140	.569	-0.136	.617	-0.130	.621	-0.122	.580	-0.113	.568	-0.114	.527	.0125	
.025	-0.129	.334	-0.158	.494	-0.144	.550	-0.150	.550	-0.110	.516	-0.099	.486	-0.079	.348	.025	
.0375	-0.126	.315	-0.168	.473	-0.154	.538	-0.132	.498	-0.116	.475	-0.099	.456	-0.040	.276	.075	
.100	-0.127	.303	-0.178	.468	-0.163	.495	-0.144	.468	-0.116	.447	-0.098	.408	-0.040	.215	.100	
.150	-0.142	.308	-0.190	.448	-0.176	.456	-0.149	.404	-0.116	.341	-0.093	.314	-0.011	.162	.150	
.200	-0.149	.346	-0.202	.430	-0.193	.404	-0.160	.377	-0.126	.320	-0.053	.235	-0.020	.091	.200	
.250	-0.155	.376	-0.202	.424	-0.204	.398	-0.172	.347	-0.137	.278	-0.046	.150	-0.054	.039	.250	
.300	-0.206	.396	-0.215	.400	-0.221	.356	-0.179	.292	-0.125	.195	-0.061	.134	-0.036	.051	.300	
.350	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.350	
.400	-0.196	.413	-0.212	.390	-0.225	.309	-0.190	.206	-0.126	.134	-0.086	.089	-0.052	.024	.400	
.450	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.450	
.500	-0.212	.370	-0.233	.337	-0.229	.225	-0.194	.138	-0.166	.085	-0.124	.053	-0.083	.006	.500	
.550	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.550	
.600	-0.244	.287	-0.213	.243	-0.198	.136	-0.212	.079	-0.195	.055	-0.153	.033	-0.153	-0.003	.600	
.650	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.650	
.700	-0.218	.213	-0.186	.147	-0.206	.063	-0.226	.028	-0.225	.020	-0.199	.027	-0.215	.002	.700	
.750	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.750	
.800	-0.212	.137	-0.171	.065	-0.194	-0.004	-0.211	-0.009	-0.245	.006	-0.239	.027	-0.280	.008	.800	
.850	-0.182	.047	-0.141	-0.007	-0.179	-0.041	-0.228	-0.032	-0.244	.012	-0.268	.026	-0.337	-0.001	.900	
.900	-0.152	.023	-0.128	-0.031	-0.168	-0.047	-0.226	-0.036	-0.232	.004	-0.275	.029	-0.381	.011	.950	
a = 10																
.0125	-0.166	.465	-0.188	.620	-0.184	.721	-0.178	.723	-0.168	.688	-0.150	.673	-0.158	.621	.0125	
.025	-0.170	.394	-0.198	.585	-0.186	.636	-0.171	.649	-0.151	.629	-0.136	.610	-0.116	.455	.050	
.0375	-0.165	.369	-0.208	.547	-0.180	.569	-0.181	.535	-0.157	.524	-0.132	.541	-0.074	.337	.100	
.100	-0.164	.358	-0.214	.547	-0.180	.569	-0.176	.535	-0.158	.524	-0.136	.573	-0.072	.223	.150	
.150	-0.177	.370	-0.225	.524	-0.216	.524	-0.198	.495	-0.157	.407	-0.132	.372	-0.053	.203	.200	
.200	-0.182	.425	-0.230	.508	-0.228	.494	-0.196	.465	-0.166	.380	-0.085	.292	-0.045	.125	.250	
.250	-0.182	.480	-0.218	.499	-0.236	.468	-0.210	.420	-0.177	.336	-0.078	.206	-0.095	.074	.300	
.300	-0.245	.479	-0.243	.474	-0.254	.425	-0.218	.348	-0.159	.248	-0.088	.190	-0.084	.084	.350	
.350	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.350	
.400	-0.228	.484	-0.239	.459	-0.260	.363	-0.222	.256	-0.159	.180	-0.119	.133	-0.117	.047	.400	
.450	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.450	
.500	-0.238	.421	-0.260	.382	-0.260	.266	-0.224	.182	-0.197	.123	-0.147	.087	-0.141	.022	.500	
.550	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.550	
.600	-0.266	.328	-0.243	.280	-0.230	.174	-0.244	.113	-0.224	.087	-0.182	.061	-0.226	.006	.600	
.650	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.650	
.700	-0.241	.249	-0.215	.179	-0.226	.097	-0.255	.056	-0.259	.037	-0.220	.044	-0.289	.008	.700	
.750	-	-	-	-	-	-	-	-	-	-	-	-	-	-	.750	
.800	-0.237	.164	-0.193	.093	-0.221	.030	-0.241	.017	-0.272	.020	-0.262	.043	-0.328	.013	.800	
.850	-0.212	.071	-0.164	.019	-0.205	.017	-0.246	-0.012	-0.271	.024	-0.301	.043	-0.365	.016	.900	
.900	-0.182	.051	-0.149	.004	-0.192	-0.022	-0.249	-0.037	-0.273	.024	-0.310	.043	-0.381	.020	.950	

TABLE III. - Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(a) $M = 1.61$ - Continued

x/c	Cp at $y/\frac{b}{2}$ of:														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 12^\circ$																
.0125	-214	.552	-238	.721	-244	.833	-242	.835	-221	.807	-219	.783	-226	.800	.0125	
.025	-216	.462	-243	.686	-234	.728	-223	.742	-197	.737	-194	.697	-172	.533	.050	
.050	-216	.462	-243	.686	-234	.728	-223	.742	-197	.737	-194	.697	-172	.533	.050	
.075	-205	.435	-249	.653	-237	.709	-210	.673	-197	.652	-186	.620	-171	.425	.075	
.100	-201	.422	-256	.633	-245	.655	-226	.630	-200	.625	-190	.541	-118	.318	.100	
.150	-211	.456	-264	.614	-246	.526	-234	.567	-200	.494	-186	.640	-119	.254	.150	
.200	-217	.555	-274	.620	-248	.544	-234	.534	-206	.457	-132	.356	-122	.167	.200	
.250	-212	.581	-274	.609	-275	.549	-252	.500	-216	.409	-129	.265	-163	.114	.250	
.300	-265	.579	-274	.560	-291	.493	-262	.410	-196	.313	-135	.245	-158	.121	.300	
.350	-256	.551	-267	.481	-299	.414	-261	.311	-192	.234	-165	.174	-209	.070	.350	
.400	-264	.470	-288	.425	-301	.309	-262	.231	-227	.171	-196	.120	-237	.033	.400	
.450	-264	.470	-288	.425	-301	.309	-262	.231	-227	.171	-196	.120	-237	.033	.450	
.500	-293	.368	-272	.225	-253	.211	-282	.153	-255	.123	-222	.080	-324	.009	.500	
.600	-266	.283	-244	.210	-253	.132	-300	.095	-288	.072	-262	.061	-362	.012	.600	
.700	-266	.283	-244	.210	-253	.132	-300	.095	-288	.072	-262	.061	-362	.012	.700	
.750	-265	.195	-217	.134	-253	.051	-281	.050	-313	.055	-303	.058	-380	.016	.800	
.800	-265	.195	-217	.134	-253	.051	-281	.050	-313	.055	-303	.058	-380	.016	.800	
.850	-245	.099	-191	.052	-235	.010	-277	.010	-328	.029	-336	.053	-389	.012	.900	
.900	-215	.085	-175	.024	-222	.004	-279	.009	-322	.030	-353	.051	-408	.024	.950	
$\alpha = 14^\circ$																
.0125	-249	.626	-277	.878	-293	.941	-298	.940	-285	.993	-272	.874	-282	.750	.0125	
.025	-252	.532	-274	.784	-276	.836	-264	.845	-252	.817	-246	.773	-227	.603	.050	
.050	-252	.532	-274	.784	-276	.836	-264	.845	-252	.817	-246	.773	-227	.603	.050	
.075	-237	.503	-277	.754	-271	.814	-257	.760	-244	.731	-225	.690	-152	.466	.075	
.100	-228	.486	-279	.742	-275	.750	-258	.714	-239	.698	-225	.612	-164	.384	.100	
.150	-232	.562	-290	.732	-285	.712	-265	.642	-243	.552	-228	.497	-171	.305	.150	
.200	-240	.680	-297	.710	-297	.652	-274	.601	-246	.508	-167	.408	-181	.206	.200	
.250	-230	.670	-303	.660	-302	.673	-283	.546	-255	.456	-166	.314	-219	.151	.250	
.300	-307	.646	-302	.625	-315	.545	-292	.445	-234	.354	-170	.264	-232	.154	.300	
.350	-276	.612	-291	.567	-325	.460	-290	.344	-228	.266	-194	.210	-276	.094	.400	
.400	-283	.511	-307	.464	-328	.348	-288	.262	-262	.204	-234	.153	-302	.057	.500	
.450	-283	.511	-307	.464	-328	.348	-288	.262	-262	.204	-234	.153	-302	.057	.500	
.500	-307	.406	-295	.361	-287	.249	-311	.183	-287	.141	-257	.104	-381	.028	.600	
.600	-282	.322	-267	.247	-271	.170	-327	.116	-220	.092	-295	.078	-406	.023	.700	
.700	-283	.224	-237	.163	-271	.081	-329	.066	-343	.065	-335	.071	-409	.029	.800	
.750	-283	.122	-198	.053	-243	.033	-293	.031	-358	.048	-367	.063	-420	.023	.900	
.800	-265	.133	-212	.082	-257	.043	-293	.031	-358	.048	-367	.063	-429	.035	.950	
$\alpha = 16^\circ$																
.0125	-295	.688	-333	.971	-368	1.051	-367	1.035	-360	.998	-341	.961	-343	.818	.0125	
.025	-293	.626	-325	.919	-336	.942	-327	.939	-307	.904	-299	.852	-292	.675	.050	
.050	-279	.604	-320	.875	-326	.911	-311	.851	-300	.814	-272	.766	-217	.533	.075	
.075	-237	.503	-327	.854	-326	.843	-308	.799	-285	.768	-243	.684	-144	.444	.100	
.100	-268	.671	-320	.854	-315	.843	-310	.799	-285	.768	-243	.684	-144	.444	.100	
.150	-258	.740	-326	.820	-310	.790	-311	.717	-289	.619	-273	.568	-231	.359	.150	
.200	-267	.753	-333	.782	-339	.723	-318	.663	-292	.558	-210	.410	-243	.256	.200	
.250	-251	.741	-339	.728	-345	.681	-327	.598	-297	.505	-205	.370	-288	.194	.250	
.300	-330	.711	-340	.683	-354	.596	-333	.503	-277	.403	-211	.334	-305	.193	.300	
.350	-303	.662	-321	.612	-362	.504	-332	.392	-271	.311	-236	.259	-349	.134	.400	
.400	-311	.548	-332	.503	-366	.389	-331	.301	-301	.244	-269	.196	-355	.086	.500	
.450	-311	.437	-325	.400	-330	.284	-347	.218	-330	.180	-294	.136	-426	.048	.600	
.500	-328	.437	-325	.400	-330	.284	-347	.218	-330	.180	-294	.136	-440	.044	.700	
.600	-320	.356	-297	.276	-305	.199	-364	.152	-360	.124	-334	.112	-440	.044	.750	
.700	-305	.294	-249	.205	-297	.111	-369	.094	-379	.092	-370	.103	-435	.053	.800	
.750	-311	.263	-264	.205	-297	.111	-369	.094	-379	.092	-370	.103	-435	.053	.850	
.800	-311	.173	-239	.120	-283	.081	-341	.067	-393	.087	-401	.094	-445	.043	.900	
.850	-294	.173	-223	.095	-269	.060	-331	.063	-398	.096	-410	.101	-450	.058	.950	
$\alpha = 18^\circ$																
.0125	-321	.823	-341	1.027	-432	1.140	-430	1.018	-355	.980	-357	.911	-355	.732	.050	
.025	-325	.902	-376	1.023	-390	1.031	-380	1.018	-362	.930	-345	.894	-329	.588	.075	
.050	-313	.798	-360	.969	-380	.992	-362	.930	-345	.873	-332	.840	-326	.500	.100	
.075	-297	.814	-362	.946	-372	.922	-355	.873	-332	.840	-326	.738	-281	.409	.150	
.100	-275	.834	-363	.896	-370	.858	-351	.784	-325	.686	-321	.624	-292	.303	.200	
.150	-284	.831	-367	.847	-376	.788	-355	.727	-328	.628	-321	.523	-314	.240	.250	
.200	-271	.809	-367	.792	-383	.742	-362	.659	-333	.563	-316	.460	-259	.385	.300	
.250	-341	.782	-370	.746	-390	.655	-370	.564	-316	.460	-259	.385	-371	.237	.350	
.300	-323	.719	-344	.663	-398	.556	-370	.451	-297	.366	-280	.303	-404	.173	.400	
.350	-334	.597	-351	.555	-399	.442	-363	.359	-330	.296	-313	.237	-391	.123	.500	
.400	-347	.495	-345	.464	-363	.346	-383	.278	-354	.236	-338	.179	-459	.086	.600	
.450	-347	.495	-345	.464	-363	.346	-383	.278	-354	.236	-338	.179	-459	.086	.650	
.500	-324	.421	-322	.336	-331	.255	-399	.210	-387	.177	-376	.151	-466	.082	.700	
.600	-324	.421	-322	.336	-331	.255	-399	.210	-387	.177	-376	.151	-466	.082	.750	
.700	-321	.325	-284	.269	-321	.177	-410	.154	-408	.141	-412	.139	-460	.085	.800	
.750	-319	.229	-271	.176	-283	.081	-341	.067	-393	.087	-401	.094	-445	.043	.850	
.800	-319	.229	-271	.176	-283	.081	-341	.067	-393	.087	-401	.094	-445	.043	.900	
.850	-319	.229	-271	.176	-283	.081	-341	.067	-393	.087	-401	.094	-445	.043	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING
(a) M = 1.61 - Concluded

x/c	Cp at $y/c = \frac{1}{2}$ of:														x/c	
	+.10		+.35		+.55		+.67		+.77		+.87		+.97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 20^\circ$																
.0125	-.371	.014	-.464	1.194	-.475	1.221	-.473	1.194	-.466	1.147	-.459	1.089	-.459	.930	.0125	
.025	-.357	.967	-.427	1.122	-.438	1.120	-.431	1.096	-.414	1.044	-.409	.982	-.410	.800	.025	
.050	-.348	.943	-.416	1.077	-.420	1.078	-.409	1.019	-.398	.972	-.381	.897	-.344	.655	.075	
.075	-.348	.937	-.416	1.077	-.420	1.078	-.409	1.019	-.398	.972	-.381	.897	-.344	.655	.075	
.100	-.316	.937	-.405	1.038	-.415	1.010	-.397	.969	-.385	.913	-.374	.812	-.337	.573	.100	
.150	-.298	.937	-.400	.980	-.405	.941	-.388	.863	-.369	.767	-.361	.697	-.352	.475	.150	
.200	-.298	.922	-.400	.933	-.409	.867	-.389	.800	-.369	.703	-.304	.598	-.360	.367	.200	
.250	-.293	.893	-.402	.870	-.413	.816	-.393	.729	-.373	.636	-.287	.494	-.400	.305	.250	
.300	-.356	.860	-.401	.819	-.420	.725	-.399	.630	-.362	.533	-.293	.462	-.420	.299	.300	
.350	-.344	.786	-.368	.725	-.429	.623	-.399	.506	-.335	.420	-.315	.357	-.445	.216	.400	
.400	-.357	.658	-.372	.619	-.433	.508	-.392	.410	-.364	.350	-.346	.293	-.408	.174	.500	
.450	-.367	.549	-.367	.523	-.387	.408	-.407	.333	-.391	.297	-.377	.242	-.484	.136	.600	
.500	-.367	.549	-.367	.523	-.387	.408	-.407	.333	-.391	.297	-.377	.242	-.484	.136	.550	
.550	-.367	.549	-.367	.523	-.387	.408	-.407	.333	-.391	.297	-.377	.242	-.484	.136	.600	
.600	-.367	.549	-.367	.523	-.387	.408	-.407	.333	-.391	.297	-.377	.242	-.484	.136	.550	
.650	-.344	.484	-.345	.404	-.357	.313	-.426	.265	-.412	.227	-.409	.206	-.484	.129	.700	
.700	-.351	.388	-.302	.334	-.348	.243	-.440	.215	-.434	.203	-.445	.198	-.484	.136	.750	
.750	-.351	.388	-.302	.334	-.348	.243	-.440	.215	-.434	.203	-.445	.198	-.484	.136	.800	
.800	-.346	.308	-.278	.252	-.339	.212	-.403	.197	-.448	.210	-.463	.206	-.489	.144	.850	
.850	-.327	.305	-.264	.241	-.300	.208	-.346	.203	-.452	.221	-.476	.224	-.489	.168	.950	
.900	-.346	.308	-.278	.252	-.339	.212	-.403	.197	-.448	.210	-.463	.206	-.489	.144	.900	
.950	-.327	.305	-.264	.241	-.300	.208	-.346	.203	-.452	.221	-.476	.224	-.489	.168	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(b) $M = 2.01$

x/c	C_p at $y/b = \frac{1}{2}$ of:														x/c					
	+10		+35		+55		+67		+77		+87		+97							
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower						
$\alpha = -20$																				
.0125	.803	-+230	.771	-+234	.829	-+243	.894	-+246	.962	-+244	1.068	-+251	1.030	-+261	.025					
.025	.720	-+240	.709	-+236	.757	-+232	.764	-+247	.891	-+242	.981	-+245	.876	-+257	.050					
.050	.690	-+233	.654	-+237	.704	-+238	.745	-+232	.849	-+242	.920	-+242	.752	-+239	.075					
.075	.669	-+224	.606	-+237	.649	-+238	.725	-+247	.833	-+242	.877	-+246	.688	-+248	.100					
.100	.669	-+221	.547	-+235	.597	-+242	.688	-+251	.788	-+254	.774	-+253	.612	-+270	.150					
.150	.595	-+219	.492	-+235	.560	-+246	.673	-+252	.739	-+245	.704	-+253	.553	-+286	.200					
.200	.590	-+219	.492	-+235	.549	-+246	.650	-+246	.681	-+253	.662	-+250	.504	-+294	.250					
.250	.541	-+193	.488	-+235	.539	-+246	.608	-+241	.627	-+259	.594	-+246	.467	-+300	.300					
.300	.476	-+188	.434	-+240	.533	-+246														
.350																.350				
.400	.421	-+227	.402	-+229	.521	-+250	.527	-+253	.549	-+253	.526	-+257	.414	-+307	.400					
.450	.349	-+209	.415	-+223	.446	-+255	.475	-+257	.491	-+243	.460	-+267	.373	-+301	.500					
.500	.316	-+226	.396	-+230	.422	-+265	.414	-+242	.428	-+245	.407	-+275	.300	-+303	.600					
.550	.290	-+204	.482	-+229	.525	-+231	.520	-+241	.522	-+241	.518	-+254	.504	-+285	.700					
.600	.264	-+196	.428	-+229	.488	-+235	.598	-+241	.580	-+238	.646	-+245	.457	-+297	.800					
.650	.239	-+172	.408	-+225	.474	-+235	.579	-+232	.528	-+245	.588	-+247	.424	-+305	.900					
.700	.217	-+169	.382	-+228	.457	-+238	.551	-+235	.574	-+250	.543	-+242			.300					
.750	.191	-+191	.354	-+209	.388	-+248	.431	-+246	.444	-+229	.410	-+263	.334	-+306	.550					
.800	.177	-+191	.354	-+221	.365	-+259	.360	-+231	.377	-+231	.360	-+272	.265	-+312	.600					
.850	.152	-+195	.337	-+247	.338	-+259	.326	-+224	.330	-+240	.298	-+279	.206	-+309	.750					
.900	.136	-+214	.321	-+261	.305	-+237	.293	-+225	.284	-+243	.236	-+288	.149	-+302	.800					
.950	.114	-+236	.330	-+274	.282	-+243	.256	-+231	.233	-+259	.182	-+286	.091	-+287	.900					
	.363	-+248	.351	-+255	.284	-+248	.242	-+231	.219	-+257	.163	-+286	.061	-+278	.950					
$\alpha = -18$																				
.0125	.718	-+218	.711	-+225	.764	-+225	.820	-+229	.876	-+222	.969	-+234	.958	-+242	.025					
.025	.634	-+225	.618	-+225	.677	-+229	.718	-+231	.810	-+226	.894	-+234	.808	-+244	.050					
.050	.619	-+215	.574	-+225	.630	-+227	.668	-+221	.764	-+231	.844	-+238	.689	-+220	.075					
.075	.590	-+208	.537	-+225	.583	-+229	.648	-+238	.745	-+233	.812	-+241	.628	-+234	.100					
.100	.534	-+204	.482	-+229	.525	-+231	.520	-+241	.722	-+241	.718	-+254	.556	-+259	.150					
.150	.454	-+196	.428	-+229	.488	-+235	.598	-+241	.580	-+238	.646	-+245	.504	-+285	.200					
.200	.398	-+172	.408	-+225	.457	-+235	.579	-+232	.528	-+245	.588	-+247	.457	-+297	.250					
.250	.348	-+172	.408	-+225	.474	-+235	.579	-+232	.574	-+250	.543	-+242	.424	-+305	.300					
.300	.282	-+169	.382	-+228	.457	-+238	.551	-+235												
.350																.400				
.400	.372	-+205	.343	-+215	.457	-+240	.473	-+241	.497	-+244	.471	-+252	.370	-+314	.450					
.450	.306	-+191	.354	-+209	.388	-+248	.431	-+246	.444	-+229	.410	-+263	.334	-+306	.500					
.500	.268	-+210	.351	-+221	.365	-+259	.360	-+231	.377	-+231	.360	-+272	.265	-+312	.600					
.550	.226	-+199	.337	-+247	.338	-+259	.326	-+224	.330	-+240	.298	-+279	.206	-+309	.750					
.600	.190	-+214	.321	-+261	.305	-+237	.293	-+225	.284	-+243	.236	-+288	.149	-+302	.800					
.650	.159	-+236	.330	-+274	.282	-+243	.256	-+231	.233	-+259	.182	-+286	.091	-+287	.900					
.700	.133	-+248	.351	-+255	.284	-+248	.242	-+231	.219	-+257	.163	-+286	.061	-+278	.950					
$\alpha = -16$																				
.0125	.618	-+196	.636	-+204	.702	-+200	.747	-+206	.777	-+206	.851	-+206	.861	-+207	.025					
.025	.543	-+200	.535	-+207	.587	-+209	.630	-+212	.706	-+212	.787	-+212	.746	-+218	.050					
.050	.534	-+195	.499	-+200	.556	-+200	.601	-+201	.680	-+212	.756	-+217	.617	-+196	.075					
.075	.514	-+185	.461	-+200	.512	-+204	.565	-+216	.660	-+214	.735	-+222	.554	-+207	.100					
.100	.465	-+177	.414	-+200	.452	-+204	.565	-+219	.691	-+224	.768	-+236	.495	-+234	.150					
.150	.396	-+157	.355	-+173	.395	-+181	.452	-+181	.507	-+220	.585	-+233	.449	-+266	.200					
.200	.379	-+151	.306	-+171	.357	-+181	.455	-+188	.541	-+197	.531	-+214	.407	-+235	.250					
.250	.366	-+126	.283	-+171	.342	-+183	.439	-+190	.511	-+205	.476	-+211	.363	-+261	.250					
.300	.322	-+130	.264	-+171	.317	-+187	.423	-+196	.461	-+211	.436	-+201	.336	-+272	.300					
.350																.350				
.400	.257	-+157	.235	-+155	.318	-+191	.356	-+200	.377	-+204	.365	-+204	.285	-+300	.400					
.450	.213	-+152	.225	-+147	.285	-+204	.321	-+211	.317	-+187	.303	-+213	.246	-+292	.500					
.500	.170	-+169	.243	-+185	.255	-+215	.265	-+197	.273	-+185	.258	-+226	.177	-+301	.600					
.550	.149	-+164	.238	-+208	.237	-+219	.228	-+185	.223	-+193	.212	-+235	.129	-+300	.700					
.600	.124	-+178	.231	-+228	.215	-+205	.200	-+182	.182	-+197	.147	-+243	.076	-+283	.800					
.650	.100	-+178	.226	-+228	.248	-+192	-+212	.166	-+184	.142	-+205	.104	-+249	.025	-+251	.900				
.700	.075	-+178	.221	-+221	.242	-+196	-+212	.157	-+188	.133	-+205	.081	-+243	.002	-+241	.950				
$\alpha = -14$																				
.0125	.540	-+177	.539	-+182	.596	-+177	.647	-+178	.677	-+180	.739	-+181	.765	-+186	.025					
.025	.473	-+178	.475	-+175	.507	-+185	.541	-+185	.604	-+183	.673	-+190	.635	-+193	.050					
.050	.437	-+174	.434	-+175	.493	-+179	.532	-+178	.600	-+188	.671	-+193	.554	-+174	.075					
.075	.411	-+167	.393	-+175	.452	-+181	.507	-+189	.584	-+190	.653	-+199	.509	-+178	.100					
.100	.379	-+167	.355	-+173	.395	-+181	.472	-+189	.560	-+198	.605	-+214	.451	-+200	.150					
.150	.336	-+157	.355	-+173	.395	-+181	.452	-+181	.495	-+190	.511	-+205	.476	-+235	.200					
.200	.379	-+151	.306	-+171	.357	-+181	.455	-+188	.541	-+197	.531	-+214	.407	-+235	.250					
.250	.366	-+126	.283	-+171	.342	-+183	.439	-+190	.511	-+205	.476	-+211	.363	-+261	.250					
.300	.322	-+130	.264	-+171	.317	-+187	.423	-+196	.461	-+211	.436	-+201	.336	-+272	.300					
.350																.350				
.400	.257	-+157	.235	-+155	.318	-+191	.356	-+200	.377	-+204	.365	-+204	.285	-+300	.400					
.450	.213	-+152	.225	-+147	.285	-+204	.321	-+211	.317	-+187	.303	-+213	.246	-+292	.500					

TABLE III.—Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

X/C	Cp of y/ $\frac{1}{2}$ of:														X/C	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
a = -12																
.0125																
.025	.466	-.155	.469	-.153	.490	-.153	.522	-.156	.545	-.158	.599	-.163	.624	-.166	.0225	
.050	.406	-.155	.400	-.147	.439	-.161	.461	-.162	.520	-.163	.574	-.168	.544	-.169	.050	
.075	.366	-.148	.371	-.147	.418	-.151	.453	-.157	.518	-.166	.577	-.172	.480	-.151	.075	
.100	.365	-.142	.339	-.147	.394	-.155	.435	-.168	.507	-.169	.568	-.178	.447	-.152	.100	
.125	.321	-.132	.293	-.147	.340	-.155	.410	-.168	.490	-.177	.542	-.193	.398	-.171	.125	
.200	.310	-.125	.250	-.143	.301	-.155	.386	-.168	.478	-.177	.479	-.193	.359	-.206	.200	
.250	.304	-.104	.225	-.143	.284	-.155	.368	-.166	.453	-.187	.430	-.191	.321	-.231	.250	
.300	.264	-.117	.209	-.143	.264	-.161	.355	-.173	.410	-.191	.385	-.179	.295	-.244	.300	
.350																
.400	.204	-.134	.185	-.121	.260	-.166	.299	-.177	.328	-.191	.319	-.182	.242	-.282	.400	
.450																
.500	.159	-.129	.176	-.121	.235	-.181	.254	-.190	.263	-.169	.259	-.188	.210	-.278	.500	
.550																
.600	.118	-.150	.195	-.166	.212	-.193	.205	-.177	.222	-.165	.207	-.200	.139	-.289	.600	
.650																
.700	.100	-.146	.195	-.185	.192	-.200	.181	-.165	.180	-.172	.164	-.208	.092	-.286	.700	
.750																
.800	.148	-.162	.189	-.208	.177	-.200	.152	-.160	.140	-.172	.111	-.217	.049	-.268	.800	
.850																
.900	.179	-.193	.186	-.232	.149	-.200	.125	-.165	.104	-.178	.065	-.219	.003	-.229	.900	
.950	.206	-.206	.197	-.227	.154	-.200	.114	-.172	.090	-.172	.045	-.212	.028	-.214	.950	
a = -10																
.0125																
.025	.393	-.124	.366	-.130	.393	-.120	.420	-.130	.438	-.130	.472	-.131	.482	-.138	.0225	
.050	.348	-.122	.334	-.121	.361	-.129	.380	-.136	.435	-.131	.468	-.139	.456	-.143	.050	
.075	.323	-.121	.303	-.121	.342	-.122	.374	-.136	.431	-.136	.480	-.144	.392	-.131	.075	
.100	.297	-.115	.280	-.120	.316	-.129	.362	-.145	.427	-.141	.474	-.147	.374	-.125	.100	
.125	.261	-.106	.235	-.119	.275	-.122	.342	-.145	.418	-.149	.468	-.166	.343	-.141	.125	
.200	.292	-.104	.198	-.112	.241	-.125	.320	-.146	.406	-.150	.419	-.168	.311	-.186	.200	
.250	.257	-.072	.167	-.113	.216	-.126	.307	-.138	.389	-.158	.374	-.166	.273	-.197	.250	
.300	.211	-.104	.153	-.112	.202	-.136	.281	-.146	.357	-.166	.336	-.155	.253	-.212	.300	
.350																
.400	.160	-.107	.131	-.086	.187	-.136	.242	-.149	.280	-.166	.267	-.154	.202	-.258	.400	
.450																
.500	.115	-.105	.126	-.100	.176	-.149	.191	-.165	.213	-.145	.212	-.160	.170	-.265	.500	
.550																
.600	.073	-.122	.141	-.145	.157	-.164	.156	-.152	.166	-.142	.159	-.167	.100	-.277	.600	
.650																
.700	.057	-.123	.148	-.165	.134	-.179	.130	-.140	.131	-.145	.116	-.174	.052	-.275	.700	
.750																
.800	.100	-.140	.142	-.194	.123	-.185	.104	-.137	.098	-.144	.070	-.181	.011	-.254	.800	
.850																
.900	.133	-.173	.136	-.218	.111	-.185	.080	-.145	.057	-.145	.029	-.184	.034	-.206	.900	
.950	.154	-.181	.152	-.219	.110	-.184	.062	-.152	.051	-.140	.010	-.179	.056	-.191	.950	
a = -8																
.0125																
.025	.320	-.086	.304	-.089	.321	-.082	.344	-.091	.350	-.094	.377	-.101	.379	-.104	.0225	
.050	.292	-.088	.273	-.081	.296	-.094	.316	-.098	.352	-.097	.377	-.107	.372	-.110	.050	
.075	.265	-.087	.246	-.078	.282	-.083	.307	-.098	.350	-.104	.388	-.112	.310	-.097	.075	
.100	.232	-.077	.224	-.077	.262	-.090	.303	-.106	.350	-.108	.387	-.119	.301	-.097	.100	
.125	.207	-.073	.186	-.072	.226	-.084	.282	-.103	.347	-.117	.389	-.136	.289	-.106	.125	
.200	.203	-.073	.153	-.067	.194	-.084	.272	-.106	.338	-.121	.360	-.141	.266	-.126	.200	
.250	.217	-.031	.127	-.067	.178	-.085	.242	-.102	.325	-.128	.317	-.141	.233	-.158	.250	
.300	.164	-.075	.104	-.071	.156	-.100	.239	-.113	.300	-.137	.286	-.133	.212	-.163	.300	
.350																
.400	.120	-.076	.092	-.051	.139	-.097	.199	-.122	.231	-.140	.225	-.125	.170	-.215	.400	
.450																
.500	.077	-.075	.086	-.068	.123	-.117	.153	-.136	.173	-.120	.173	-.129	.138	-.227	.500	
.550																
.600	.044	-.092	.097	-.117	.112	-.136	.107	-.131	.124	-.115	.119	-.133	.073	-.250	.600	
.650																
.700	.029	-.092	.111	-.136	.102	-.158	.087	-.117	.085	-.115	.074	-.137	.026	-.244	.700	
.750																
.800	.058	-.114	.107	-.165	.093	-.166	.067	-.117	.058	-.112	.026	-.137	.018	-.220	.800	
.850																
.900	.086	-.142	.101	-.194	.079	-.163	.051	-.127	.026	-.115	.008	-.138	.054	-.177	.900	
.950	.108	-.147	.093	-.194	.081	-.163	.051	-.135	.018	-.114	.020	-.132	.069	-.157	.950	
a = -6																
.0125																
.025	.260	-.049	.238	-.053	.253	-.048	.265	-.057	.271	-.063	.295	-.068	.292	-.066	.0225	
.050	.234	-.050	.212	-.045	.231	-.055	.244	-.063	.272	-.066	.294	-.076	.292	-.079	.050	
.075	.197	-.050	.199	-.042	.216	-.052	.238	-.064	.270	-.073	.303	-.082	.239	-.069	.075	
.100	.160	-.045	.179	-.042	.207	-.051	.227	-.067	.272	-.077	.308	-.093	.232	-.072	.100	
.125	.161	-.044	.174	-.034	.171	-.049	.217	-.073	.272	-.082	.312	-.108	.226	-.065	.125	
.200	.148	-.045	.104	-.028	.144	-.049	.210	-.075	.270	-.091	.295	-.115	.208	-.085	.200	
.250	.165	-.000	.093	-.028	.125	-.046	.195	-.071	.257	-.094	.260	-.118	.189	-.114	.160	
.300	.110	-.061	.060	-.030	.106	-.065	.178	-.081	.242	-.111	.234	-.140	.169	-.108	.200	
.350																
.400	.075	-.044	.053	-.010	.090	-.058	.144	-.092	.181	-.115	.176	-.101	.124	-.148	.400	
.450																
.500	.038	-.043	.039	-.038	.068	-.080	.104	-.108	.124	-.098	.129	-.101	.127	-.176	.500	
.550																
.600	-.006	-.062	.053	-.081	.068	-.105	.018	-.107	.076	-.087	.076	-.101	.041	-.126	.600	
.650																
.700	-.000	-.070	.067	-.107	.061	-.133	.004	-.109	.011	-.089	.031	-.102	-.003	-.194	.700	
.750																
.800	.018	-.090	.067	-.142	.050	-.152	.027	-.101	.014	-.086	.020	-.101	.042	-.183	.800	
.850																
.900	.043	-.116	.066	-.169	.038	-.151	.004	-.116	.013	-.083	.045	-.097	.078	-.143	.900	
.950	.068	-.125	.072	-.169	.040	-.150	.003	-.114	.020	-.089	.047	-.091	.091	-.126	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING
(b) $M = 2.01$ - Continued

x/c	Cp at $y/\frac{c}{2}$ of :														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -4$																
.0125	.208	.004	.182	-.004	.198	.001	.208	-.015	.211	-.021	.232	-.027	.229	-.032	.025	
.025	.186	-.005	.164	.003	.179	-.015	.192	-.021	.210	-.024	.229	-.033	.226	-.045	.050	
.050	.141	-.010	.143	-.001	.165	-.007	.182	-.026	.208	-.031	.237	-.038	.184	-.045	.075	
.075	.133	-.007	.122	.004	.151	-.007	.179	-.032	.215	-.036	.243	-.053	.181	-.050	.100	
.100	.113	-.007	.096	.016	.126	-.003	.166	-.030	.212	-.052	.247	-.070	.179	-.052	.150	
.125	.113	-.007	.096	.016	.126	-.003	.166	-.030	.212	-.052	.247	-.070	.179	-.052	.150	
.150	.101	-.007	.070	.023	.130	-.002	.157	-.034	.212	-.053	.249	-.080	.173	-.063	.200	
.175	.123	.035	.069	.024	.084	-.002	.146	-.026	.202	-.052	.214	-.090	.148	-.048	.250	
.200	.071	-.030	.023	.016	.068	-.024	.133	-.044	.190	-.076	.190	-.081	.133	-.074	.350	
.225	.036	-.007	.015	.034	.053	-.015	.100	-.055	.137	-.080	.144	-.070	.099	-.102	.450	
.250	.003	-.007	.008	-.003	.025	-.042	.065	-.072	.089	-.068	.101	-.064	.076	-.171	.500	
.275	.022	-.026	.021	-.047	.031	-.070	.027	-.084	.044	-.055	.052	-.063	.018	-.147	.600	
.300	.034	-.024	.032	-.071	.025	-.109	.000	-.077	.002	-.059	.009	-.061	-.023	-.148	.700	
.325	.015	-.054	.039	-.108	.016	-.129	-.007	-.084	-.028	-.054	-.035	-.058	-.055	-.136	.800	
.350	.003	-.085	.035	-.144	.012	-.127	-.020	-.092	-.039	-.061	-.070	-.052	-.084	-.107	.900	
.375	.021	-.097	.045	-.147	.012	-.126	-.024	-.090	-.047	-.062	-.080	-.049	-.093	-.088	.950	
$\alpha = -2$																
.0125	.141	.030	.120	.029	.133	.039	.143	.021	.152	.021	.166	.013	.163	.014	.025	
.025	.127	.030	.103	.043	.120	.019	.124	.011	.153	.017	.162	.005	.163	-.004	.050	
.050	.090	.018	.086	.040	.107	.030	.116	.009	.151	.008	.171	-.001	.133	-.015	.075	
.075	.078	.021	.067	.041	.091	.030	.114	.000	.157	.003	.180	-.016	.130	-.025	.100	
.100	.063	.018	.043	.058	.074	.040	.105	.000	.159	.013	.180	-.022	.130	-.032	.150	
.125	.051	.019	.021	.067	.052	.040	.097	.000	.157	.014	.177	-.049	.128	-.062	.200	
.150	.076	.059	.003	.061	.036	.038	.088	.025	.147	.010	.160	-.063	.113	-.059	.250	
.175	.021	-.001	.018	.052	.020	.019	.077	-.001	.135	.039	.138	-.058	.101	-.046	.350	
.200	-.008	.024	-.030	.071	.006	.018	.042	-.009	.095	-.049	.099	-.050	.072	-.057	.400	
.225	-.039	.022	-.036	.022	-.021	-.015	.018	-.036	.050	-.046	.061	-.036	.057	-.056	.500	
.250	-.061	.005	-.025	-.015	-.016	-.047	-.021	-.058	.067	-.036	.012	-.030	.000	-.068	.600	
.275	-.066	-.022	-.010	-.047	-.018	-.093	-.047	-.059	-.028	-.036	-.024	-.027	-.038	-.069	.700	
.300	-.051	-.033	-.003	-.087	-.023	-.122	-.052	-.068	-.062	-.029	-.064	-.028	-.063	-.065	.750	
.325	-.032	-.066	.001	-.124	-.030	-.128	-.062	-.071	-.079	-.042	-.095	-.018	-.078	-.050	.800	
.350	-.016	-.084	.008	-.129	-.026	-.123	-.056	-.070	-.077	-.042	-.107	-.015	-.088	-.033	.900	
$\alpha = 0$																
.0125	.089	.080	.074	.082	.079	.090	.091	.072	.094	.067	.110	.064	.104	.061	.025	
.025	.021	.080	.057	.095	.057	.071	.075	.063	.093	.063	.109	.053	.107	.042	.050	
.050	.049	.058	.043	.092	.059	.043	.067	.054	.093	.053	.119	.047	.091	.014	.075	
.075	.042	.057	.028	.098	.045	.085	.064	.052	.096	.048	.117	.030	.090	-.001	.100	
.100	.029	.048	.003	.117	.029	.098	.058	.051	.099	.026	.124	.011	.093	-.012	.150	
.125	.016	.059	-.017	.127	.011	.094	.046	.051	.097	.030	.125	-.007	.076	-.023	.200	
.150	.032	.080	-.031	.112	-.005	.094	.041	.072	.093	.036	.112	-.028	.085	-.043	.250	
.175	-.016	.049	-.050	.096	-.020	.072	.034	.044	.080	.004	.100	-.025	.077	-.026	.300	
.200	-.042	.067	-.062	.116	-.034	.067	.003	.034	.053	-.012	.064	-.024	.050	-.028	.400	
.225	-.066	.063	-.066	.061	-.056	.023	-.021	-.001	.013	-.021	.035	-.011	.035	-.017	.500	
.250	-.085	.042	-.055	.025	-.059	-.016	-.051	-.032	-.028	-.022	-.007	-.005	-.014	-.012	.600	
.275	-.070	.013	-.039	-.013	-.055	-.059	-.077	-.052	-.061	-.023	-.047	-.000	-.046	-.002	.700	
.300	-.051	-.033	-.003	-.087	-.023	-.122	-.052	-.068	-.062	-.029	-.084	-.028	-.063	-.065	.750	
.325	-.032	-.066	.001	-.124	-.030	-.128	-.062	-.071	-.079	-.042	-.107	-.015	-.088	-.033	.800	
.350	-.016	-.084	.008	-.129	-.026	-.123	-.056	-.070	-.077	-.042	-.115	-.008	-.087	-.009	.900	
$\alpha = 2$																
.0125	.042	.134	.129	.142	.036	.150	.047	.129	.045	.117	.059	.106	.053	.108	.025	
.025	.034	.133	.015	.155	.029	.130	.033	.118	.048	.113	.059	.096	.053	.089	.050	
.050	.010	.102	.002	.158	.020	.138	.025	.108	.047	.098	.064	.086	.064	.074	.075	
.075	.005	.095	-.010	.163	.009	.142	.021	.113	.047	.095	.064	.080	.055	.074	.100	
.100	-.008	.087	-.033	.183	-.010	.161	.017	.106	.052	.075	.067	.088	.061	.002	.150	
.125	-.020	.095	-.051	.184	-.022	.153	.006	.113	.048	.077	.072	.027	.061	-.010	.200	
.150	.006	.118	-.058	.164	-.036	.158	.001	.124	.045	.083	.069	-.001	.056	-.019	.250	
.175	-.069	.099	-.075	.149	-.052	.131	-.006	.095	.037	.052	.058	-.001	.052	-.013	.300	
.200	-.076	.118	-.086	.168	-.064	.114	-.032	.073	-.014	.034	.031	-.005	.029	-.006	.400	
.225	-.099	.117	-.098	.116	-.084	.070	-.056	.035	-.019	.005	-.001	.000	.024	-.007	.500	
.250	-.118	.082	-.086	.074	-.080	.024	-.081	-.005	.056	-.006	.033	-.002	-.023	.013	.600	
.275	-.110	.060	-.070	.034	-.079	-.019	-.101	-.037	-.090	-.012	-.073	-.006	-.041	-.017	.700	
.300	-.063	-.039	-.029	-.097	-.057	-.112	-.052	-.057	-.110	-.052	-.109	-.012	-.033	-.002	.750	
.325	-.101	.037	-.063	-.016	-.083	-.063	-.110	-.052	-.109	-.012	-.103	-.009	-.066	-.010	.800	
.350	-.098	-.001	-.052	-.061	-.087	-.091	-.110	-.052	-.132	-.012	-.120	-.013	-.095	-.015	.900	
.375	-.079	-.022	-.041	-.060	-.078	-.091	-.117	-.052	-.132	-.010	-.128	-.010	-.109	-.022	.950	

TABLE III.- Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(b) $M = 2.01$ - Continued

x/c	C_p at $y/c = \frac{1}{2}$ of:														x/c	
	+10		+35		+55		+67		+77		+87		+97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 4$																
.0125	.002	.194	-.014	.198	-.013	.213	.006	.188	.005	.170	.017	.165	.010	.165	.0125	
.025	-.008	.187	+.025	.211	-.019	.194	-.005	.180	-.005	.169	.013	.154	.013	.144	.050	
.050	-.030	.139	-.050	.227	-.039	.215	-.013	.167	-.006	.150	.023	.141	.019	.089	.075	
.075	-.022	.149	-.037	.218	-.027	.205	-.013	.167	-.006	.150	.023	.141	.019	.089	.075	
.100	-.030	.139	-.050	.227	-.039	.215	-.013	.175	-.007	.149	.021	.123	.025	.056	.100	
.150	-.035	.131	-.066	.246	-.052	.244	-.022	.168	-.014	.149	.026	.099	.037	.031	.150	
.200	-.044	.139	-.080	.255	-.077	.228	-.030	.164	-.006	.131	.030	.077	.037	.032	.200	
.250	-.022	.168	-.094	.221	-.078	.225	-.037	.187	-.000	.145	.035	.045	.035	.014	.250	
.300	-.106	.147	-.107	.193	-.090	.183	-.004	.152	-.004	.107	.029	.041	.034	.006	.350	
.350	-.102	.164	-.118	.214	-.105	.163	-.066	.121	-.019	.073	.008	.031	.014	.018	.400	
.400	-.102	.164	-.118	.214	-.105	.163	-.066	.121	-.019	.073	.008	.031	.014	.018	.450	
.450	-.124	.160	-.126	.160	-.125	.115	-.083	.073	-.045	.033	-.020	.023	.019	.030	.500	
.500	-.137	.128	-.119	.110	-.122	.064	-.105	.032	-.081	.013	-.049	.013	-.026	.022	.550	
.550	-.128	.105	-.104	.064	-.114	.013	-.122	-.011	-.111	-.001	-.086	.013	-.050	.023	.600	
.600	-.128	.105	-.104	.064	-.114	.013	-.122	-.011	-.111	-.001	-.086	.013	-.050	.023	.650	
.650	-.128	.105	-.104	.064	-.114	.013	-.122	-.011	-.111	-.001	-.086	.013	-.050	.023	.700	
.700	-.128	.105	-.104	.064	-.114	.013	-.122	-.011	-.111	-.001	-.086	.013	-.050	.023	.750	
.750	-.124	.077	-.095	.013	-.118	-.033	-.134	-.030	-.129	-.003	-.110	.016	-.079	.017	.800	
.800	-.116	.030	-.084	-.039	-.110	-.067	-.127	-.037	-.143	-.007	-.132	.019	-.108	.012	.850	
.850	-.106	.006	-.072	-.041	-.101	-.071	-.129	-.037	-.147	-.003	-.141	.018	-.134	.017	.900	
.900	-.106	.006	-.072	-.041	-.101	-.071	-.129	-.037	-.147	-.003	-.141	.018	-.134	.017	.950	
$\alpha = 6$																
.0125	-.039	.253	-.056	.267	-.050	.273	-.036	.245	-.035	.226	-.027	.226	-.030	.222	.025	
.025	-.046	.239	-.064	.280	-.054	.250	-.050	.234	-.037	.221	-.032	.212	-.027	.199	.050	
.050	-.059	.188	-.076	.286	-.058	.265	-.054	.226	-.035	.201	-.026	.201	-.008	.128	.075	
.075	-.062	.192	-.083	.307	-.070	.281	-.059	.234	-.036	.198	-.025	.180	-.001	.083	.100	
.100	-.069	.171	-.103	.311	-.083	.305	-.065	.227	-.032	.181	-.022	.151	-.015	.053	.150	
.150	-.076	.179	-.118	.293	-.097	.294	-.066	.252	-.036	.186	-.014	.129	-.020	.025	.200	
.200	-.076	.179	-.118	.293	-.097	.294	-.066	.252	-.036	.186	-.014	.129	-.020	.025	.200	
.250	-.058	.210	-.124	.266	-.105	.276	-.076	.250	-.043	.198	-.003	.091	-.007	-.003	.250	
.300	-.137	.199	-.132	.246	-.119	.229	-.082	.200	-.046	.154	-.004	.092	-.010	.018	.350	
.350	-.129	.214	-.142	.271	-.134	.207	-.099	.162	-.057	.103	-.021	.062	-.005	.023	.400	
.400	-.149	.217	-.146	.209	-.150	.157	-.117	.114	-.078	.050	-.044	.040	-.002	.020	.450	
.450	-.155	.176	-.143	.160	-.144	.099	-.133	.060	-.107	.029	-.072	.023	-.042	.011	.500	
.500	-.150	.147	-.129	.109	-.138	.043	-.150	.017	-.138	.005	-.102	.022	-.073	.007	.550	
.550	-.146	.118	-.120	.053	-.135	-.008	-.157	-.015	-.152	-.006	-.129	.018	-.111	.003	.600	
.600	-.140	.065	-.105	-.001	-.142	-.050	-.151	-.026	-.146	-.031	-.169	-.005	-.154	-.022	-.156	.003
.650	-.133	.037	-.095	-.012	-.126	-.059	-.146	-.031	-.146	-.031	-.169	-.005	-.167	-.024	-.187	.012
$\alpha = 8$																
.0125	-.071	.324	-.085	.344	-.089	.353	-.073	.313	-.068	.294	-.067	.289	-.068	.287	.025	
.025	-.081	.294	-.093	.356	-.089	.328	-.085	.312	-.069	.290	-.068	.273	-.050	.262	.050	
.050	-.089	.240	-.104	.374	-.093	.352	-.085	.327	-.070	.269	-.060	.257	-.037	.174	.075	
.075	-.089	.226	-.112	.383	-.104	.370	-.088	.313	-.107	.271	-.062	.238	-.025	.127	.100	
.100	-.096	.221	-.121	.311	-.117	.395	-.094	.320	-.170	.258	-.060	.207	-.009	.086	.150	
.150	-.108	.231	-.143	.351	-.129	.363	-.098	.229	-.069	.276	-.052	.184	-.005	.051	.200	
.200	-.108	.231	-.143	.351	-.129	.363	-.098	.229	-.069	.276	-.052	.184	-.005	.051	.200	
.250	-.082	.278	-.149	.325	-.118	.333	-.104	.309	-.176	.272	-.052	.195	-.013	.029	.250	
.300	-.154	.254	-.160	.305	-.151	.289	-.115	.255	-.081	.215	-.031	.144	-.013	.054	.350	
.350	-.150	.273	-.165	.320	-.161	.263	-.132	.212	-.080	.150	-.048	.090	-.025	.036	.400	
.400	-.168	.276	-.168	.262	-.177	.198	-.141	.156	-.096	.093	-.064	.059	-.023	.020	.450	
.450	-.176	.232	-.168	.211	-.177	.142	-.158	.095	-.123	.064	-.093	.037	-.071	.004	.500	
.500	-.176	.204	-.155	.152	-.165	.079	-.173	.043	-.150	.026	-.125	.026	-.104	.007	.550	
.550	-.167	.163	-.145	.091	-.155	.023	-.179	.012	-.171	.014	-.154	.022	-.154	.005	.600	
.600	-.158	.102	-.133	.032	-.154	-.017	-.173	-.007	-.187	.005	-.182	.023	-.211	.008	.650	
.650	-.153	.078	-.118	.021	-.142	-.030	-.171	-.014	-.193	.010	-.196	.029	-.235	.018	.700	
$\alpha = 16$																
.0125	-.103	.389	-.113	.443	-.110	.437	-.105	.390	-.097	.371	-.096	.359	-.107	.358	.025	
.025	-.108	.350	-.120	.467	-.115	.423	-.114	.393	-.098	.362	-.101	.339	-.294	.327	.050	
.050	-.114	.294	-.127	.462	-.119	.458	-.115	.381	-.099	.340	-.094	.320	-.070	.222	.075	
.075	-.114	.280	-.137	.458	-.127	.473	-.119	.401	-.100	.353	-.096	.304	-.058	.179	.100	
.100	-.118	.268	-.152	.435	-.138	.456	-.126	.412	-.100	.339	-.094	.281	-.049	.130	.150	
.150	-.132	.282	-.165	.416	-.151	.414	-.130	.394	-.097	.355	-.086	.276	-.037	.104	.200	
.200	-.132	.282	-.165	.416	-.151	.414	-.130	.394	-.097	.355	-.086	.276	-.037	.104	.200	
.250	-.099	.340	-.167	.386	-.158	.388	-.136	.367	-.104	.330	-.064	.215	-.039	.090	.250	
.300	-.173	.317	-.177	.373	-.169	.343	-.144	.312	-.111	.263	-.060	.186	-.043	.097	.300	
.350	-.174	.317	-.181	.377	-.180	.312	-.163	.264	-.100	.186	-.074	.122	-.046	.054	.350	
.400	-.187	.337	-.182	.321	-.196	.250	-.168	.200	-.116	.133	-.091	.083	-.046	.032	.400	
.450	-.195	.290	-.188	.265	-.196	.184	-.179	.133	-.139	.092	-.114	.058	-.112	.016	.450	
.500	-.195	.290	-.188	.265	-.196	.184	-.179	.133	-.139	.092	-.114	.058	-.112	.016	.500	
.550	-.182	.259	-.175	.199	-.180	.116	-.195	.081	-.164	.054	-.146	.041	-.154	.016	.550	
.600	-.177	.214	-.164	.137	-.167	.057	-.203	.046	-.186	.032	-.177	.034	-.200	.015	.600	
.650	-.175	.144	-.145	.073	-.172	.011	-.203	.024	-.204	.026	-.204	.039	-.236	.023	.650	
.700	-.175	.144	-.145	.073	-.172	.011	-.203	.024	-.204							

TABLE III - Continued
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(b) $M = 2.01$ - Continued

x/c	Cp at $y/\delta =$														x/c	
	.10		.35		.55		.67		.77		.87		.97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 12$																
.0125	-1.132	.447	-1.146	.570	-1.145	.571	-1.129	.497	-1.127	.459	-1.129	.438	-1.133	.441	.025	
.025	-1.140	.398	-1.153	.545	-1.148	.539	-1.139	.515	-1.132	.451	-1.132	.416	-1.122	.395	.050	
.050	-1.147	.345	-1.161	.532	-1.151	.579	-1.143	.515	-1.132	.455	-1.125	.405	-1.099	.280	.075	
.075	-1.147	.315	-1.161	.502	-1.158	.558	-1.146	.516	-1.130	.473	-1.128	.398	-1.080	.229	.100	
.100	-1.143	.329	-1.167	.520	-1.158	.530	-1.152	.489	-1.132	.419	-1.126	.395	-1.076	.198	.150	
.150	-1.147	.315	-1.181	.502	-1.169	.485	-1.158	.458	-1.131	.419	-1.119	.346	-1.073	.179	.200	
.200	-1.156	.338	-1.191	.471	-1.169	.452	-1.159	.423	-1.137	.390	-1.091	.268	-1.079	.132	.250	
.250	-1.135	.392	-1.199	.443	-1.188	.402	-1.175	.377	-1.153	.350	-1.091	.268	-1.079	.132	.250	
.300	-1.194	.384	-1.204	.435	-1.194	.400	-1.177	.364	-1.144	.311	-1.087	.235	-1.081	.129	.300	
.350	-1.198	.401	-1.204	.431	-1.208	.379	-1.179	.316	-1.133	.235	-1.098	.164	-1.097	.076	.400	
.400	-1.207	.401	-1.204	.371	-1.221	.303	-1.185	.236	-1.145	.180	-1.119	.125	-1.105	.051	.450	
.500	-1.213	.348	-1.208	.316	-1.217	.227	-1.193	.166	-1.168	.132	-1.136	.086	-1.177	.034	.600	
.600	-1.204	.312	-1.197	.244	-1.202	.155	-1.209	.111	-1.193	.085	-1.169	.069	-1.212	.030	.700	
.700	-1.174	.407	-1.206	.546	-1.179	.543	-1.181	.512	-1.160	.478	-1.147	.398	-1.107	.222	.200	
.800	-1.197	.255	-1.186	.179	-1.193	.098	-1.219	.073	-1.209	.062	-1.199	.058	-1.239	.030	.800	
.850	-1.194	.175	-1.168	.104	-1.188	.046	-1.222	.048	-1.229	.053	-1.220	.064	-1.260	.035	.900	
.900	-1.194	.144	-1.155	.083	-1.183	.033	-1.218	.034	-1.233	.056	-1.232	.069	-1.267	.049	.950	
$a = 14$																
.0125	-1.156	.513	-1.167	.704	-1.162	.746	-1.158	.693	-1.154	.629	-1.151	.546	-1.158	.528	.025	
.025	-1.165	.460	-1.174	.633	-1.167	.662	-1.167	.628	-1.157	.585	-1.155	.540	-1.146	.474	.050	
.050	-1.164	.405	-1.180	.597	-1.170	.668	-1.170	.613	-1.160	.589	-1.151	.534	-1.129	.367	.075	
.075	-1.163	.388	-1.186	.594	-1.177	.626	-1.171	.603	-1.158	.585	-1.153	.521	-1.105	.318	.100	
.100	-1.163	.372	-1.197	.573	-1.186	.599	-1.177	.547	-1.159	.489	-1.152	.476	-1.105	.280	.200	
.150	-1.174	.407	-1.206	.546	-1.179	.543	-1.181	.512	-1.160	.478	-1.147	.398	-1.107	.222	.200	
.200	-1.145	.456	-1.212	.573	-1.202	.507	-1.185	.477	-1.162	.443	-1.117	.309	-1.123	.166	.250	
.300	-1.210	.467	-1.216	.514	-1.210	.463	-1.191	.414	-1.168	.361	-1.114	.275	-1.130	.156	.300	
.350	-1.210	.479	-1.216	.505	-1.221	.431	-1.204	.366	-1.155	.290	-1.122	.206	-1.162	.098	.400	
.400	-1.216	.477	-1.216	.441	-1.237	.356	-1.208	.280	-1.160	.215	-1.142	.156	-1.161	.078	.450	
.500	-1.223	.416	-1.220	.377	-1.233	.270	-1.213	.206	-1.186	.170	-1.165	.121	-1.232	.049	.600	
.600	-1.209	.366	-1.209	.304	-1.219	.199	-1.227	.154	-1.209	.126	-1.190	.097	-1.255	.045	.700	
.700	-1.209	.307	-1.200	.226	-1.209	.133	-1.239	.112	-1.226	.105	-1.222	.087	-1.258	.044	.800	
.800	-1.209	.222	-1.180	.148	-1.209	.081	-1.245	.076	-1.244	.086	-1.240	.095	-1.275	.054	.900	
.900	-1.209	.171	-1.169	.121	-1.199	.072	-1.244	.068	-1.245	.086	-1.254	.095	-1.281	.067	.950	
$a = 16$																
.0125	-1.189	.574	-1.194	.777	-1.191	.847	-1.184	.822	-1.181	.804	-1.183	.724	-1.192	.700	.025	
.025	-1.197	.513	-1.203	.714	-1.200	.746	-1.194	.767	-1.185	.716	-1.166	.666	-1.178	.576	.050	
.050	-1.197	.467	-1.204	.646	-1.203	.753	-1.197	.717	-1.188	.679	-1.183	.652	-1.160	.468	.075	
.075	-1.191	.452	-1.214	.670	-1.208	.710	-1.197	.676	-1.187	.656	-1.183	.604	-1.140	.394	.100	
.100	-1.191	.455	-1.223	.643	-1.218	.649	-1.204	.612	-1.187	.550	-1.183	.527	-1.147	.332	.150	
.150	-1.193	.468	-1.232	.613	-1.230	.601	-1.209	.572	-1.190	.525	-1.173	.442	-1.155	.261	.200	
.200	-1.174	.520	-1.240	.597	-1.234	.577	-1.213	.536	-1.193	.484	-1.143	.348	-1.175	.200	.250	
.300	-1.235	.542	-1.242	.585	-1.242	.535	-1.218	.482	-1.200	.396	-1.140	.313	-1.188	.188	.300	
.350	-1.234	.559	-1.242	.572	-1.253	.489	-1.228	.411	-1.188	.324	-1.150	.248	-1.219	.133	.400	
.400	-1.234	.559	-1.242	.572	-1.253	.489	-1.228	.411	-1.188	.324	-1.150	.248	-1.219	.133	.450	
.500	-1.239	.553	-1.242	.503	-1.268	.402	-1.231	.321	-1.190	.249	-1.170	.191	-1.201	.102	.500	
.600	-1.243	.482	-1.242	.436	-1.261	.317	-1.237	.251	-1.214	.199	-1.190	.150	-1.273	.080	.600	
.650	-1.227	.423	-1.237	.344	-1.253	.243	-1.251	.193	-1.236	.155	-1.215	.129	-1.286	.074	.700	
.700	-1.227	.423	-1.237	.261	-1.240	.168	-1.261	.144	-1.249	.133	-1.242	.120	-1.291	.073	.750	
.800	-1.227	.349	-1.226	.261	-1.231	.117	-1.269	.116	-1.269	.112	-1.262	.120	-1.296	.081	.900	
.850	-1.228	.254	-1.204	.180	-1.231	.101	-1.268	.098	-1.273	.114	-1.276	.123	-1.300	.093	.950	
$a = 18$																
.0125	-1.205	.647	-1.210	.836	-1.207	.963	-1.195	.948	-1.194	.920	-1.195	.865	-1.201	.823	.025	
.025	-1.211	.573	-1.218	.803	-1.215	.883	-1.202	.877	-1.199	.845	-1.200	.795	-1.189	.682	.050	
.050	-1.211	.528	-1.225	.760	-1.220	.843	-1.207	.798	-1.200	.766	-1.198	.743	-1.171	.544	.075	
.075	-1.211	.516	-1.229	.756	-1.226	.789	-1.207	.765	-1.199	.742	-1.200	.683	-1.162	.460	.100	
.100	-1.205	.510	-1.238	.720	-1.234	.741	-1.213	.683	-1.201	.630	-1.200	.583	-1.171	.387	.150	
.150	-1.205	.547	-1.246	.696	-1.242	.680	-1.221	.645	-1.204	.594	-1.192	.495	-1.182	.310	.200	
.200	-1.185	.604	-1.253	.683	-1.248	.654	-1.226	.606	-1.207	.541	-1.161	.398	-1.204	.249	.250	
.300	-1.248	.640	-1.257	.658	-1.258	.606	-1.231	.547	-1.209	.462	-1.161	.369	-1.217	.236	.300	
.350	-1.247	.675	-1.254	.645	-1.266	.544	-1.236	.462	-1.202	.382	-1.170	.298	-1.246	.183	.400	
.400	-1.247	.648	-1.254	.581	-1.280	.466	-1.238	.376	-1.200	.314	-1.189	.240	-1.211	.147	.450	
.500	-1.247	.541	-1.253	.493	-1.272	.373	-1.246	.301	-1.219	.262	-1.209	.201	-1.286	.124	.500	
.600	-1.249	.454	-1.253	.493	-1.272	.373	-1.246	.301	-1.219	.262	-1.209	.201	-1.286	.124	.600	
.700	-1.234	.461	-1.249	.384	-1.266	.290	-1.257	.240	-1.237	.213	-1.230	.178	-1.293	.116	.700	
.750	-1.240	.379	-1.241	.294	-1.255	.208	-1.269	.187	-1.251	.183	-1.253	.161	-1.297	.114	.800	
.800	-1.240	.283	-1.217	.207	-1.239	.143	-1.276	.148	-1.267	.167	-1.274	.160	-1.303	.116	.900	
.900	-1.235	.283	-1.217	.182	-1.234	.141	-1.277	.132	-1.272	.152	-1.283	.160	-1.305	.127	.950	

TABLE III.- Concluded
PRESSURE COEFFICIENTS FOR WARPED TRAPEZOIDAL WING

(b) M = 2.01 - Concluded

x/c	C _p at $y/c = \frac{1}{4}$ of : $\alpha = 20^\circ$														x/c	
	+.10		+.35		+.55		+.67		+.77		+.87		+.97			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
.0125	-.224	+.715	-.227	+.934	-.226	1.073	-.222	1.057	-.214	1.031	-.221	+.984	-.221	+.909	.025	
.050	-.233	+.638	-.235	+.894	-.230	.968	-.226	.960	-.213	.931	-.217	+.891	-.208	+.762	.050	
.075	-.226	+.608	-.241	+.849	-.233	.926	-.227	.881	-.217	.853	-.214	+.820	-.188	+.616	.075	
.100	-.226	+.586	-.245	+.841	-.240	.867	-.228	.840	-.215	.816	-.217	+.757	-.181	.519	.100	
.150	-.222	+.586	-.254	+.802	-.248	.810	-.235	.752	-.220	.692	-.217	.543	-.196	.442	.150	
.200	-.218	+.644	-.260	+.787	-.254	.754	-.242	.714	-.224	.650	-.210	.553	-.206	.366	.200	
.250	-.199	+.707	-.264	+.767	-.263	.729	-.249	.677	-.227	.599	-.213	.461	-.235	.291	.250	
.300	-.265	+.749	-.270	+.743	-.270	.681	-.252	.614	-.232	.513	-.202	.431	-.245	.285	.300	
.350	-.260	+.766	-.264	+.731	-.278	.616	-.260	.520	-.221	.429	-.194	.356	-.273	.228	.350	
.400	-.263	+.720	-.264	+.649	-.291	.521	-.256	.433	-.221	.360	-.210	.295	-.225	.186	.400	
.450	-.264	+.592	-.266	+.541	-.285	.415	-.267	.349	-.242	.296	-.229	.248	-.301	.167	.550	
.500	-.264	+.592	-.266	+.541	-.285	.415	-.267	.349	-.242	.296	-.229	.248	-.301	.167	.600	
.600	-.245	+.497	-.260	+.419	-.277	.325	-.275	.276	-.257	.242	-.248	.221	-.308	.153	.700	
.700	-.245	+.497	-.260	+.419	-.277	.325	-.275	.276	-.257	.242	-.248	.221	-.308	.153	.750	
.750	-.253	+.412	-.253	+.328	-.270	.239	-.286	.221	-.267	.207	-.271	.198	-.310	.145	.800	
.800	-.240	+.316	-.229	+.241	-.261	.184	-.292	.177	-.287	.182	-.289	.193	-.314	.145	.850	
.850	-.228	+.285	-.219	+.217	-.252	.168	-.295	.164	-.286	.178	-.298	.193	-.314	.153	.900	
.900	-.228	+.285	-.219	+.217	-.252	.168	-.295	.164	-.286	.178	-.298	.193	-.314	.153	.950	

TABLE IV
PRESSURE COEFFICIENTS FOR FLAT DELTA WING
(a) M = 1.61

x/c, nominal	Cp at $y/b = \frac{1}{2}$														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 0$																
.0125	.032		.015		.017		.007		.008		.004		.012		.0125	
.025			.001		.018		.004		.017		.007		.028		.050	
.050	.004				.019		.002		.005		.003		.008		.075	
.075	-.046				.019		.002		.005		.003		.019		.100	
.100	.008				.019		.002		.005		.003		.028		.150	
.150	-.011				.003		.002		.006		.025		.030		.200	
.200	.001				.008		.034		.007		.020		.016		.250	
.250	-.015				-.002		-.036		-.020		-.017		-.028		.250	
.300	-.026						-.015		-.025		-.020		-.011		.300	
.350															.350	
.400	.006				-.026		-.018		-.019		-.013		-.023		.400	
.450															.450	
.500	.004				-.019		-.009		-.029		-.043		-.060		.500	
.550															.550	
.600															.600	
.650															.650	
.700															.700	
.750															.750	
.800															.800	
.850															.850	
.900															.900	
.950															.950	
$\alpha = 2$																
.0125	.007	.059	-.061	.077	-.083	.09	-.115	.082	-.154	.075	-.183	.097	-.245	.057	.0125	
.025			-.056	.054	-.082	.059		.059		.054		.074		.050		
.050	-.027	.032	-.055	.052	-.058	.059	-.106	.040	-.154	.059	-.179	.072	-.245	.057		
.075	-.067	-.018	.019	.054	-.047	.043	-.158	.051	-.157	.065	-.130	.043	-.239	.075		
.100	-.010	.018	-.011	.049	-.035	.040	-.155	.057	-.140	.060	-.179	.050	-.239	.100		
.150	-.026	.014	-.020	.036	-.032	.037	-.157	.072	-.154	.080	-.182	.036	-.184	.150		
.200	-.020	.024	-.015	.034	-.032	.037	-.157	.070	-.154	.063	-.166	.024	-.123	.200		
.250	-.029	.036	-.018	.026	-.056	-.013	-.153	.022	-.179	.019	-.190	.031	-.094	.250		
.300	-.035	.003			-.046	.021	-.063	.020	-.071	.027	-.076	.032	-.100	.300		
.350															.350	
.400	-.028	.044	-.050	.007	-.050	.010	-.055	.014	-.057	.023	-.075	.010	-.093	-.006	.400	
.450															.450	
.500	-.019	.032	-.042	.007	-.038	.015	-.060	.012	-.079	-.012	-.104	.031	-.116	-.022	.500	
.550															.550	
.600	-.086	-.051	-.052	-.012											.600	
.650															.650	
.700	-.080	-.051	-.067	-.030											.700	
.750															.750	
.800	-.047	-.021	-.060	-.031	-.070	-.028	-.076	-.031	-.084	-.028	-.095	-.033	-.110	-.031	.800	
.850															.850	
.900	-.050	-.023	-.055	-.022	-.062	-.023	-.069	-.017	-.077	-.024	-.129	-.068	-.110	-.041	.900	
.950															.950	
$\alpha = 4$																
.0125	-.021	.09	-.204	.159	-.227	.147	-.230	.139	-.209	.166	-.227	.131			.0125	
.025			-.170	.105	-.230	.104	-.230	.124	-.211	.126	-.227	.133	-.246	.116	.025	
.050	-.057	.062	-.089	.059	-.207	.104	-.240	.079	-.211	.118	-.203	.112	-.234	.096	.050	
.075	-.087	.073	-.060	.089	-.155	.094	-.216	.118	-.203	.106	-.227	.101	-.246	.123	.075	
.100	-.035	.059	-.066	.086	-.120	.087	-.198	.093	-.208	.106	-.235	.076	-.246	.100	.100	
.150	-.051	.046	-.054	.070	-.077	.076	-.180	.034	-.220	.058	-.235	.088	-.246	.150		
.200	-.043	.056	-.048	.067	-.035	.099	-.077	.080	-.210	.068	-.239	.075	-.246	.084	.200	
.250	-.051	.037	-.048	.060	-.075	.027	-.066	.059	-.200	.060	-.238	.074	-.246	.077	.250	
.300	-.056	.033			-.061	.052	-.069	.052	-.171	.066	-.230	.072	-.250	.050	.300	
.350															.350	
.400	-.064	.073	-.085	.037	-.074	.044	-.079	.045	-.105	.055	-.224	.050	-.250	.039	.400	
.450															.450	
.500	-.044	.062	-.071	.035	-.059	.044	-.081	.029	-.096	.025	-.205	.015	-.262	.029	.500	
.550															.550	
.600	-.111	-.023	-.080	.019											.600	
.650															.650	
.700	-.091	-.027	-.090	-.006											.700	
.750															.750	
.800	-.071	.005	-.083	-.003	-.085	-.004	-.096	-.004	-.101	.001	-.103	-.006	-.254	.001	.800	
.850															.850	
.900	-.075	.005	-.079	.005	-.077	.005	-.086	.007	-.088	-.000	-.133	-.031	-.254	.001	.900	
.950															.950	
$\alpha = 6$																
.0125	-.047	.123	-.324	.159	-.265	.189	-.251	.176	-.237	.224	-.270	.159			.0125	
.025	-.074	.082	-.153	.106	-.265	.136	-.259	.126	-.244	.166	-.269	.176	-.277	.150	.025	
.050															.050	
.075	-.107	.038	-.098	.122	-.260	.129	-.251	.167	-.241	.151	-.270	.134	-.277	.100	.075	
.100	-.063	.089	-.081	.118	-.264	.120	-.257	.124	-.245	.146	-.270	.132	-.277	.141	.100	
.150	-.075	.071	-.085	.101	-.272	.119	-.292	.078	-.262	.090	-.276	.111	-.277	.141	.150	
.200	-.066	.076	-.079	.097	-.246	.119	-.289	.117	-.266	.110	-.280	.116	-.280	.124	.200	
.250	-.073	.063	-.075	.090	-.213	.073	-.316	.097	-.278	.102	-.280	.111	-.280	.117	.250	
.300	-.081	.048			-.026	.089	-.298	.093	-.287	.104	-.282	.109	-.287	.089	.300	
.350															.350	
.400	-.115	.109	-.113	.062	-.084	.081	-.119	.079	-.293	.089	-.302	.087	-.291	.079	.400	
.450															.450	
.500	-.063	.086	-.095	.062	-.078	.080	-.060	.068	-.284	.062	-.317	.053	-.298	.071	.500	
.550															.550	
.600	-.140	-.003	-.102	.045											.600	
.650															.650	
.700	-.113	-.010	-.113	.016											.700	
.750															.750	
.800	-.091	.030	-.104	.020	-.103	.024	-.104	.027	-.089	.033	-.312	.040	-.329	.038	.800	
.850															.850	
.900	-.098	.030	-.103	.030	-.096	.041	-.099	.047	-.073	.040	-.307	.003	-.329	.038	.900	
.950															.950	

TABLE IV.- Continued
PRESSURE COEFFICIENTS FOR FLAT DELTA WING

(a) $M = 1.61$ - Continued

x/c , nominal	C_p at $y/D = 0$														x/c , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 8$																
.0125	-.099	.156	-.329	.194	-.285	.224	-.267	.213	-.271	.239	-.302	.175	-.310	.172	.0125	
.025	-.131	.114	-.327	.176	-.291	.164	-.277	.165	-.279	.194	-.297	.167	-.307	.165	.025	
.050	-.164	.065	-.330	.162	-.291	.172	-.277	.173	-.277	.188	-.297	.178	-.307	.176	.050	
.100	-.174	.107	-.309	.159	-.297	.164	-.282	.156	-.297	.177	-.297	.178	-.307	.180	.100	
.150	-.098	.108	-.076	.141	-.322	.113	-.305	.121	-.306	.135	-.304	.156	-.309	.171	.150	
.200	-.086	.108	-.081	.135	-.319	.160	-.303	.154	-.306	.149	-.308	.146	-.313	.153	.200	
.250	-.091	.099	-.074	.134	-.343	.106	-.363	.135	-.310	.139	-.314	.145	-.313	.144	.250	
.300	-.101	.083	-.074	.130	-.310	.125	-.362	.131	-.317	.139	-.317	.141	-.317	.122	.300	
.350	-.166	.153	-.119	.100	-.099	.116	-.355	.111	-.348	.127	-.327	.117	-.322	.115	.400	
.450	-.066	.124	-.109	.100	-.081	.113	-.301	.100	-.357	.095	-.347	.082	-.329	.102	.450	
.550	-.163	.028	-.117	.080	-.215	.054	-.353	.058	-.359	.086	-.333	.087	-.331	.060	.550	
.650	-.126	.020	-.124	.049	-.126	.063	-.339	.089	-.372	.063	-.335	.092	-.341	.077	.650	
.750	-.105	.066	-.116	.056	-.117	.057	-.105	.058	-.314	.068	-.376	.059	-.351	.066	.800	
.850	-.114	.063	-.114	.069	-.111	.074	-.101	.082	-.257	-.050	-.383	.037	-.351	.060	.850	
.900	-.114	.063	-.114	.069	-.111	.074	-.101	.082	-.257	-.050	-.383	.037	-.351	.066	.900	
.950	-.114	.063	-.114	.069	-.111	.074	-.101	.082	-.257	-.050	-.383	.037	-.351	.066	.950	
$a = 10$																
.0125	-.182	.182	-.347	.225	-.310	.242	-.286	.245	-.308	.263	-.328	.193	-.332	.187	.0125	
.025	-.268	.144	-.379	.177	-.316	.203	-.301	.190	-.314	.222	-.323	.236	-.332	.187	.025	
.075	-.199	.114	-.402	.198	-.323	.205	-.296	.219	-.314	.222	-.328	.189	-.331	.175	.075	
.100	-.132	.164	-.402	.191	-.330	.207	-.305	.202	-.320	.219	-.324	.200	-.328	.207	.100	
.150	-.127	.137	-.306	.175	-.331	.218	-.338	.159	-.330	.172	-.329	.184	-.331	.195	.150	
.200	-.112	.144	-.112	.169	-.385	.199	-.333	.193	-.333	.194	-.334	.179	-.334	.180	.200	
.250	-.112	.129	-.092	.173	-.407	.138	-.347	.172	-.348	.179	-.333	.177	-.334	.174	.250	
.300	-.127	.116	-.092	.173	-.412	.160	-.362	.177	-.345	.178	-.338	.177	-.338	.152	.300	
.350	-.184	.184	-.139	.137	-.315	.148	-.413	.153	-.352	.169	-.354	.152	-.340	.147	.400	
.450	-.108	.159	-.129	.134	-.162	.149	-.397	.143	-.387	.142	-.360	.114	-.350	.130	.500	
.550	-.184	.057	-.136	.112	-.360	.094	-.399	.098	-.364	.114	-.358	.124	-.361	.120	.650	
.650	-.145	.044	-.142	.080	-.301	.102	-.405	.128	-.373	.097	-.361	.127	-.361	.111	.750	
.750	-.125	.095	-.135	.087	-.124	.091	-.213	.096	-.403	.105	-.385	.099	-.360	.113	.800	
.850	-.151	.095	-.133	.100	-.124	.104	-.166	.114	-.388	.111	-.411	.060	-.374	.137	.850	
.900	-.151	.095	-.133	.100	-.124	.104	-.166	.114	-.388	.111	-.411	.060	-.374	.113	.900	
.950	-.151	.095	-.133	.100	-.124	.104	-.166	.114	-.388	.111	-.411	.060	-.374	.113	.950	
$a = 12$																
.0125	-.240	.217	-.349	.248	-.322	.256	-.305	.261	-.337	.263	-.348	.187	-.347	.204	.0125	
.025	-.322	.176	-.363	.214	-.327	.226	-.318	.202	-.342	.231	-.346	.241	-.347	.204	.025	
.075	-.309	.148	-.381	.235	-.332	.231	-.314	.245	-.339	.243	-.350	.215	-.351	.075		
.100	-.234	.201	-.398	.228	-.329	.245	-.324	.232	-.344	.240	-.346	.229	-.346	.100		
.150	-.150	.169	-.391	.209	-.337	.256	-.345	.184	-.354	.199	-.349	.220	-.349	.150		
.200	-.122	.180	-.352	.206	-.356	.230	-.346	.221	-.357	.214	-.352	.212	-.359	.200		
.250	-.114	.160	-.327	.211	-.392	.174	-.360	.203	-.365	.199	-.356	.213	-.355	.212		
.300	-.126	.147	-.411	.197	-.367	.206	-.366	.205	-.366	.205	-.362	.213	-.355	.190		
.350	-.182	.219	-.183	.173	-.410	.179	-.410	.181	-.374	.193	-.370	.189	-.362	.183	.400	
.450	-.123	.193	-.138	.169	-.370	.182	-.411	.171	-.399	.166	-.377	.156	-.371	.168	.500	
.550	-.194	.094	-.149	.144	-.429	.117	-.429	.117	-.425	.119	-.379	.156	-.376	.163	.600	
.650	-.160	.076	-.160	.110	-.409	.123	-.424	.149	-.389	.134	-.377	.167	-.377	.157	.650	
.750	-.143	.127	-.147	.119	-.182	.124	-.369	.119	-.435	.129	-.396	.147	-.374	.157	.750	
.850	-.150	.124	-.150	.124	-.148	.126	-.334	.129	-.442	.125	-.418	.087	-.374	.137	.850	
.900	-.150	.124	-.150	.124	-.148	.126	-.334	.129	-.442	.125	-.418	.087	-.374	.137	.900	
.950	-.150	.124	-.150	.124	-.148	.126	-.334	.129	-.442	.125	-.418	.087	-.374	.137	.950	
$a = 14$																
.0125	-.286	.260	-.350	.269	-.350	.271	-.317	.281	-.353	.280	-.365	.201	-.363	.217	.0125	
.025	-.354	.214	-.361	.252	-.355	.259	-.330	.234	-.358	.262	-.363	.247	-.362	.222	.025	
.075	-.183	.198	-.368	.272	-.358	.270	-.326	.283	-.355	.282	-.363	.247	-.362	.075		
.100	-.131	.245	-.375	.266	-.355	.279	-.339	.279	-.360	.281	-.362	.259	-.362	.100		
.150	-.197	.206	-.401	.249	-.360	.293	-.367	.221	-.368	.240	-.362	.256	-.365	.150		
.200	-.146	.219	-.425	.239	-.375	.269	-.369	.269	-.375	.256	-.366	.252	-.365	.200		
.250	-.122	.200	-.419	.255	-.400	.213	-.381	.251	-.387	.244	-.371	.247	-.365	.250		
.300	-.150	.192	-.419	.243	-.425	.243	-.383	.246	-.391	.252	-.379	.247	-.369	.223		
.350	-.198	.261	-.246	.212	-.450	.219	-.409	.231	-.398	.237	-.397	.219	-.374	.218	.400	
.450	-.149	.231	-.187	.211	-.425	.224	-.421	.217	-.408	.207	-.407	.194	-.386	.206	.500	
.550	-.211	.132	-.175	.182	-.449	.161	-.425	.165	-.405	.195	-.397	.215	-.374	.600		
.650	-.179	.110	-.182	.145	-.447	.165	-.424	.197	-.413	.179	-.405	.215	-.410	.191		
.750	-.165	.165	-.176	.160	-.266	.173	-.420	.182	-.434	.195	-.416	.185	-.415	.197	.800	
.850	-.172	.175	-.176	.181	-.221	.193	-.395	.202	-.448	.199	-.412	.158	-.415	.197	.850	
.900	-.172	.175	-.176	.181	-.221	.193	-.395	.202	-.448	.199	-.412	.158	-.415	.197	.900	
.950	-.172	.175	-.176	.181	-.221	.193	-.395	.202	-.448	.199	-.412	.158	-.415	.197	.950	

TABLE IV.- Continued
PRESSURE COEFFICIENTS FOR FLAT DELTA WING

(a) $M = 1.61$ - Concluded

x/c , nominal	C_p at $y/b/2$ of:														x/c , nominal	
	.10		.25		.40		.50		.60		.70		.80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 16$																
.0125	-320	.312	-334	.297	-356	.293	-370	.295	-370	.288	-385	.212	-393	.244	.0125	
.025	-373	.266	-360	.297	-361	.259	-331	.295	-370	.288	-383	.292	-393	.244	.025	
.050	-423	.238	-367	.303	-356	.292	-346	.261	-377	.288	-383	.286	-393	.295	.050	
.075	-439	.295	-385	.314	-365	.322	-343	.321	-375	.315	-393	.296	-390	.295	.075	
.100	-456	.255	-408	.314	-365	.328	-353	.321	-378	.313	-391	.293	-391	.300	.100	
.150	-456	.255	-408	.295	-394	.327	-383	.255	-384	.267	-391	.293	-391	.292	.200	
.200	-419	.268	-433	.285	-387	.317	-395	.313	-384	.290	-394	.293	-392	.298	.250	
.250	-419	.253	-433	.313	-422	.256	-416	.287	-388	.285	-390	.291	-392	.298	.300	
.300	-419	.244	-444	.313	-434	.287	-421	.281	-397	.288	-389	.291	-395	.283	.350	
.350	-214	.313	-313	.266	-440	.263	-422	.266	-428	.272	-400	.284	-395	.281	.400	
.400	-153	.283	-243	.261	-447	.264	-415	.256	-433	.256	-433	.260	-400	.281	.450	
.450	-206	.180	-204	.231				-435	.209	-427	.217	-440	.267	-403	.266	.500
.500	-206	.180	-204	.231				-441	.229	-421	.266	-440	.245	-407	.283	.700
.550	-190	.155	-211	.194				-441	.229	-421	.266	-440	.245	-414	.268	.750
.600	-171	.232	-189	.231	-331	.248	-444	.232	-425	.248	-429	.261	-425	.277	.800	
.650	-180	.246	-184	.256	-283	.261	-425	.270	-434	.264	-420	.236			.850	
.700	-180	.246	-184	.256	-283	.261	-425	.270	-434	.264	-420	.236			.900	
.750	-195	.348	-206	.374	-338	.401	-449	.438	-439	.445	-433				.950	
$a = 18$																
.0125	-358	.356	-374	.323	-370	.311	-397	.295	-411	.232					.0125	
.025	-374	.331	-374	.320	-360	.303	-400	.314	-412	.328	-406	.271			.025	
.050	-405	.321	-383	.340	-368	.338	-376	.303	-400	.314	-412	.331			.050	
.075	-437	.269	-403	.357	-368	.362	-372	.360	-398	.338	-415	.331			.075	
.100	-431	.342	-409	.358	-375	.379	-375	.360	-402	.340	-412	.341	-404	.333	.100	
.150	-323	.300	-343	.340	-404	.358	-400	.294	-407	.309	-412	.346	-404	.341	.150	
.200	-256	.316	-440	.332	-394	.370	-410	.364	-407	.343	-411	.346	-403	.336	.200	
.250	-171	.301	-445	.372	-430	.298	-442	.323	-405	.340	-410	.346	-404	.343	.250	
.300	-204	.288			-445	.343	-452	.327	-408	.349	-405	.353	-404	.358	.300	
.350	-246	.356	-381	.318	-441	.319	-437	.326	-439	.338	-405	.343	-405	.336	.400	
.400	-187	.323	-306	.313	-451	.320	-427	.331	-446	.326	-434	.312	-405	.330	.450	
.450	-254	.232	-253	.293				-442	.279	-440	.280	-450	.324	-408	.361	.500
.500	-219	.227	-243	.269				-451	.306	-436	.323	-453	.324	-415	.413	.600
.550	-189	.301	-221	.304	-380	.309	-463	.313	-436	.340	-438	.393	-423	.423	.700	
.600	-191	.348	-206	.374	-338	.401	-449	.438	-439	.445	-433	.429			.800	
.650	-191	.348	-206	.374	-338	.401	-449	.438	-439	.445	-433				.850	
.700	-191	.348	-206	.374	-338	.401	-449	.438	-439	.445	-433				.900	
.750	-195	.348	-206	.374	-338	.401	-449	.438	-439	.445	-433				.950	
$a = 20$																
.0125	-385	.390	-389	.345	-392	.331	-405	.312	-418	.262					.0125	
.025	-389	.389	-365	.397	-372	.340	-405	.313	-419	.373	-421	.306			.025	
.050	-415	.396	-395	.387	-394	.375	-485	.342	-410	.355	-421	.375			.050	
.075	-458	.308	-426	.402	-391	.404	-386	.398	-408	.389	-421	.375			.075	
.100	-458	.394	-444	.400	-391	.404	-386	.397	-412	.403	-424	.388	-419	.384	.100	
.150	-376	.348	-445	.387	-411	.416	-404	.358	-415	.389	-425	.397	-419	.406	.150	
.200	-303	.358	-448	.391	-419	.417	-418	.420	-415	.419	-425	.409	-419	.402	.200	
.250	-230	.348	-443	.435	-428	.349	-446	.390	-419	.414	-419	.410	-419	.418	.250	
.300	-241	.354			-453	.400	-456	.417	-426	.415	-413	.418	-419	.418	.300	
.350	-284	.402	-411	.373	-445	.401	-448	.400	-452	.412	-413	.410	-419	.467	.400	
.400	-211	.381	-349	.391	-453	.393	-430	.407	-454	.395	-449	.397	-421	.499	.500	
.450	-228	.296	-287	.343				-448	.344	-446	.370	-463	.475	-424	.496	.550
.500	-200	.378	-242	.395	-406	.458	-473	.486	-446	.506	-451	.501	-441	.482	.600	
.550	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452	-452	.475	.650	
.600	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452			.700	
.650	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452			.750	
.700	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452			.800	
.750	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452			.850	
.800	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452			.900	
.850	-196	.479	-233	.499	-365	.501	-460	.511	-446	.496	-443	.452			.950	

TABLE IV.- Continued
PRESSURE COEFFICIENTS FOR FLAT DELTA WING

(b) M = 2.01

x/c, nominal	Cp at $\sqrt{\frac{V}{E}}$ of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 0$																
.025	.038		.022		.020		.005		.010		.015		.025		.0125	
.050	.014		.003		.002		.016		.016		.017		.040		.050	
.075	.015		.003		.001		.006		.009		.020		.075		.075	
.100	.008		.002		.003		.012		.012		.017		.029		.100	
.150	.002		.003		.008		.019		.020		.022		.036		.150	
.200	.003		.002		.010		.014		.021		.027		.043		.200	
.250	.005		.000		.017		.020		.031		.034		.044		.250	
.300	.011		.004		.016		.022		.038		.035		.046		.300	
.350															.350	
.400	.011		.026		.022		.030		.031		.035		.042		.400	
.450															.450	
.500	.009		.033		.029		.039		.055		.057		.060		.500	
.550															.550	
.600	.050		.027				.056		.063		.055		.065		.600	
.650															.650	
.700	.059		.043				.051		.043		.063		.070		.700	
.750															.750	
.800	.039		.053		.049		.051		.055		.059		.070		.800	
.850															.850	
.900	.043		.044		.051		.056		.052		.079		.070		.900	
.950															.950	
$a = 2$																
.0125	.004	.076	-.032	.073	-.054	.086	-.074	.072	-.074	.083	-.117	.073			.0125	
.025	-.002	.066	-.040	.058	-.063	.054	-.070	.038	-.087	.055	-.114	.058	-.183	.031	.050	
.050	-.009	.045	-.033	.041	-.049	.053	-.062	.048	-.074	.052	-.152	.033			.075	
.075	-.005	.039	-.026	.041	-.045	.045	-.062	.048	-.074	.042	-.160	.039	-.178	.034	.100	
.100	-.006	.032	-.020	.040	-.040	.040	-.068	.032	-.075	.022	-.131	.018	-.179	.021	.150	
.150	-.010	.019	-.018	.029	-.035	.028	-.065	.016	-.070	.024	-.144	.024	-.097	.008	.200	
.200	-.012	.024	-.018	.037	-.044	.030	-.060	.021	-.074	.024	-.160	.003	-.154	.016	.250	
.250	-.019	.017	-.018	.027	-.043	.003	-.037	.016	-.051	.003	-.105	.003	-.154	.016	.300	
.300	-.026	.010	-.018	.020	-.030	.011	-.042	.011	-.051	.008	-.101	.014	-.118	.005	.350	
.350															.350	
.400	-.053	.024	-.045	.002	-.036	.009	-.046	.001	-.036	.018	-.096	.004	-.111	-.005	.400	
.450															.450	
.500	-.032	.012	-.053	-.006	-.053	.000	-.064	-.008	-.076	-.024	-.125	-.034	-.125	-.029	.500	
.550															.550	
.600	-.067	-.027	-.044	-.004			-.075	-.026	-.089	-.031	-.129	-.024	-.127	-.032	.600	
.650															.650	
.700	-.065	-.038	-.062	-.020			-.074	-.023	-.066	-.012	-.138	-.038	-.132	-.037	.700	
.750															.750	
.800	-.046	-.021	-.062	-.037	-.065	-.027	-.065	-.023	-.074	-.025	-.133	-.034	-.144	-.043	.800	
.850															.850	
.900	-.051	-.025	-.055	-.029	-.064	-.026	-.068	-.027	-.074	-.024	-.148	-.054	-.177	-.008	.900	
.950															.950	
$a = 4$																
.0125	-.046	.109	-.111	.114	-.146	.129	-.171	.121	-.171	.136	-.205	.114			.0125	
.025	-.042	.098	-.099	.099	-.148	.096	-.174	.079	-.190	.101	-.199	.101	-.209	.094	.050	
.050	-.038	.068	-.098	.075	-.148	.090	-.174	.083	-.170	.096	-.193	.083			.075	
.075	-.033	.064	-.088	.072	-.142	.080	-.174	.075	-.177	.069	-.166	.083	-.197	.083	.100	
.100	-.033	.057	-.045	.072	-.133	.075	-.174	.069	-.166	.083	-.197	.083	-.212	.090	.150	
.150	-.038	.037	-.039	.060	-.071	.056	-.167	.048	-.150	.056	-.190	.066	-.212	.070	.200	
.200	-.032	.046	-.037	.050	-.047	.072	-.176	.053	-.115	.051	-.153	.059	-.211	.058	.250	
.250	-.036	.039	-.042	.051	-.077	.028	-.168	.043	-.114	.040	-.188	.052	-.205	.051	.300	
.300	-.043	.029	-.046	.038	-.064	.037	-.173	.053	-.105	.048	-.171	.050	-.197	.041	.350	
.350															.350	
.400	-.066	.046	-.065	.025	-.063	.032	-.074	.024	-.087	.040	-.150	.037	-.192	.026	.400	
.450															.450	
.500	-.056	.034	-.076	.017	-.085	.026	-.095	.018	-.112	.006	-.126	.004	-.190	.008	.500	
.550															.550	
.600	-.090	-.010	-.069	.022			-.110	.000	-.118	-.002	-.101	.007	-.187	.008	.600	
.650															.650	
.700	-.082	-.033	-.082	-.004			-.099	-.001	-.107	-.015	-.110	-.003	-.185	.008	.700	
.750															.750	
.800	-.061	-.008	-.082	-.022	-.092	-.009	-.093	-.002	-.109	-.001	-.111	-.001	-.186	.002	.800	
.850															.850	
.900	-.074	-.009	-.076	-.009	-.091	-.012	-.097	-.010	-.109	-.000	-.136	-.004	-.177	-.008	.900	
.950															.950	
$a = 6$																
.0125	-.091	.142	-.162	.145	-.195	.164	-.201	.156	-.188	.166	-.212	.142	-.234	.121	.050	
.025	-.078	.133	-.161	.139	-.192	.133	-.201	.156	-.188	.138	-.203	.139	-.234	.121		
.050	-.046	.083	-.120	.129	-.193	.108	-.202	.124	-.193	.137	-.218	.121	-.235	.075		
.075	-.042	.092	-.124	.114	-.194	.118	-.202	.124	-.194	.136	-.218	.121	-.235	.100		
.100	-.046	.083	-.120	.129	-.193	.108	-.202	.124	-.194	.136	-.218	.121	-.235	.150		
.150	-.057	.063	-.099	.094	-.169	.098	-.206	.065	-.192	.084	-.221	.085	-.237	.108		
.200	-.047	.073	-.070	.093	-.157	.107	-.184	.086	-.185	.087	-.203	.093	-.240	.095	.200	
.250	-.055	.067	-.065	.082	-.148	.062	-.186	.078	-.186	.078	-.225	.085	-.257	.089	.250	
.300	-.060	.058	-.062	.067	-.108	.074	-.165	.084	-.186	.078	-.221	.089	-.237	.076	.300	
.350															.350	
.400	-.063	.076	-.074	.054	-.062	.063	-.160	.057	-.186	.074	-.221	.068	-.237	.063	.400	
.450															.450	
.500	-.066	.067	-.097	.047	-.093	.060	-.127	.051	-.191	.037	-.214	.038	-.240	.048	.500	
.550															.550	
.600	-.103	.010	-.081	.050			-.112	.031	-.189	.031	-.211	.043	-.240	.046	.600	
.650															.650	
.700	-.091	-.013	-.096	.018			-.097	.027	-.177	.047	-.214	.031	-.244	.035	.700	
.750															.750	
.800	-.075	.017	-.092	.004	-.101	.012	-.092	.022	-.161	.025	-.214	.029	-.244	.028	.800	
.850															.850	
.900	-.089	.014	-.091	.014	-.101	.014	-.100	.013	-.142	.022	-.222	.009	-.244	.028	.900	
.950					</td											

TABLE IV.- Continued
PRESSURE COEFFICIENTS FOR FLAT DELTA WING
(b) $M = 2.01$ - Continued

TABLE IV:- Concluded
PRESSURE COEFFICIENTS FOR FLAT DELTA WING

(b) M = 2.01 - Concluded

x/c, nominal	Cp at $y/c = \frac{1}{2}$ of:														x/c, nominal	
	.10		.25		.40		.50		.60		.70		.80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 16^\circ$																
.0125	-.241	.213	-.240	.289	-.258	.207	-.264	.307	-.242	.294	-.268	.240	-.240	.248	.0125	
.025	-.243	.337	-.260	.312	-.262	.270	-.264	.275	-.257	.296	-.256	.300	-.287	.248	.025	
.050	-.240	.291	-.264	.296	-.261	.299	-.267	.275	-.257	.295	-.279	.281	-.287	.248	.050	
.075	-.240	.300	-.264	.311	-.262	.307	-.265	.307	-.244	.295	-.279	.300	-.289	.274	.075	
.100	-.245	.285	-.267	.252	-.255	.303	-.264	.304	-.258	.295	-.284	.281	-.289	.280	.100	
.150	-.208	.254	-.275	.289	-.276	.290	-.266	.258	-.264	.260	-.284	.279	-.289	.274	.150	
.200	-.144	.258	-.282	.284	-.272	.311	-.268	.283	-.270	.276	-.273	.275	-.290	.263	.200	
.250	-.123	.251	-.274	.275	-.278	.254	-.275	.266	-.278	.269	-.285	.271	-.290	.263	.250	
.300	-.130	.242	-.284	.255	-.290	.266	-.267	.276	-.278	.273	-.286	.277	-.290	.245	.300	
.350															.350	
.400	-.112	.275	-.245	.247	-.298	.259	-.294	.249	-.283	.261	-.290	.253	-.290	.235	.400	
.450															.450	
.500	-.133	.289	-.214	.242	-.293	.252	-.284	.232	-.290	.226	-.290	.222	-.290	.221	.500	
.550															.550	
.600	-.173	.152	-.192	.211			-.294	.190	-.296	.193	-.294	.212	-.294	.215	.600	
.650															.650	
.700	-.158	.115	-.189	.169			-.298	.184	-.299	.216	-.301	.192	-.295	.209	.700	
.750															.750	
.800	-.146	.168	-.175	.164	-.264	.173	-.292	.175	-.300	.185	-.304	.189	-.299	.185	.800	
.850															.850	
.900	-.161	.179	-.168	.178	-.247	.181	-.294	.178	-.301	.183	-.305	.159	-.299	.185	.900	
.950															.950	
$\alpha = 18^\circ$																
.0125	-.250	.347	-.268	.318	-.268	.331	-.270	.336	-.258	.316	-.276	.265	-.276	.270	.0125	
.025	-.256	.377	-.268	.350	-.269	.301	-.273	.312	-.264	.327	-.257	.329	-.296	.270	.025	
.050	-.249	.342	-.273	.337	-.270	.333	-.273	.351	-.250	.335	-.286	.311	-.296	.311	.050	
.075	-.275	.342	-.276	.354	-.259	.346	-.273	.351	-.266	.339	-.265	.337	-.291	.312	.075	
.100	-.280	.326	-.277	.357	-.262	.344	-.266	.349	-.265	.337	-.291	.312	-.296	.304	.100	
.150	-.250	.295	-.288	.331	-.274	.328	-.269	.294	-.270	.301	-.291	.312	-.296	.304	.150	
.200	-.197	.299	-.292	.326	-.280	.357	-.272	.326	-.277	.323	-.281	.308	-.296	.300	.200	
.250	-.162	.293	-.280	.314	-.286	.301	-.283	.311	-.281	.317	-.292	.308	-.294	.299	.250	
.300	-.156	.280	-.294	.299	-.293	.311	-.282	.325	-.281	.317	-.291	.308	-.295	.285	.300	
.350															.350	
.400	-.133	.320	-.271	.285	-.302	.305	-.297	.295	-.286	.304	-.293	.292	-.295	.279	.400	
.450															.450	
.500	-.152	.323	-.249	.279	-.295	.297	-.285	.281	-.298	.270	-.294	.262	-.296	.263	.500	
.550															.550	
.600	-.188	.189	-.230	.251			-.297	.231	-.302	.235	-.298	.251	-.301	.259	.600	
.650															.650	
.700	-.172	.149	-.218	.206			-.300	.225	-.300	.257	-.304	.228	-.301	.256	.700	
.750															.750	
.800	-.159	.209	-.203	.206	-.280	.217	-.296	.215	-.301	.226	-.308	.228	-.301	.232	.800	
.850															.850	
.900	-.175	.221	-.194	.216	-.266	.211	-.299	.222	-.302	.225	-.305	.183	-.301	.232	.900	
.950															.950	
$\alpha = 20^\circ$																
.0125	-.258	.377	-.274	.348	-.274	.353	-.275	.358	-.247	.329	-.282	.267	-.302	.282	.0125	
.025	-.257	.438	-.276	.384	-.272	.332	-.276	.343	-.265	.346	-.257	.346	-.302	.305	.025	
.050	-.277	.394	-.276	.378	-.274	.360	-.276	.343	-.297	.357	-.297	.338	-.302	.375	.050	
.075	-.284	.395	-.282	.397	-.273	.383	-.276	.390	-.253	.357	-.299	.342	-.304	.329	.075	
.100	-.290	.373	-.283	.403	-.266	.373	-.272	.388	-.267	.365	-.299	.342	-.304	.329	.100	
.150	-.271	.342	-.294	.377	-.278	.366	-.274	.327	-.275	.330	-.301	.342	-.302	.328	.150	
.200	-.237	.348	-.294	.371	-.284	.404	-.274	.361	-.279	.355	-.289	.342	-.303	.326	.200	
.250	-.203	.340	-.280	.364	-.290	.332	-.287	.346	-.282	.344	-.300	.342	-.304	.326	.250	
.300	-.191	.327	-.297	.354	-.294	.350	-.275	.334	-.282	.350	-.300	.347	-.304	.315	.300	
.350															.350	
.400	-.160	.370	-.286	.335	-.301	.342	-.298	.329	-.289	.336	-.303	.325	-.302	.309	.400	
.450															.450	
.500	-.172	.373	-.270	.323	-.300	.333	-.286	.315	-.300	.297	-.304	.297	-.302	.292	.500	
.550															.550	
.600	-.208	.235	-.256	.298			-.298	.264	-.305	.263	-.305	.282	-.305	.289	.600	
.650															.650	
.700	-.185	.194	-.245	.246			-.298	.259	-.305	.288	-.313	.264	-.305	.287	.700	
.750															.750	
.800	-.180	.255	-.228	.247	-.291	.251	-.298	.248	-.301	.260	-.316	.264	-.305	.283	.800	
.850															.850	
.900	-.193	.265	-.221	.269	-.276	.261	-.299	.262	-.298	.260	-.311	.217	-.305	.263	.900	
.950															.950	

TABLE V
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

TABLE V. - Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

(a) $M = 1.61$ - Continued

x/c , nominal	C_p at y/c of:														x/c , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -12^\circ$																
.0125	.282	-.354	.260	-.338	.217	-.322	.231	-.304	.209	-.313	.173	-.312	.168	-.310	.0125	
.050	.281	-.423	.266	-.343	.221	-.323	.237	-.306	.241	-.318	.231	-.314	.226	-.310	.050	
.075	.248	-.421	.266	-.357	.257	-.322	.237	-.306	.231	-.319	.226	-.310	.187	-.310	.075	
.100	.264	-.294	.230	-.368	.231	-.332	.237	-.313	.231	-.320	.197	-.306	.307	-.307	.100	
.150	.261	-.091	.219	-.407	.227	-.337	.224	-.331	.195	-.322	.187	-.312	.190	-.312	.150	
.200	.197	-.111	.188	-.445	.224	-.351	.210	-.344	.198	-.323	.194	-.312	.172	-.319	.200	
.250	.201	-.135	.189	-.477	.224	-.372	.198	-.368	.190	-.321	.171	-.312	.162	-.319	.250	
.300	.188	-.117	.189	-.328	.190	-.380	.188	-.339	.166	-.338	.166	-.314	.129	-.318	.300	
.350															.350	
.400	.216	-.158	.150	-.166	.183	-.425	.147	-.364	.152	-.355			.324	-.321	.400	
.450															.450	
.500	.704	-.107			.120	-.132			.126	-.151	.139	-.148	.124	-.150	.500	
.550															.550	
.600	.120	-.136	.115	-.117	.105	-.345	.113	-.426	.100	-.354	.098	-.341	.094	-.328	.600	
.650															.650	
.700	.053		.075	-.105	.075	-.229	.082	-.422	.078	-.361	.085	-.348	.090	-.328	.700	
.750															.750	
.800	.073	-.105	.071	-.105	.069	-.161	.056	-.389	.065	-.369	.062	-.347			.800	
.850															.850	
.900	.056	-.108	.069	-.086	.059	-.116	.076	-.317	.075	-.410					.900	
.950															.950	
$\alpha = -10^\circ$																
.0125	.240	-.342	.235	-.300	.206	-.300	.209	-.276	.202	-.276	.165	-.278	.158	-.279	.0125	
.050	.240	-.317	.210	-.310	.196	-.295	.207	-.276	.211	-.280	.205	-.281			.050	
.075	.217	-.292	.223	-.328	.223	-.313	.202	-.276	.202	-.276	.165	-.277			.075	
.100	.224	-.118	.192	-.363	.191	-.311	.201	-.285	.202	-.285	.164	-.278			.100	
.150	.218	-.077	.179	-.392	.188	-.312	.187	-.306	.165	-.284	.164	-.283			.150	
.200	.152	-.088	.153	-.414	.182	-.331	.174	-.309	.166	-.278	.143	-.290			.200	
.250	.164	-.116	.149	-.317	.155	-.373	.162	-.307	.159	-.290	.141	-.292			.250	
.300	.152	-.091	.149	-.145	.155	-.393			.138	-.300	.132	-.288	.098	-.288	.300	
.350															.350	
.400	.184	-.127	.122	-.076	.145	-.403	.114	-.368	.119	-.306			.298	-.290	.400	
.450															.450	
.500	.164	-.080			.087	.095			.097	-.394	.091	-.314	.104	-.304	.500	
.550															.550	
.600	.089	-.112	.090	-.095	.071	-.179	.080	-.394	.068	-.355	.069	-.301	.063	-.301	.600	
.650															.650	
.700	.024		.044	-.084	.042	-.094	.049	-.341	.050	-.362	.054	-.306	.063	-.304	.700	
.750															.750	
.800	.041	-.081	.039	-.087	.033	-.083	.026	-.257	.039	-.367	.052	-.315			.800	
.850															.850	
.900	.029	-.084	.038	-.067	.033	-.064	.051	-.148	.053	-.372			.331		.900	
.950															.950	
$\alpha = -8^\circ$																
.0125	.208	-.341	.211	-.300	.192	-.275	.186	-.252	.176	-.241	.146	-.241	.143	-.246	.0125	
.050	.201	-.228	.196	-.293	.159	-.270	.172	-.255	.154	-.245	.181	-.243			.050	
.075	.182	-.080			.303	.183	-.270			.174	-.251	.168	-.238	.147	-.275	.075
.100	.182	-.074	.152	-.324	.149	-.279	.159	-.258	.174	-.248	.147	-.236			.100	
.150	.179	-.063	.140	-.343	.146	-.295	.149	-.265	.137	-.253	.130	-.239	.139	-.244	.150	
.200	.113	-.073	.116	-.304	.136	-.327	.140	-.279	.128	-.259	.137	-.241	.116	-.254	.200	
.250	.177	-.095	.107	-.068	.122	-.345	.123	-.304	.123	-.256	.111	-.255			.250	
.300	.114	-.073	.108	-.038					.106	-.256	.104	-.248	.069	-.253	.300	
.350															.350	
.400	.143	-.096			.068	.104	-.295	.079	-.339	.088	-.289			.253	-.255	.400
.450															.450	
.500	.111	-.063	.073	-.075	.060				.060	-.347	.060	-.311	.071	-.260	.500	
.550															.550	
.600	.054	-.082	.046	-.079	.037	-.055	.045	-.286	.041	-.327	.039	-.279	.032	-.267	.600	
.650															.650	
.700															.700	
.750															.750	
.800															.800	
.850															.850	
.900															.900	
.950															.950	
$\alpha = -6^\circ$																
.0125	.176	-.295	.188	-.267	.176	-.247	.163	-.228	.152	-.209	.131	-.207	.129	-.214	.0125	
.050	.190	-.051	.198	-.267	.155	-.241	.142	-.212	.176	-.207	.159	-.207			.050	
.075	.141	-.080			.279	.148	-.241			.227	.139	-.200	.122	-.202	.075	
.100	.147	-.074	.117	-.286	.113	-.246	.123	-.228	.114	-.209	.122	-.202			.100	
.150	.142	-.058	.104	-.194	.113	-.266	.112	-.232	.099	-.212	.096	-.204	.110	-.214	.150	
.200	.081	-.063	.081	-.031	.098	-.289	.105	-.247	.093	-.214	.104	-.203	.083	-.219	.200	
.250	.093	-.083	.073	-.043	.080	-.280	.091	-.269	.087	-.210	.078	-.205	.075	-.220	.250	
.300	.081	-.061	.074	-.046	.088	-.277	.076	-.269	.070	-.221	.072	-.205	.039	-.219	.300	
.350															.350	
.400	.107	-.069	.056	-.061	.070	-.015	.044	-.276	.057	-.244			.212	-.216	.400	
.450															.450	
.500	.096	-.039			.059	.031			.027	-.264	.026	-.256	.032	-.228	.500	
.550															.550	
.600	.019	-.068	.018	-.063	.008	-.057	.014	-.038	.008	-.250	.003	-.238	.000	-.225	.600	
.650															.650	
.700	-.027		-.013	-.064	-.015	-.053	-.013	-.022	-.009	-.215	-.006	-.242	-.003	-.226	.700	
.750															.750	
.800	-.015	-.042	-.020	-.057	-.024	-.062	-.019	-.038	-.019	-.156	-.009	-.254			.800	
.850															.850	
.900	-.025	-.045	-.020	-.037	-.028	-.045	-.019	-.018	-.015	-.046			.249		.900	
.950															.950	

TABLE V.—Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

TABLE V.—Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING
 $M_\infty = 1.61$.—Continued.

x/c, nominal	Cp or $y/\frac{c}{2}$ of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 4$																
.0125	-.020	.060	-.088	.081	-.111	.125	-.120	.155	-.152	.167	-.154	.159	-.129	.167	.0125	
.025	-.009	.067	-.061	.081	-.106	.096	-.124	.122	-.149	.146	-.151	.146	-.167	.050	.025	
.050	-.014	.062	-.043	.084	-.106	.082	-.102	.113	-.126	.126	-.149	.149	-.175	.075	.050	
.100	-.009	.054	-.045	.071	-.102	.082	-.134	.122	-.158	.126	-.172	.147	-.195	.100	.075	
.200	-.000	.041	-.045	.059	-.082	.074	-.099	.093	-.109	.100	-.126	.126	-.131	.114	.100	
.400	-.021	.055	-.047	.062	-.117	.092	-.182	.125	-.196	.108	-.212	.137	-.204	.092	.200	
.800	-.027	.024	-.045	.066	-.117	.074	-.192	.125	-.215	.129	-.219	.137	-.208	.097	.250	
.1600	-.030	.048	-.058	.065	-.167	.081	-.206	.126	-.227	.120	-.281	.111	-.315	.105	.300	
.3200	-.057	.079	-.085	.083	-.203	.109	-.227	.184	-.291	.076	-.319	.092	-.400	.400	.350	
.6500	-.048	.091	-.070	.109	-.219	.109	-.218	.171	-.215	.097	-.237	.071	-.324	.500	.500	
.13000	-.052	.093	-.094	.126	-.273	.132	-.286	.157	-.271	.174	-.282	.166	-.322	.600	.600	
.26000	-.038	.082	-.100	.100	-.251	.132	-.276	.153	-.276	.174	-.283	.186	-.322	.700	.700	
.52000	-.017	.078	-.088	.108	-.242	.124	-.294	.140	-.281	.161	-.292	.194	-.317	.800	.750	
.100000	.020	.091	-.090	.086	-.191	.114	-.300	.145	-.304	.157	-.295	.171	-.326	.155	.150	
.200000	-.064	.073	-.092	.089	-.291	.128	-.285	.151	-.316	.147	-.302	.160	-.327	.130	.200	
.400000	-.051	.048	-.090	.096	-.112	.260	-.131	-.325	.170	-.312	.160	-.331	.133	.250	.250	
.800000	-.066	.069	-.084	.092	-.203	.113	-.335	.160	-.322	.159	-.339	.146	-.300	.300	.350	
.1600000	-.081	.108	-.114	.110	-.174	.140	-.124	.115	-.335	.121	-.126	.136	-.341	.136	.450	
$a = 6$																
.0125	-.073	.103	-.229	.119	-.268	.140	-.278	.196	-.273	.129	-.287	.178	-.320	.193	.0125	
.025	-.053	.101	-.200	.119	-.273	.148	-.286	.157	-.271	.174	-.282	.166	-.350	.050	.025	
.050	-.042	.093	-.094	.126	-.251	.132	-.276	.153	-.276	.174	-.283	.186	-.322	.075	.050	
.100	-.017	.078	-.088	.108	-.242	.124	-.294	.140	-.281	.161	-.292	.194	-.317	.100	.100	
.200	-.020	.091	-.090	.086	-.191	.114	-.300	.145	-.304	.157	-.295	.171	-.326	.155	.150	
.400	-.064	.073	-.092	.089	-.291	.128	-.285	.151	-.316	.147	-.302	.160	-.327	.130	.200	
.800	-.051	.048	-.090	.096	-.112	.260	-.131	-.325	.170	-.312	.160	-.331	.133	.250	.250	
.1600	-.066	.069	-.084	.092	-.203	.113	-.335	.160	-.322	.159	-.339	.146	-.300	.300	.350	
.3200	-.081	.108	-.114	.110	-.174	.140	-.124	.115	-.335	.121	-.126	.136	-.341	.136	.450	
.6500	-.074	.118	-.093	.124	-.174	.140	-.120	.105	-.246	.134	-.352	.122	-.354	.500	.500	
.130000	-.112	.082	-.127	.086	-.148	.096	-.140	.098	-.122	.120	-.365	.099	-.371	.094	.600	
.260000	-.129	.137	-.075	.153	-.154	.084	-.154	.099	-.131	.111	-.354	.108	-.378	.112	.700	
.520000	-.126	.081	-.133	.079	-.153	.078	-.168	.095	-.140	.097	-.337	.103	-.325	.800	.800	
.1000000	-.142	.096	-.138	.101	-.152	.098	-.149	.122	-.134	.111	-.304	.084	-.317	.900	.900	
$a = 8$																
.0125	.126	.146	-.300	.159	-.297	.178	-.301	.224	-.295	.222	-.313	.194	-.339	.025	.0125	
.025	-.127	.138	-.287	.159	-.276	.176	-.301	.194	-.294	.202	-.309	.192	-.207	.050	.025	
.050	-.075	.128	-.276	.160	-.304	.168	-.307	.193	-.298	.209	-.309	.216	-.340	.075	.050	
.100	-.065	.113	-.243	.143	-.304	.175	-.316	.193	-.316	.199	-.316	.228	-.213	.100	.100	
.200	-.056	.107	-.151	.147	-.216	.167	-.312	.184	-.304	.199	-.323	.205	-.343	.185	.150	
.400	-.049	.123	-.086	.127	-.238	.153	-.310	.187	-.319	.194	-.328	.198	-.347	.162	.200	
.800	-.084	.108	-.106	.128	-.223	.170	-.336	.187	-.332	.185	-.337	.208	-.351	.171	.250	
.1600	-.079	.104	-.106	.137	-.189	.152	-.170	.169	-.317	.198	-.342	.193	-.360	.172	.350	
.3200	-.107	.142	-.132	.147	-.213	.182	-.351	.151	-.395	.162	-.171	.184	-.384	.168	.450	
.6500	-.102	.151	-.132	.132	-.217	.187	-.195	.148	-.395	.170	-.385	.149	-.467	.500	.500	
.130000	-.107	.142	-.132	.147	-.213	.182	-.351	.151	-.395	.162	-.171	.184	-.384	.168	.450	
.260000	-.121	.151	-.132	.132	-.217	.187	-.195	.148	-.395	.170	-.385	.149	-.467	.500	.500	
.520000	-.126	.151	-.132	.132	-.217	.187	-.195	.148	-.395	.170	-.385	.149	-.467	.500	.500	
.1000000	-.136	.116	-.147	.123	-.160	.131	-.120	.141	-.393	.151	-.404	.136	-.476	.600	.600	
$a = 10$																
.0125	-.231	.185	-.304	.195	-.320	.202	-.322	.245	-.315	.241	-.340	.204	-.358	.025	.0125	
.025	-.212	.177	-.315	.192	-.325	.208	-.326	.230	-.311	.227	-.334	.216	-.221	.050	.025	
.050	-.156	.174	-.317	.192	-.325	.208	-.326	.231	-.311	.227	-.334	.216	-.360	.075	.050	
.100	-.151	.184	-.325	.205	-.326	.213	-.327	.231	-.311	.227	-.334	.216	-.360	.100	.100	
.200	-.094	.142	-.340	.184	-.338	.203	-.334	.220	-.325	.221	-.336	.216	-.374	.150	.150	
.400	-.100	.163	-.285	.167	-.378	.194	-.362	.225	-.347	.227	-.356	.216	-.385	.227	.227	
.800	-.109	.139	-.299	.162	-.390	.196	-.370	.220	-.351	.221	-.365	.215	-.395	.194	.250	
.1600	-.105	.116	-.105	.174	-.391	.190	-.213	-.367	-.241	-.356	-.364	-.366	-.375	-.304	.350	
.3200	-.097	.137	-.106	.174	-.391	.190	-.213	-.367	-.241	-.356	-.364	-.366	-.375	-.204	.400	
.6500	-.136	.180	-.147	.184	-.262	.221	-.409	.192	-.399	.208	-.215	.181	-.404	.400	.450	
.130000	-.123	.191	-.169	.146	-.399	.192	-.420	.204	-.385	.187	-.419	.192	-.492	.500	.500	
.260000	-.154	.151	-.169	.158	-.366	.169	-.436	.177	-.427	.185	-.446	.171	-.492	.600	.600	
.520000	-.164	.151	-.169	.146	-.373	.171	-.416	.170	-.407	.168	-.423	.175	-.492	.800	.800	
.1000000	-.173	.167	-.173	.174	-.390	.171	-.418	.192	-.421	.180	-.427	.156	-.492	.900	.900	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

(a) M = 1.61 - Continued

x/c, nominal	Cp at $y/\frac{c}{2}$ of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 12^\circ$																
.0125	-+.336	.218	-.340		-.339	.220	-.346	.260	-.343	.254	-.362	.220	-.375		.0125	
.025	-.355	.202	-.335	.239	-.341	.234	-.346	.257	-.351	.251	-.358	.249	-.375	.237	.025	
.050	-.346	.203	-.342	.250	-.349	.244	-.346	.271	-.347	.269	-.358	.273	-.375	.050	.050	
.075	-.306	.189	-.384	.224	-.349	.233	-.356	.264	-.347	.263	-.360	.275	-.375	.075	.075	
.100	-.291	.181	-.384	.224	-.349	.230	-.356	.261	-.367	.259	-.366	.272	-.376	.100	.100	
.125	-.285	.193	-.390	.201	-.380	.231	-.368	.264	-.376	.259	-.369	.276	-.379	.125	.125	
.150	-.274	.174	-.336	.200	-.409	.254	-.378	.264	-.376	.259	-.369	.276	-.379	.150	.150	
.175	-.263	.151	-.156	.215	-.441	.226	-.395	.218	-.375	.279	-.375	.269	-.381	.226	.250	
.200	-.250	.116	-.174	.216	-.441	.226	-.395	.252	-.382	.267	-.386	.269	-.387	.236	.300	
.225	-.235	.116	-.126	.215	-.441	.226	-.395	.252	-.382	.267	-.386	.269	-.387	.236	.350	
.250	-.220	.116	-.174	.215	-.441	.226	-.395	.252	-.382	.267	-.386	.269	-.387	.241	.400	
.275	-.205	.116	-.217	-.159	.223	-.419	.254	-.434	.229	-.395	.242	-.396	.256	-.397	.450	
.300	-.190	.116	-.227		.208	-.326		-.457	.225	-.416	.249	-.390	.224	-.412	.500	
.325	-.175	.116	-.227		.208	-.326		-.457	.208	-.440	.220	-.405	.209	-.407	.550	
.350	-.160	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.600	
.375	-.145	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.650	
.400	-.130	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.700	
.425	-.115	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.750	
.450	-.100	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.800	
.475	-.085	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.850	
.500	-.070	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.900	
.525	-.055	.116	-.190	-.180	.197	-.228	.203	-.434	.208	-.440	.220	-.405	.209	-.407	.950	
$\alpha = 14^\circ$																
.0125	-.38%	.247	-.353		-.383	.246	-.366	.276	-.362	.274	-.379	.293	-.387		.0125	
.025	-.386	.247	-.347	.271	-.367	.271	-.366	.284	-.359	.281	-.374	.262	-.387	.249	.025	
.050	-.296	.252	-.360	.286	-.358	.291	-.366	.303	-.368	.300	-.375	.293	-.390	.279	.050	
.075	-.135	.256	-.362	.281	-.362	.294	-.366	.303	-.368	.304	-.375	.304	-.390	.279	.075	
.100	-.102	.256	-.402	.264	-.380	.274	-.367	.297	-.368	.297	-.375	.304	-.390	.100	.100	
.125	-.107	.267	-.431	.242	-.382	.274	-.380	.294	-.373	.299	-.379	.309	-.390	.125	.125	
.150	-.137	.214	-.431	.242	-.402	.297	-.400	.298	-.397	.293	-.379	.303	-.393	.147	.200	
.175	-.124	.189	-.338	.257	-.407	.260	-.407	.271	-.404	.216	-.388	.293	-.395	.255	.255	
.200	-.131	.216	-.233	.259	-.453	.271	-.436	.289	-.410	.310	-.399	.302	-.399	.270	.340	
.225	-.187	.263	-.192	.262	-.464	.294	-.436	.278	-.418	.279	-.429	.279	-.403	.277	.400	
.250	-.150	.274	-.191	.262	-.421		-.461	.262	-.410	.281	-.425	.261	-.413	.500	.500	
.275	-.187	.232	-.191	.239	-.356	.263	-.470	.240	-.441	.256	-.470	.247	-.425	.253	.600	
.300	-.197	.232	-.201	.216	-.372	.233	-.450	.263	-.446	.258	-.433	.267	-.430	.273	.700	
.325	-.198	.237	-.199	.236	-.274	.246	-.426	.262	-.451	.266	-.435	.274		.800		
.350	-.207	.277	-.207	.276	-.251	.273	-.388	.290	-.457	.286	-.439	.261		.850		
$\alpha = 16^\circ$																
.0125	-.388	.279	-.368		-.374	.262	-.384	.301	-.375	.301	-.384	.300	-.397	.244	-.411	.0125
.025	-.390	.288	-.358	.306	-.371	.309	-.378	.326	-.370	.326	-.384	.321	-.391	.274	.025	
.050	-.390	.291	-.378	.328	-.371	.320	-.375	.342	-.367	.344	-.394	.330	-.411	.050	.050	
.075	-.363	.276	-.376	.303	-.375	.330	-.379	.334	-.377	.334	-.395	.344	-.395	.075	.075	
.100	-.196	.245	-.410	.294	-.379	.315	-.379	.333	-.386	.333	-.395	.344	-.395	.100	.100	
.125	-.214	.346	-.448	.286	-.428	.325	-.406	.337	-.406	.337	-.406	.340	-.411	.125	.125	
.150	-.214	.247	-.458	.286	-.427	.311	-.422	.346	-.405	.336	-.406	.340	-.411	.150	.150	
.175	-.150	.225	-.427	.293	-.430	.314	-.439	.346	-.406	.346	-.406	.343	-.413	.200	.200	
.200	-.180	.244	-.427	.293	-.430	.314	-.439	.346	-.406	.346	-.406	.343	-.414	.237	.250	
.225	-.213	.301	-.271	.311	-.446	.362	-.439	.318	-.439	.325	-.437	.318	-.418	.237	.340	
.250	-.185	.320		.300	-.455		-.446	.306	-.434	.335	-.432	.326	-.420	.500	.500	
.275	-.212	.271	-.224	.283	-.419	.298	-.455	.296	-.437	.327	-.446	.321	-.428	.324	.600	
.300	-.216	.233	-.223	.273	-.381	.303	-.468	.319	-.438	.335	-.444	.344	-.434	.353	.700	
.325	-.216	.296	-.223	.306	-.345	.315	-.453	.337	-.441	.341	-.440	.349		.800		
.350	-.220	.317	-.223	.364	-.412	.364	-.430	.387	-.449	.383	-.467	.367		.900		
.375	-.228	.348	-.347	.357	-.467	.384	-.444	.390	-.448	.397	-.466	.398		.950		
$\alpha = 18^\circ$																
.0125	-.307	.320	-.329		-.393	.332	-.401	.323	-.404	.325	-.408	.285	-.419		.0125	
.025	-.304	.324	-.320	.346	-.393	.347	-.401	.351	-.407	.348	-.406	.342	-.409	.050	.050	
.050	-.1477	.333	-.389	.360	-.393	.366	-.401	.376	-.403	.377	-.405	.375	-.419	.075	.075	
.075	-.435	.328	-.346	.346	-.393	.350	-.401	.376	-.403	.375	-.405	.380	-.419	.100	.100	
.100	-.134	.312	-.440	.350	-.400	.356	-.401	.376	-.403	.375	-.405	.389	-.417	.135	.135	
.125	-.164	.334	-.446	.326	-.427	.367	-.408	.379	-.410	.370	-.410	.394	-.417	.165	.165	
.150	-.180	.330	-.447	.330	-.445	.385	-.426	.389	-.405	.389	-.406	.403	-.420	.200	.200	
.175	-.167	.274	-.460	.350	-.450	.342	-.450	.365	-.456	.349	-.456	.414	-.454	.237	.250	
.200	-.174	.308	-.422	.356	-.450	.365	-.454	.384	-.450	.347	-.452	.408	-.454	.388	.388	
.225	-.234	.348	-.347	.357	-.467	.384	-.444	.390	-.448	.397	-.466	.396	-.423	.398	.400	
.250	-.216	.365	-.244	.357	-.473		-.449	.386	-.441	.402	-.440	.393	-.426	.450	.450	
.275	-.226	.327	-.275	.348	-.451	.358	-.455	.367	-.461	.394	-.457	.402	-.431	.464	.450	
.300	-.225	.327	-.257	.345	-.421	.378	-.474	.399	-.462	.425	-.452	.456	-.440	.542	.450	
.325	-.225	.369	-.244	.370	-.488	.401	-.467	.430	-.443	.497	-.464	.558		.700	.700	
.350	-.225	.369	-.244	.370	-.488	.401	-.467	.430	-.443	.497	-.464	.558		.800	.800	
.375	-.228	.401	-.238	.343	-.474	.571	-.450	.596	-.451	.605	-.455	.574		.900	.900	
.400	-.228	.401	-.238	.343	-.474	.571	-.450	.596	-.451	.605	-.455	.574		.950	.950	

TABLE V. - Continued
 PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING
 (a) $M = 1.61$ - Concluded

x/c, nominal	C_p at $y/\frac{c}{2}$ of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 20^\circ$																
.3125	-.391	.353	-.382	.381	-.405	.314	.366	-.411	.340	-.407	.374	-.410	.312	-.423	.0125	
.375	-.377	.379	-.389	.415	-.405	.385	.411	-.411	.382	-.403	.405	-.409	.384	-.422	.050	
.4375	-.441	.372	-.391	.415	-.405	.408	.408	-.408	.408	-.439	.439	-.407	.423	-.422	.075	
.500	-.455	.372	-.391	.415	-.405	.408	.413	-.416	.437	-.414	.439	-.407	.439	-.422	.100	
.5625	-.418	.362	-.453	.396	-.408	.395	.410	-.418	.408	-.408	.439	-.411	.442	-.420	.125	
.625	-.282	.384	-.441	.372	-.425	.413	.416	-.437	.414	-.439	.411	-.411	.442	-.422	.150	
.6875	-.227	.347	-.444	.389	-.451	.427	.434	-.459	.459	-.409	.472	-.409	.453	-.423	.175	
.750	-.208	.323	-.458	.406	-.452	.393	.458	-.414	.493	-.420	.493	-.422	.498	-.422	.200	
.8125	-.204	.359	-.468	.403	-.452	.436	.461	-.433	.501	-.416	.472	-.423	.533	-.400	.225	
.875	-.255	.399	-.390	.423	-.456	.466	.451	-.461	.458	-.452	.470	-.426	.566	-.400	.250	
.9375	-.216	.429	-.441	.480	-.470	.421	.460	-.441	.448	-.472	.454	-.430	.522	-.430	.275	
.999	-.240	.411	-.305	.422	-.470	.421	.460	-.441	.449	-.516	.466	-.437	.572	-.437	.300	
.000	-.233	-.286	.428	-.445	.475	-.480	.533	-.445	.616	-.458	.608	-.447	.595	-.700	.325	
.0625	-.228	.476	-.262	.523	-.416	.594	-.478	.623	-.446	.620	-.450	.614	-.450	.800	.350	
.125	-.234	.601	-.258	.619	-.400	.623	-.463	.640	-.451	.630	-.450	.580	-.450	.850	.375	
.1875	-.234	.601	-.258	.619	-.400	.623	-.463	.640	-.451	.630	-.450	.580	-.450	.900	.400	
.250	-.234	.601	-.258	.619	-.400	.623	-.463	.640	-.451	.630	-.450	.580	-.450	.950	.425	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

(b) M = 2.01

%	Cp at $y/\frac{c}{2}$ of:														%		
	+10		+25		+40		+50		+60		+70		+80				
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower			
$\alpha = -20^\circ$																	
.0125	.431	-.240	.385	-.234	.313	-.235	.242	-.237	.230	.324	-.240	.270	-.230	.249	-.249	.0125	
.025	.474	-.230	.436	-.229	.240	.365	.246	.361	.232	.361	.237	.315	-.240	-.231	.025		
.050	.464	-.239	.429	-.240	.240	.407	.238	.238	.236	.358	.234	.351	-.236	.296	-.231	.050	
.075	.447	-.248	.403	-.241	.372	-.245	.364	-.237	.372	.366	.232	.329	-.236	.239	-.227	.075	
.100	.458	-.238	.403	-.241	.372	-.245	.364	-.237	.372	.366	.232	.329	-.236	.239	-.227	.100	
.150	.439	-.234	.387	-.245	.364	-.240	.367	-.241	.355	.345	.235	.327	-.227	.306	-.227	.150	
.200	.358	-.233	.359	-.250	.370	-.240	.355	-.234	.344	.330	.233	.308	-.230	.301	-.237	.200	
.250	.370	-.213	.347	-.250	.370	-.240	.349	-.234	.349	.335	.233	.308	-.229	.293	-.246	.250	
.300	.363	-.201	.342	-.247	.342	-.256	-.234	-.234	.317	.235	.304	.242	.261	-.236	.300	.350	
.350	.394	-.194	.328	-.243	.337	-.248	.291	-.244	.306	.235	-.240	.274	-.235	.400		.400	
.400	.374	-.174	.276	-.229	.288	-.284	.250	-.250	.275	-.232	.281	-.246	.256	-.256	.450	.450	
.500	.374	-.174	.266	-.241	.248	-.253	.254	-.254	.265	-.241	.232	-.236	.229	-.242	.500	.550	
.600	.248	-.185	.266	-.241	.248	-.253	.254	-.254	.265	-.241	.232	-.228	.225	-.236	.600	.650	
.650	.166		.206	-.235	.207	-.252	.214	-.246	.217	-.243	.219	-.228	.225	-.236	.700	.750	
.750	.196	-.154	.201	-.224	.194	-.251	.186	-.240	.200	-.229	.214	-.250	-.250		.800	.850	
.800	.186	-.153	.195	-.196	.187	-.247	.197	-.238	.204	-.241	-.245				.900	.950	
$\alpha = -18^\circ$																	
.0125	.402	-.262	.349	-.262	.292	-.263	.264	-.264	.310	-.259	.297	.259	.249	-.258	.231	.0125	
.025	.427	-.261	.392	-.261	.327	-.267	.362	-.267	.323	.335	.260	.289	-.264	-.260	.025		
.050	.427	-.274	.380	-.267	.327	-.267	.362	-.267	.323	.327	.258	.313	-.259	.270	.050		
.075	.394	-.246	.376	-.267	.362	-.267	.351	-.267	.321	.327	.257	.290	-.260	-.265	.075		
.100	.401	-.276	.351	-.267	.332	-.272	.318	-.264	.318	.324	.255	.290	-.255	.279	-.259	.100	
.150	.383	-.270	.332	-.277	.318	-.262	.318	-.264	.318	.325	.255	.290	-.255	.279	-.259	.150	
.200	.309	-.254	.305	-.281	.321	-.269	.313	-.264	.329	.321	.261	.291	-.262	.269	-.265	.200	
.250	.315	-.214	.299	-.281	.321	-.287	.303	-.284	.320	.328	.260	.277	-.263	.264	-.250	.250	
.300	.304	-.197	.291	-.282	.299	-.284	.284	-.284	.260	-.277	.259	.270	-.264	.233	-.258	.300	
.350	.342	-.196	.281	-.278	.292	-.283	.249	-.281	.264	-.261	-.263	.239	-.259	-.259	.400	.400	
.400	.332	-.161		-.264	.244	-.240	-.281	-.281	.234	-.264	-.247	-.265	.224	-.265	.500	.550	
.500	.200	-.160	.217	-.263	.208	-.288	.211	-.281	.205	-.276	.204	-.260	.195	-.265	.600	.650	
.600	.128		.165	-.246	.165	-.287	.172	-.275	.178	-.275	.184	-.257	.189	-.260	.700	.750	
.700	.152	-.146	.156	-.215	.151	-.286	.147	-.272	.162	-.257	.178	-.276	.169	-.275	-.275	.800	.850
.800	.152	-.157	.152	-.189	.163	-.279	.153	-.274	.165	-.275						.900	.950
$\alpha = -16^\circ$																	
.0125	.372	-.264	.306	-.267	.273	-.268	.284	-.267	.270	-.255	.277	.260	.209			.0125	
.025	.381	-.265	.348	-.267	.267	-.268	.285	-.267	.285	.267	.294	.255	.281	-.260	.243	.025	
.050	.363	-.278	.339	-.271	.297	-.272	.322	-.269	.323	.267	.294	.255	.281	-.262	.243	.050	
.075	.349	-.285		-.272	.327	-.274	.293	-.274	.282	.267	.293	.256	.263	-.260	-.264	.075	
.100	.347	-.285	.308	-.276	.293	-.274	.282	-.274	.271	.267	.276	.256	.258	-.258	.246	-.260	.100
.150	.337	-.277	.291	-.285	.277	-.267	.267	-.267	.271	.267	.276	.256	.258	-.258	.246	-.260	.150
.200	.268	-.221	.267	-.293	.282	-.273	.267	-.267	.260	.259	.263	.256	.263	-.260	.234	-.265	.200
.250	.277	-.176	.257	-.293	.279	-.271	.291	-.271	.260	.256	.259	.249	.242	-.229	.229	-.250	.250
.300	.265	-.152	.251	-.293	.260	-.298	.264	-.264	.240	.255	.235	.262	.200	-.262	.226	-.250	.300
.350	.294	-.164	.244	-.284	.248	-.291	.211	-.289	.228	-.258	-.263	.204	-.264			.400	.400
.400	.311	-.132		-.273	.202	-.296	.200	-.294	.200	-.265	.206	-.263	.189			.500	.550
.500	.301	-.132		-.273	.202	-.296	.200	-.294	.200	-.265	.206	-.263	.189			.600	.650
.600	.169	-.185	.185	-.244	.171	-.296	.173	-.289	.173	-.281	.167	-.260	.161	-.266	-.266	.600	.650
.700	.089		.136	-.205	.131	-.305	.138	-.289	.144	-.281	.151	-.264	.156	-.263	-.263	.700	.750
.800	.116	-.133	.123	-.173	.117	-.294	.108	-.278	.111	-.259	.143	-.280	.134	-.285		.800	.850
.900	.116	-.133	.122	-.144	.107	-.285	.113	-.280	.101	-.276						.900	.950
$\alpha = -14^\circ$																	
.0125	.331	-.254	.279	-.261	.249	-.264	.259	-.259	.243	-.240	.245	.214	-.245	.216	-.245	.0125	
.025	.342	-.263	.310	-.266	.266	-.265	.274	-.265	.270	-.261	.264	.214	-.245	.216	-.245	.025	
.050	.316	-.277	.298	-.266	.266	-.265	.281	-.266	.274	-.261	.264	.214	-.245	.216	-.245	.050	
.075	.317	-.280		-.266	.271	-.271	.276	-.271	.264	-.261	.264	.214	-.245	.216	-.245	.075	
.100	.282	-.280	.264	-.271	.250	-.276	.248	-.271	.250	-.261	.264	.214	-.245	.216	-.245	.100	
.150	.292	-.262	.266	-.283	.253	-.266	.233	-.262	.222	-.264	.225	.215	-.248	.229	-.250	.150	
.200	.224	-.153	.220	-.294	.238	-.266	.227	-.267	.221	-.267	.214	-.246	.216	-.250	.200	.250	.200
.250	.232	-.110	.210	-.297	.214	-.277	.224	-.277	.219	-.249	.219	.215	-.246	.208	-.250	.200	.250
.300	.223	-.118	.206	-.292	.214	-.299	.206	-.296	.200	-.247	.209	-.246	.180	-.250	-.250	.300	.350
.350	.249	-.129	.198	-.279	.200	-.294	.177	-.284	.194	-.253	-.245	.178	-.254	-.254	.400	.450	.400
.400	.316	-.177	.208	-.210	.161	-.261	.200	-.261	.167	-.260	.170	-.250	.161	-.250		.500	.550
.500	.267	-.112		-.210	.192	-.204	.139	-.293	.140	-.270	.142	-.252	.140	-.252		.600	.650
.600	.135	-.119	.144	-.176	.192	-.204	.139	-.293	.140	-.274	.141	-.254	.142	-.254		.700	.750
.700	.055		.094	-.144	.093	-.204	.104	-.284	.111	-.274	.128	-.255	.114	-.269		.800	.850
.800	.079	-.116	.084	-.134	.065	-.208	.079	-.285	.098	-.252	.119	-.269	.111	-.270		.900	.950
.900	.079	-.124	.080	-.110	.070	-.258	.082	-.280	.095	-.274						.900	.950

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

%	Cap. at $\sqrt{\frac{b}{2}}$ of:														%	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
a = -12																
.0125	.304	-.240	.251	-.227	.253	-.252	.231	-.241	.224	-.215	.200	-.232	.186	-.237	.0125	
.025	.305	-.249	.277	-.260	.233	-.245	.226	-.243	.254	-.218	.224	-.230	.025	-.025	.050	
.050	.271	-.273	.263	-.259	.237	-.251	.211	-.257	.224	-.217	.224	-.229	.050	-.075	.075	
.075	.260	-.283	.259	-.259	.237	-.251	.211	-.257	.224	-.220	.210	-.228	.075	-.100	.100	
.100	.242	-.283	.227	-.266	.211	-.257	.211	-.245	.194	-.222	.194	-.230	.200	-.240	.150	
.125	.251	-.191	.210	-.281	.193	-.256	.194	-.245	.194	-.222	.194	-.230	.200	-.240	.150	
.150	.184	-.092	.186	-.292	.190	-.257	.190	-.251	.192	-.225	.197	-.230	.184	-.240	.200	
.175	.194	-.103	.175	-.305	.176	-.268	.186	-.247	.187	-.225	.181	-.231	.176	-.236	.250	
.200	.183	-.093	.176	-.315	.172	-.296	.172	-.247	.169	-.225	.178	-.231	.150	-.237	.300	
.250	.200	-.193	.161	-.253	.157	-.292	.138	-.274	.166	-.234	.147	-.231	.145	-.241	.400	
.300	.227	-.086	.172	-.128	.132	-.287	.137	-.238	.111	-.246	.113	-.244	.105	-.243	.500	
.350	.227	-.086	.172	-.128	.132	-.287	.137	-.238	.111	-.246	.113	-.244	.105	-.243	.600	
.400	.227	-.086	.172	-.128	.132	-.287	.137	-.238	.111	-.246	.113	-.244	.105	-.243	.700	
.450	.227	-.086	.172	-.128	.132	-.287	.137	-.238	.111	-.246	.113	-.244	.105	-.243	.800	
.500	.227	-.086	.172	-.128	.132	-.287	.137	-.238	.111	-.246	.113	-.244	.105	-.243	.900	
.550	.227	-.086	.172	-.128	.132	-.287	.137	-.238	.111	-.246	.113	-.244	.105	-.243	.950	
.600	.108	-.101	.112	-.120	.109	-.305	.109	-.287	.111	-.246	.113	-.244	.105	-.243	.650	
.650	.029	-.068	.112	-.066	.112	-.284	.109	-.282	.091	-.252	.098	-.247	.098	-.246	.700	
.700	.029	-.068	.112	-.066	.112	-.284	.109	-.282	.091	-.252	.098	-.247	.098	-.246	.750	
.750	.049	-.097	.057	-.103	.049	-.238	.053	-.285	.073	-.239	.090	-.251	.083	-.256	.800	
.800	.049	-.097	.057	-.103	.049	-.238	.053	-.285	.073	-.239	.090	-.251	.083	-.256	.850	
.850	.045	-.103	.052	-.091	.040	-.174	.057	-.283	.073	-.262	.083	-.256	.083	-.256	.900	
.900	.045	-.103	.052	-.091	.040	-.174	.057	-.283	.073	-.262	.083	-.256	.083	-.256	.950	
a = -10																
.0125	.257	-.228	.221	-.209	.242	-.239	.206	-.226	.195	-.197	.184	-.208	.172	-.213	.0125	
.025	.248	-.234	.243	-.249	.200	-.238	.193	-.228	.222	-.200	.205	-.209	.179	-.215	.025	
.050	.214	-.260	.216	-.249	.200	-.238	.193	-.228	.195	-.199	.194	-.205	.179	-.215	.050	
.075	.208	-.254	.216	-.249	.204	-.238	.197	-.228	.199	-.199	.200	-.207	.179	-.215	.075	
.100	.200	-.251	.182	-.246	.176	-.238	.172	-.230	.189	-.199	.186	-.207	.172	-.212	.100	
.125	.201	-.249	.163	-.242	.151	-.241	.150	-.231	.163	-.201	.169	-.209	.172	-.216	.150	
.150	.201	-.249	.163	-.242	.151	-.241	.150	-.231	.163	-.201	.169	-.209	.172	-.216	.150	
.200	.073	-.141	.173	-.173	.146	-.249	.110	-.232	.168	-.207	.152	-.209	.156	-.216	.200	
.250	.147	-.090	.135	-.280	.132	-.277	.132	-.270	.137	-.209	.138	-.210	.126	-.217	.300	
.300	.141	-.081	.132	-.280	.132	-.277	.132	-.270	.137	-.209	.138	-.210	.126	-.217	.350	
.350	.153	-.089	.116	-.120	.116	-.278	.090	-.271	.127	-.215	.116	-.211	.121	-.219	.400	
.400	.153	-.089	.116	-.120	.116	-.278	.090	-.271	.127	-.215	.116	-.211	.121	-.219	.450	
.450	.175	-.075	.092	-.092	.075	-.283	.098	-.283	.108	-.226	.116	-.219	.102	-.219	.500	
.500	.071	-.086	.073	-.082	.068	-.272	.074	-.283	.083	-.238	.081	-.224	.076	-.219	.600	
.600	.071	-.086	.073	-.082	.068	-.272	.074	-.283	.083	-.238	.081	-.224	.076	-.219	.650	
.650	.006	-.033	.019	-.096	.019	-.114	.024	-.270	.053	-.232	.063	-.235	.053	-.245	.700	
.700	.013	-.083	.019	-.096	.019	-.114	.024	-.270	.053	-.232	.063	-.235	.053	-.245	.750	
.750	.017	-.088	.019	-.077	.013	-.084	.024	-.240	.046	-.250	.046	-.250	.046	-.250	.800	
.800	.017	-.088	.019	-.077	.013	-.084	.024	-.240	.046	-.250	.046	-.250	.046	-.250	.850	
.850	.017	-.088	.019	-.077	.013	-.084	.024	-.240	.046	-.250	.046	-.250	.046	-.250	.900	
.900	.017	-.088	.019	-.077	.013	-.084	.024	-.240	.046	-.250	.046	-.250	.046	-.250	.950	
a = -8																
.0125	.216	-.201	.203	-.194	.224	-.226	.186	-.213	.164	-.191	.156	-.197	.148	-.202	.0125	
.025	.211	-.212	.212	-.229	.212	-.227	.159	-.223	.162	-.213	.153	-.190	.175	-.197	.050	
.050	.176	-.232	.187	-.227	.163	-.226	.163	-.214	.163	-.214	.158	-.191	.154	-.194	.075	
.075	.165	-.206	.226	-.226	.122	-.226	.143	-.215	.151	-.192	.148	-.193	.132	-.208	.100	
.100	.158	-.182	.147	-.229	.140	-.226	.120	-.216	.118	-.215	.123	-.191	.129	-.196	.150	
.125	.157	-.061	.128	-.229	.122	-.226	.122	-.216	.118	-.215	.123	-.191	.132	-.208	.200	
.150	.105	-.067	.112	-.229	.118	-.232	.118	-.216	.116	-.215	.116	-.193	.130	-.196	.220	
.200	.121	-.082	.099	-.217	.092	-.237	.111	-.217	.113	-.192	.107	-.197	.111	-.203	.250	
.250	.104	-.068	.093	-.175	.100	-.243	.112	-.212	.100	-.196	.100	-.197	.089	-.204	.300	
.300	.114	-.081	.081	-.015	.086	-.246	.064	-.240	.087	-.206	.087	-.197	.084	-.206	.350	
.350	.114	-.081	.081	-.015	.086	-.246	.064	-.240	.087	-.206	.087	-.197	.084	-.206	.400	
.400	.129	-.064	.074	-.074	.066	-.246	.069	-.250	.071	-.217	.081	-.202	.066	-.205	.450	
.450	.049	-.070	.047	-.071	.041	-.147	.049	-.246	.048	-.226	.046	-.207	.041	-.205	.500	
.500	.049	-.070	.047	-.071	.041	-.147	.049	-.246	.048	-.226	.046	-.207	.041	-.205	.550	
.550	.021	-.008	.008	-.080	.013	-.055	.023	-.232	.031	-.231	.036	-.214	.035	-.205	.600	
.600	.008	-.064	.006	-.077	.006	-.071	.000	-.197	.020	-.220	.031	-.222	.024	-.228	.650	
.650	.013	-.072	.006	-.060	.014	-.064	.000	-.078	.013	-.231	.013	-.228	.014	-.228	.700	
.700	.013	-.072	.006	-.061	.014	-.067	.006	-.077	.013	-.231	.013	-.228	.014	-.228	.750	
.750	.013	-.072	.006	-.061	.014	-.067	.006	-.077	.013	-.231	.013	-.228	.014	-.228	.800	
.800	.013	-.072	.006	-.061	.014	-.067	.006	-.077	.013	-.231	.013	-.228	.014	-.228	.850	
.850	.013	-.072	.006	-.061	.014	-.067	.006	-.077	.013	-.231	.013	-.228	.014	-.228	.900	
.900	.013	-.072	.006	-.061	.014	-.067	.006	-.077	.013	-.231	.013	-.228	.014	-.228	.950	
a = -6																
.0125	.182	-.164	.173	-.218	.168	-.206	.151	-.197	.136	-.173	.129	-.166	.135	-.168	.0125	
.025	.171	-.181	.180	-.218	.153	-.209	.127	-.193	.152	-.172	.146	-.166	.119	-.168	.025	
.050	.137	-.190	.147	-.202	.127	-.209	.127	-.193	.121	-.171	.119	-.164	.119	-.173	.050	
.075	.127	-.117	.185	-.185	.131	-.209	.143	-.193	.106	-.171	.108	-.162	.106	-.173	.075	
.100	.127	-.093	.111	-.181	.103	-.208	.106	-.193	.095	-.171	.098	-.165	.096	-.173	.100	
.125	.127	-.056	.092	-.156	.047	-.046	.078	-.199	.071	-.191	.082	-.169	.094	-.177	.125	
.150	.070	-.060	.072	-.122	.078	-.199	.071	-.191	.076	-.187	.078	-.168	.079	-.177	.150	
.200	.017	-.070	.070	-.061	.066	-.177	.071	-.193	.071	-.183	.064	-.167	.059	-.176	.200	
.250	.073	-.080	.048	-.037	.048	-.144	.019	-.191	.049	-.176	.076	-.162	.054	-.176	.300	
.300	.054	-.050	.024	-.058	.016	-.067	.016	-.150	.017	-.180	.020	-.169	.017	-.171	.350	
.350	.024	-.059	.024	-.058	.016	-.067	.016	-.150	.017	-.180	.020	-.169	.017	-.171	.400	
.400	.054	-.050	.024	-.058	.016	-.067	.016	-.150	.017	-.180	.020	-.169	.017	-.171	.	

TABLE V - Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

(b) M = 2.01 - Continued

x/c , nominal	Cp at y/δ of:														x/c , nominal	
	+10		+25		+40		+50		+50		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -4$																
.0125	.152	-.116	.145	.145	.142	-.189	.116	-.177	.110	-.160	.103	-.161	.096	-.168	.0125	
.025	.143	-.115	.149	.117	.143	-.093	.186	-.186	.112	-.156	.105	-.162	.050	-.025	.050	
.050	.138	-.127	.117	.113	.086	-.186	.093	-.177	.093	-.155	.089	-.158	.081	-.169	.075	
.075	.094	-.062	.082	.090	.071	-.184	.071	-.176	.077	-.155	.079	-.157	.100	-.169	.100	
.100	.104	-.050	.082	.090	.071	-.184	.071	-.176	.077	-.155	.079	-.157	.112	-.169	.112	
.150	.090	-.060	.065	.078	.052	-.133	.049	-.153	.056	-.139	.064	-.152	.051	-.172	.150	
.200	.068	-.040	.045	.065	.049	-.096	.038	-.128	.051	-.122	.059	-.139	.042	-.175	.200	
.250	.054	-.056	.038	.051	.037	-.078	.044	-.106	.046	-.113	.043	-.121	.035	-.155	.250	
.300	.049	-.038	.025	.052	.037	-.060	-.098	-.036	-.098	-.030	-.121	-.025	-.140	-.300	.350	
.350	.048	-.038	.016	.034	.013	-.043	-.010	-.078	.023	-.090	-.014	-.015	-.132	-.400	.450	
.400	.027	-.030	-.055	-.002	-.003	-.073	-.005	-.080	-.002	-.108	-.013	-.000	-.150	-.500	.550	
.500	.004	-.037	-.005	-.046	-.017	-.056	-.017	-.068	-.010	-.073	-.015	-.103	-.023	-.120	.600	
.600	.004	-.050	-.029	-.052	-.030	-.050	-.029	-.049	-.020	-.064	-.024	-.098	-.023	-.111	.700	
.650	.004	-.041	-.035	-.046	-.051	-.045	-.061	-.045	-.042	-.025	-.055	-.024	-.090	-.800	.850	
.700	.004	-.045	-.039	-.047	-.038	-.060	-.049	-.047	-.029	-.026	-.050	-.029	-.087	-.900	.950	
$\alpha = -2$																
.0125	.110	-.087	.104	.110	.110	-.114	.074	-.117	.064	-.078	.060	-.071	.043	-.049	.0125	
.025	.134	-.062	.110	-.091	-.108	-.059	-.102	-.053	-.112	-.066	-.057	-.055	-.049	-.050	.025	
.050	.074	-.054	.080	-.078	-.059	-.043	-.077	-.043	-.102	-.049	-.072	-.040	-.051	-.030	.075	
.075	.068	-.025	-.065	-.043	-.043	-.077	-.043	-.102	-.049	-.066	-.033	-.042	-.036	-.100	.100	
.100	.060	-.028	.045	-.030	-.030	-.054	-.032	-.030	-.036	-.026	-.021	-.032	.012	-.040	.150	
.150	.068	-.018	.027	-.036	-.019	-.038	-.013	-.034	-.025	-.019	-.013	-.023	-.018	-.025	.200	
.200	.020	-.023	.012	-.028	-.019	-.028	-.009	-.019	-.013	-.023	-.013	-.025	-.009	-.044	.250	
.250	.029	-.035	.012	-.072	-.022	-.039	-.008	-.024	-.008	-.012	-.013	-.026	-.005	-.047	.300	
.300	.022	-.026	-.001	-.077	-.004	-.032	-.003	-.024	-.003	-.011	-.003	-.026	-.007	-.047	.350	
.350	.019	-.021	-.008	-.013	-.017	-.011	-.039	-.024	-.008	-.020	-.027	-.013	-.047	-.400	.450	
.400	.007	-.007	-.007	-.028	-.029	-.033	-.039	-.026	-.022	-.025	-.037	-.040	-.500	-.550	.550	
.500	.007	-.017	-.019	-.030	-.022	-.046	-.036	-.039	-.035	-.040	-.029	-.048	-.044	-.051	.600	
.600	.007	-.066	-.048	-.028	-.056	-.029	-.056	-.026	-.026	-.048	-.023	-.047	-.056	-.700	.750	
.700	.005	-.053	-.020	-.062	-.034	-.067	-.039	-.066	-.022	-.052	-.021	-.047	-.051	-.032	.800	
.800	.006	-.060	-.025	-.067	-.017	-.077	-.028	-.064	-.013	-.047	-.020	-.037	-.051	-.900	.950	
$\alpha = 0$																
.0125	.071	-.053	.059	.062	.062	-.018	.018	.007	.000	.029	-.025	.018	-.070	-.025	.0125	
.025	.071	-.021	.064	-.022	-.016	-.018	-.040	-.016	.005	.028	-.025	.032	-.055	-.050	.025	
.050	.044	-.014	.042	-.016	-.013	-.005	-.006	-.005	-.001	.023	-.030	.027	-.058	-.075	.075	
.075	.038	-.009	.013	-.013	-.013	-.006	-.010	-.008	-.005	.001	.015	-.024	.033	-.037	.100	
.100	.038	-.008	.016	-.011	-.011	-.004	-.017	-.011	-.005	.015	-.018	.007	.025	-.024	.150	
.150	.037	-.007	.017	-.011	-.011	-.003	-.004	-.017	-.005	.015	-.018	.007	.020	-.020	.200	
.200	.003	-.007	.012	-.005	-.016	-.004	-.022	-.005	-.017	-.018	-.029	.020	-.055	.013	.250	
.250	.002	-.020	.013	-.002	-.002	-.004	-.022	-.017	-.017	-.020	-.012	-.030	-.055	.010	.300	
.300	.002	-.008	.022	-.009	-.024	-.004	-.004	-.013	-.030	-.016	-.017	-.056	-.060	-.300	.350	
.350	.001	-.013	.010	-.005	-.046	.018	-.067	-.008	-.039	-.008	-.010	-.055	-.005	-.040	.400	
.400	-.013	-.010	-.035	-.008	-.046	.018	-.067	-.008	-.039	-.008	-.010	-.055	-.005	-.045	.450	
.500	-.023	-.010	-.010	-.004	-.054	-.007	-.057	-.004	-.056	-.005	-.066	-.003	-.074	-.500	.550	
.600	-.039	-.001	-.050	-.002	-.065	-.012	-.063	-.001	-.069	-.003	-.072	-.009	-.090	-.022	.600	
.650	-.068	-.011	-.011	-.077	-.005	-.005	-.072	-.000	-.075	-.002	-.079	-.007	-.091	-.018	.700	
.700	-.081	-.068	-.011	-.077	-.005	-.005	-.072	-.000	-.075	-.002	-.079	-.007	-.091	-.018	.750	
.750	-.070	-.008	-.079	-.016	-.084	-.017	-.083	-.002	-.076	-.003	-.077	-.005	-.080	-.800	.850	
.800	-.082	-.007	-.083	-.001	-.092	-.005	-.080	-.007	-.073	-.003	-.079	-.010	-.090	-.900	.950	
$\alpha = 2$																
.0125	.018	-.013	.002	.016	.017	.059	.062	.081	.097	.090	.126	.134	.103	.184	.0125	
.025	.027	-.017	.016	.004	.026	-.027	.043	-.054	.068	-.078	.052	-.137	.093	-.121	.050	
.050	.012	-.011	-.004	.023	-.033	.038	-.053	.063	-.082	.078	.145	.093	-.186	.075		
.075	.005	-.011	.009	-.008	.022	-.039	.029	-.052	.045	-.066	.150	.085	-.098	.100		
.100	.005	-.005	.009	-.008	.021	-.045	.028	-.054	.040	-.066	.116	.075	-.181	.091		
.150	.011	-.010	-.026	.021	-.045	.028	-.054	.039	-.061	.056	.068	.182	.063	-.200	.150	
.200	-.028	-.014	-.036	.021	-.041	.038	-.054	.039	-.061	.056	.072	.166	.061	-.200	.200	
.250	-.018	-.003	-.034	-.027	-.047	.022	-.056	.022	-.060	.060	.071	.145	.057	-.300	.300	
.300	-.017	-.010	-.042	.022	-.045	.027	-.056	.044	-.060	.060	.071	.063	-.145	-.300	.350	
.350	-.041	-.035	-.052	.038	-.067	.049	-.083	.036	-.066	.046	-.066	.105	.051	-.400	.450	
.400	-.045	-.034	-.052	.022	-.078	-.079	.023	-.082	.044	-.088	.030	-.114	-.040	-.500	.550	
.500	-.045	-.034	-.052	.027	-.084	.018	-.087	.023	-.095	.032	-.102	.022	-.124	.018	.600	
.600	-.071	-.028	-.070	.027	-.084	.018	-.087	.023	-.095	.032	-.102	.022	-.124	.018	.650	
.650	-.095	-.085	-.085	.017	-.093	.024	-.096	.031	-.098	.036	-.109	.025	-.119	.029	.700	
.700	-.085	-.011	-.089	.005	-.100	.008	-.103	.026	-.098	.036	-.107	.026	-.114	.029	.750	
.750	-.097	-.012	-.093	.023	-.105	.016	-.100	.034	-.094	.033	-.107	.018	-.114	.029	.800	
.800	-.097	-.012	-.093	.023	-.105	.016	-.100	.034	-.094	.033	-.107	.018	-.114	.029	.850	
.850	-.097	-.012	-.093	.023	-.105	.016	-.100	.034	-.094	.033	-.107	.018	-.114	.029	.900	
.900	-.097	-.012	-.093	.023	-.105	.016	-.100	.034	-.094	.033	-.107	.018	-.114	.029	.950	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING
(b) M = 2.01 - Continued

x/c, nominal	Cp at $y/\frac{c}{2}$ of :														x/c_1 , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 4$																
.0125	-.017	.042	-.068	.070	-.123	.110	-.158	.144	-.181	.146	-.200	.146	-.223	.146	.0125	
.025	-.009	.064	-.048	.045	-.114	.087	-.165	.116	-.173	.132	-.200	.136	-.223	.162	.050	
.050	-.015	.048	-.045	.065	-.059	.086	-.081	.108	-.186	.117	-.203	.139	-.223	.175	.075	
.075	-.016	.047	-.042	.046	-.077	.087	-.129	.107	-.182	.109	-.202	.129	-.223	.141	.100	
.100	-.015	.042	-.040	.058	-.077	.087	-.129	.107	-.173	.103	-.202	.117	-.225	.123	.150	
.125	-.006	.045	-.050	.056	-.077	.083	-.097	.080	-.173	.103	-.191	.112	-.228	.101	.200	
.150	-.045	.038	-.053	.056	-.072	.074	-.094	.080	-.155	.095	-.191	.101	-.228	.096	.250	
.175	-.033	.024	-.057	.055	-.064	.083	-.134	.101	-.182	.101	-.192	.101	-.228	.072	.300	
.200	-.037	.041	-.062	.042	-.074	.069	-.082	.082	-.110	.107	-.182	.101	-.223	.072	.350	
.225	-.055	.060	-.072	.067	-.091	.082	-.111	.067	-.088	.077	-.165	.064	-.209	.085	.400	
.250	-.063	.067	-.076	.049	-.099	-.099	-.105	.054	-.123	.074	-.165	.056	-.207	.051	.500	
.275	-.083	.060	-.091	.060	-.108	.053	-.112	.054	-.117	.066	-.156	.056	-.207	.051	.550	
.300	-.083	.060	-.105	.041	-.116	.049	-.119	.064	-.124	.071	-.146	.062	-.202	.062	.700	
.325	-.101	-.105	-.116	-.041	-.116	.049	-.119	.064	-.124	.071	-.146	.062	-.202	.062	.750	
.350	-.096	.037	-.107	.027	-.121	.029	-.126	.052	-.122	.062	-.126	.063	-.207	.080	.800	
.375	-.115	.064	-.122	.054	-.137	.057	-.138	.081	-.143	.094	-.212	.097	-.239	.093	.850	
.400	-.122	.063	-.125	.071	-.139	.069	-.135	.088	-.115	.088	-.215	.084	-.239	.090	.900	
.425	-.125	.093	-.138	.113	-.183	.147	-.204	.179	-.212	.188	-.222	.180	-.236	.196	.025	
.450	.051	.099	-.135	.102	-.184	.125	-.207	.160	-.205	.175	-.220	.178	-.237	.196	.050	
.475	-.045	.081	-.136	.102	-.185	.118	-.207	.161	-.212	.161	-.222	.177	-.237	.175	.075	
.500	-.043	.076	-.136	.094	-.185	.104	-.204	.128	-.209	.151	-.221	.171	-.238	.180	.100	
.525	-.043	.068	-.069	.093	-.180	.104	-.204	.128	-.213	.146	-.224	.161	-.238	.161	.150	
.550	-.029	.074	-.070	.080	-.154	.104	-.204	.120	-.213	.150	-.219	.156	-.242	.138	.200	
.575	-.057	.065	-.078	.082	-.121	.115	-.189	.138	-.210	.135	-.210	.150	-.223	.132	.250	
.600	-.050	.047	-.079	.084	-.152	.092	-.173	.121	-.202	.148	-.222	.140	-.243	.131	.300	
.625	-.056	.065	-.083	.079	-.097	.098	-.121	.121	-.202	.148	-.222	.140	-.243	.131	.300	
.650	-.078	.098	-.089	.098	-.108	.119	-.143	.106	-.192	.116	-.211	.121	-.243	.125	.400	
.675	-.077	.113	-.083	.083	-.117	-.117	-.117	.093	-.194	.115	-.224	.105	-.242	.105	.500	
.700	-.101	.088	-.110	.086	-.126	.089	-.122	.092	-.190	.105	-.222	.098	-.242	.093	.600	
.725	-.116	-.119	-.059	.133	-.076	-.131	-.131	.091	-.173	.105	-.221	.102	-.239	.104	.700	
.750	-.115	.064	-.122	.054	-.137	.057	-.138	.081	-.143	.094	-.212	.097	-.239	.090	.800	
.775	-.122	.063	-.125	.071	-.139	.069	-.135	.088	-.115	.088	-.215	.084	-.239	.080	.850	
.800	-.125	.093	-.138	.129	-.181	.147	-.204	.179	-.212	.188	-.222	.180	-.236	.196	.025	
.825	-.097	.134	-.194	.156	-.218	.188	-.233	.204	-.230	.214	-.238	.201	-.251	.216	.050	
.850	-.086	.135	-.194	.156	-.214	.162	-.233	.189	-.232	.199	-.236	.207	-.252	.207	.075	
.875	-.085	.118	-.214	.145	-.224	.156	-.235	.161	-.230	.179	-.236	.207	-.247	.204	.100	
.900	-.042	.096	-.116	.133	-.224	.141	-.235	.161	-.230	.179	-.236	.207	-.247	.183	.150	
.925	-.069	.095	-.091	.118	-.218	.142	-.242	.156	-.233	.179	-.239	.193	-.255	.157	.200	
.950	-.069	.094	-.096	.118	-.202	.147	-.238	.177	-.234	.172	-.237	.190	-.258	.157	.200	
.975	-.069	.074	-.092	.121	-.212	.123	-.232	.177	-.239	.190	-.241	.171	-.258	.164	.250	
.0125	-.096	.096	-.092	.120	-.183	.135	-.159	.239	-.184	.184	-.243	.176	-.259	.168	.300	
.025	-.095	.134	-.100	.138	-.129	.156	-.218	.141	-.235	.151	-.235	.159	-.260	.163	.400	
.050	-.089	.151	-.124	.111	-.111	-.111	-.218	.130	-.240	.154	-.252	.141	-.260	.161	.500	
.075	-.109	.116	-.118	.121	-.126	.120	-.205	.127	-.240	.143	-.250	.134	-.263	.130	.600	
.100	-.124	-.130	-.090	.135	-.104	-.148	-.148	.119	-.234	.134	-.252	.135	-.260	.141	.700	
.125	-.124	.085	-.130	.087	-.140	.086	-.136	.106	-.230	.120	-.246	.124	-.250	.124	.800	
.150	-.131	.085	-.135	.107	-.142	.099	-.135	.118	-.215	.115	-.215	.112	-.225	.090	.900	
.175	-.127	.172	-.225	.189	-.239	.209	-.208	-.251	.244	-.243	.240	-.249	.217	-.269	.231	.050
.200	-.132	.167	-.229	.184	-.234	.196	-.251	.225	-.240	.225	-.247	.218	-.268	.231	.075	
.225	-.152	.152	-.245	.184	-.248	.195	-.242	.216	-.244	.228	-.249	.238	-.268	.231	.100	
.250	-.086	.143	-.168	.156	-.248	.179	-.252	.200	-.241	.217	-.247	.247	-.268	.210	.150	
.275	-.075	.127	-.207	.171	-.248	.179	-.252	.200	-.241	.217	-.247	.247	-.268	.192	.200	
.300	-.068	.137	-.188	.155	-.249	.179	-.256	.200	-.246	.219	-.252	.228	-.269	.210	.250	
.325	-.097	.123	-.166	.155	-.240	.179	-.259	.215	-.249	.212	-.250	.224	-.272	.192	.300	
.350	-.091	.108	-.099	.155	-.242	.163	-.255	.202	-.254	.227	-.255	.227	-.272	.195	.350	
.375	-.090	.122	-.096	.151	-.242	.175	-.255	.222	-.255	.222	-.256	.214	-.274	.199	.300	
.400	-.113	.164	-.108	.178	-.232	.196	-.255	.184	-.259	.189	-.266	.196	-.275	.194	.400	
.425	-.102	.193	-.163	.209	-.262	.172	-.265	.189	-.266	.172	-.277	.172	-.277	.160	.450	
.450	-.102	.137	-.127	.151	-.141	.151	-.258	.165	-.265	.176	-.270	.165	-.279	.157	.500	
.475	-.130	.137	-.141	.118	-.144	.134	-.270	.152	-.263	.166	-.274	.163	-.280	.169	.550	
.500	-.141	-.141	-.141	.118	-.144	.134	-.270	.152	-.263	.165	-.274	.163	-.280	.169	.600	
.525	-.139	.112	-.140	.118	-.151	.119	-.224	.142	-.262	.152	-.271	.155	-.275	.175	.650	
.550	-.151	.119	-.146	.146	-.151	.139	-.253	.150	-.258	.148	-.275	.135	-.280	.169	.700	
.575	-.131	.119	-.146	.146	-.151	.139	-.253	.150	-.258	.148	-.275	.135	-.280	.169	.750	
.600	-.131	.119	-.146	.146	-.151	.139	-.253	.150	-.258	.148	-.275	.135	-.280	.169	.800	
.625	-.151	.119	-.146	.146	-.151	.139	-.253	.150	-.258	.148	-.275	.135	-.280	.169	.850	
.650	-.151	.119	-.146	.146	-.151	.139	-.253	.150	-.258	.148	-.275	.135	-.280	.169	.900	
.675	-.151	.119	-.146	.146	-.151	.139	-.253	.150	-.258	.148	-.275	.135	-.280	.169	.950	

TABLE V.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

(b) M = 2.01 - Continued

x/c, nominal	Cp at $y/\frac{c}{2}$ of :														x/c, nominal	
	.10		.25		.40		.50		.60		.70		.80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 12^\circ$																
.0125	-1.162	.201	-1.254	.215	-1.259	.228	-1.263	.237	-1.263	.256	-1.254	.263	-1.259	.233	-1.269	.125
.025	-1.182	.201	-1.252	.215	-1.251	.232	-1.263	.253	-1.251	.252	-1.259	.243	-1.252	.252	.025	
.050	-1.195	.190	-1.262	.215	-1.261	.232	-1.263	.250	-1.254	.259	-1.260	.266	-1.270	.050	.075	
.075	-1.165	.181	-1.262	.200	-1.262	.229	-1.263	.239	-1.251	.253	-1.259	.278	-1.263	.100	.100	
.100	-1.103	.162	-1.254	.200	-1.267	.215	-1.263	.239	-1.251	.253	-1.259	.278	-1.270	.225	.150	
.150	-1.085	.174	-1.247	.182	-1.269	.215	-1.268	.241	-1.258	.253	-1.264	.261	-1.270	.247	.150	
.200	-1.108	.154	-1.247	.185	-1.269	.222	-1.271	.254	-1.259	.247	-1.263	.258	-1.272	.225	.200	
.250	-1.105	.128	-1.209	.188	-1.271	.211	-1.271	.264	-1.262	.265	-1.268	.247	-1.272	.234	.250	
.300	-1.105	.152	-1.140	.183	-1.276	.218	-1.274	.242	-1.268	.258	-1.268	.247	-1.272	.234	.300	
.350	-1.118	.207	-1.123	.207	-1.280	.214	-1.278	.221	-1.271	.223	-1.271	.214	-1.274	.227	.400	
.400	-1.118	.228		.193	-1.268		-1.285	.209	-1.270	.222	-1.274	.208	-1.276		.450	
.450	-1.111	.228		.193	-1.268		-1.285	.209	-1.270	.222	-1.274	.208	-1.276		.500	
.500	-1.148	.168	-1.146	.181	-1.230	.187	-1.284	.196	-1.283	.208	-1.274	.200	-1.274	.192	.600	
.600	-1.153	.153	-1.160	.147	-1.202	.170	-1.278	.185	-1.281	.198	-1.282	.200	-1.278	.201	.700	
.700	-1.153	.153	-1.160	.147	-1.202	.170	-1.278	.185	-1.281	.198	-1.282	.200	-1.278	.201	.750	
.800	-1.152	.142	-1.159	.151	-1.188	.156	-1.268	.175	-1.284	.185	-1.283	.191	-1.285		.800	
.850	-1.163	.150	-1.164	.173	-1.177	.172	-1.257	.193	-1.281	.179	-1.283	.170	-1.285		.850	
.900	-1.179	.194	-1.178	.209	-1.231	.206	-1.289	.229	-1.289	.227	-1.286	.210	-1.287		.900	
.950	-1.179	.194	-1.178	.209	-1.231	.206	-1.289	.229	-1.289	.227	-1.286	.210	-1.287		.950	
$\alpha = 14^\circ$																
.0125	-1.204	.238	-1.258	.248	-1.271	.251	-1.266	.273	-1.284	.264	-1.289	.266	-1.252	-1.276	.125	
.025	-1.218	.235	-1.259	.248	-1.261	.261	-1.261	.266	-1.273	.279	-1.258	.283	-1.266	.267	.050	
.050	-1.228	.226	-1.266	.261	-1.261	.272	-1.266	.283	-1.259	.297	-1.268	.298	-1.271	.275	.075	
.075	-1.220	.217	-1.240	.272	-1.267	.274	-1.275	.274	-1.257	.294	-1.265	.311	-1.265	.287	.100	
.100	-1.183	.200	-1.267	.244	-1.275	.257	-1.273	.274	-1.259	.291	-1.266	.296	-1.276	.272	.150	
.150	-1.095	.211	-1.274	.221	-1.287	.250	-1.278	.274	-1.266	.291	-1.269	.295	-1.276	.272	.200	
.200	-1.114	.191	-1.278	.225	-1.282	.261	-1.282	.293	-1.266	.286	-1.269	.294	-1.278	.254	.250	
.250	-1.113	.166	-1.270	.226	-1.281	.241	-1.281	.272	-1.273	.302	-1.273	.284	-1.279	.271	.300	
.300	-1.118	.178	-1.249	.219	-1.294	.255	-1.275	.272	-1.273	.302	-1.273	.284	-1.279	.271	.350	
.350	-1.128	.244	-1.188	.250	-1.299	.275	-1.292	.254	-1.278	.265	-1.269	.281	-1.281	.265	.400	
.400	-1.128	.244	-1.188	.250	-1.299	.275	-1.292	.254	-1.278	.265	-1.269	.281	-1.284	.265	.450	
.450	-1.128	.270		.228	-1.299		-1.300	.259	-1.285	.264	-1.280	.248	-1.284		.500	
.500	-1.158	.206	-1.165	.218	-1.278	.223	-1.304	.227	-1.289	.247	-1.283	.237	-1.285	.230	.600	
.600	-1.166	.173	-1.173	.187	-1.267	.206	-1.300	.222	-1.288	.240	-1.286	.237	-1.286	.246	.700	
.700	-1.166	.173	-1.172	.189	-1.268	.205	-1.302	.222	-1.288	.240	-1.286	.237	-1.286	.246	.750	
.800	-1.169	.183	-1.172	.189	-1.268	.203	-1.302	.221	-1.286	.239	-1.286	.227	-1.290		.800	
.850	-1.179	.194	-1.178	.209	-1.231	.206	-1.289	.229	-1.289	.227	-1.286	.210	-1.287		.850	
.900	-1.179	.194	-1.178	.209	-1.231	.206	-1.289	.229	-1.289	.227	-1.286	.210	-1.287		.900	
.950	-1.179	.194	-1.178	.209	-1.231	.206	-1.289	.229	-1.289	.227	-1.286	.210	-1.287		.950	
$\alpha = 16^\circ$																
.0125	-1.231	.276	-1.266	.273	-1.277	.278	-1.277	.303	-1.273	.313	-1.263	.322	-1.270	.281	-1.277	.125
.025	-1.240	.276	-1.264	.273	-1.261	.276	-1.261	.312	-1.273	.324	-1.262	.322	-1.269	.306	-1.297	.050
.050	-1.253	.273	-1.269	.273	-1.261	.276	-1.261	.312	-1.273	.331	-1.264	.334	-1.271	.337	-1.275	.075
.075	-1.242	.272	-1.266	.273	-1.261	.276	-1.261	.312	-1.273	.332	-1.264	.334	-1.272	.332	-1.275	.100
.100	-1.143	.250	-1.215	.289	-1.281	.314	-1.278	.314	-1.273	.329	-1.265	.324	-1.271	.341	-1.277	.150
.150	-1.156	.262	-1.283	.264	-1.286	.301	-1.278	.317	-1.273	.328	-1.267	.332	-1.275	.341	-1.282	.200
.200	-1.129	.236	-1.291	.269	-1.287	.317	-1.271	.321	-1.271	.346	-1.269	.322	-1.273	.337	-1.277	.250
.250	-1.128	.209	-1.294	.269	-1.294	.320	-1.277	.347	-1.271	.349	-1.274	.347	-1.272	.322	-1.279	.250
.300	-1.131	.248	-1.286	.270	-1.296	.309	-1.274	.349	-1.272	.347	-1.272	.333	-1.280	.309	-1.300	.350
.350	-1.133	.293	-1.249	.205	-1.298	.329	-1.295	.307	-1.279	.308	-1.279	.320	-1.281	.307	-1.400	.400
.400	-1.133	.293	-1.249	.205	-1.298	.329	-1.295	.307	-1.279	.308	-1.279	.320	-1.281	.307	-1.450	.450
.450	-1.141	.317		.273	-1.296		-1.304	.291	-1.286	.308	-1.279	.296	-1.284		.500	
.500	-1.141	.317		.273	-1.296		-1.304	.291	-1.286	.308	-1.279	.296	-1.284		.550	
.600	-1.172	.250	-1.200	.267	-1.297	.274	-1.303	.278	-1.291	.292	-1.283	.284	-1.286	.273	.600	
.700	-1.178	.198	-1.220	.229	-1.291	.255	-1.303	.265	-1.290	.284	-1.286	.287	-1.287	.286	.700	
.800	-1.183	.232	-1.196	.230	-1.280	.240	-1.297	.257	-1.296	.262	-1.288	.274	-1.291		.800	
.850	-1.204	.240	-1.198	.256	-1.261	.257	-1.297	.274	-1.289	.264	-1.288	.256	-1.291		.850	
.900	-1.204	.240	-1.198	.256	-1.261	.257	-1.297	.274	-1.289	.264	-1.288	.256	-1.291		.900	
.950	-1.204	.240	-1.198	.256	-1.261	.257	-1.297	.274	-1.289	.264	-1.288	.256	-1.291		.950	
$\alpha = 18^\circ$																
.0125	-1.245	.307	-1.271	.300	-1.277	.314	-1.280	.332	-1.269	.351	-1.273	.295	-1.286	.318	.025	
.025	-1.253	.318	-1.271	.307	-1.267	.343	-1.281	.352	-1.265	.356	-1.272	.330	-1.285	.318	.050	
.050	-1.266	.316	-1.272	.307	-1.279	.359	-1.287	.367	-1.270	.370	-1.275	.363	-1.285	.318	.075	
.075	-1.274	.304	-1.279	.334	-1.284	.349	-1.284	.359	-1.268	.370	-1.272	.380	-1.285	.315	.100	
.100	-1.269	.289	-1.277	.334	-1.283	.341	-1.284	.370	-1.271	.371	-1.275	.370	-1.285	.315	.150	
.150	-1.218	.310	-1.287	.307	-1.283	.341	-1.284	.370	-1.271	.371	-1.275	.371	-1.285	.317	.200	
.200	-1.162	.282	-1.299	.319	-1.289	.359	-1.289	.387	-1.276	.374	-1.275	.371	-1.286	.327	.250	
.250	-1.148	.254	-1.306	.304	-1.298	.318	-1.289	.387	-1.270	.396	-1.282	.388	-1.288	.330	.300	
.300	-1.149	.304	-1.300	.314	-1.300	.347	-1.307	.367	-1.290	.391	-1.285	.366	-1.289	.345	.350	
.350	-1.154	.337	-1.278	.343	-1.306	.370	-1.301									

TABLE V.- Concluded
PRESSURE COEFFICIENTS FOR CAMBERED DELTA WING

(b) $M = 2.01$ - Concluded

x/c , nominal	Cp at $y/b = \frac{1}{2}$ of :														x/c , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 20^\circ$																
+.0125	-.261	.347	-.277	.377	-.285	.337	-.296	.355	-.277	.378	-.275	.326	-.282	.342	.0125	
.025	-.264	.368	-.277	.377	.401	-.272	.388	-.296	.387	-.275	.390	-.273	.368	.342	.025	
.050	-.274	.366	-.283	.383	.379	-.286	.405	-.296	.403	-.275	.414	-.274	.403	.342	.050	
.075	-.288	.358	-.287	.380	.387	-.289	.383	-.296	.404	-.273	.414	-.271	.418	.386	.075	
.100	-.289	.342	-.287	.380	.387	-.289	.383	-.297	.414	-.278	.415	-.275	.413	.382	.100	
.150	-.267	.365	-.300	.358	-.296	.389	-.300	.406	-.300	.428	-.278	.422	-.275	.416	.364	.200
.200	-.213	.333	-.311	.369	-.296	.406	-.300	.430	-.300	.430	-.280	.443	-.278	.437	.367	.250
.250	-.192	.306	-.311	.368	-.296	.381	-.300	.430	-.300	.437	-.279	.418	-.285	.385	.360	.350
.300	-.174	.333	-.312	.368	-.311	.394	-.312	.418	-.312	.437	-.280	.437	-.279	.418	.385	.360
.350	-.191	.390	-.300	.395	-.315	.418	-.317	.436	-.317	.436	-.286	.403	-.288	.385	.350	.400
.400	-.181	.419	-.300	.374	-.315	.418	-.318	.432	-.318	.432	-.297	.399	-.281	.382	.350	.450
.450	-.204	.344	-.271	.355	-.315	.355	-.320	.353	-.315	.353	-.298	.380	-.287	.367	.354	.500
.500	-.229	.324	-.315	.338	-.321	.357	-.321	.374	-.321	.374	-.295	.374	-.292	.372	.354	.600
.650	-.208	.329	-.324	.335	-.315	.338	-.321	.357	-.321	.357	-.295	.374	-.292	.372	.354	.650
.700	-.229	.335	-.324	.338	-.315	.338	-.321	.357	-.321	.357	-.295	.374	-.292	.372	.354	.700
.750	-.205	.335	-.249	.328	-.307	.323	-.321	.352	-.321	.352	-.296	.362	-.294	.362	.354	.750
.800	-.216	.335	-.248	.352	-.301	.342	-.321	.368	-.321	.368	-.290	.363	-.292	.362	.354	.800
.850	-.216	.335	-.248	.352	-.301	.342	-.321	.368	-.321	.368	-.290	.363	-.292	.362	.354	.850
.900	-.216	.335	-.248	.352	-.301	.342	-.321	.368	-.321	.368	-.290	.363	-.292	.362	.354	.900
.950	-.216	.335	-.248	.352	-.301	.342	-.321	.368	-.321	.368	-.290	.363	-.292	.362	.354	.950

TABLE VI
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING

(a) M = 1.61

x/c, nominal	Cp at $y/\frac{c}{2}$ of :														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -20$																
.0125	.431	-.380	.341	-.399	.281	-.400	.227	-.391	.179	-.400	.143	-.388	.222	-.388	.0125	
.025	.372	-.401	.329	-.399	.288	-.397	.305	-.390	.279	-.387	.198	-.390	.211	-.391	.050	
.050	.337	-.436	.378	-.392	.363	-.400	.313	-.375	.304	-.397	.248	-.394	.211	-.387	.075	
.075	.386	-.446	.314	-.402	.274	-.400	.313	-.375	.304	-.397	.248	-.394	.211	-.387	.100	
.100	.416	-.433	.383	-.409	.314	-.391	.313	-.389	.316	-.399	.238	-.394	.211	-.387	.125	
.125	.399	-.359	.367	-.436	.349	-.400	.324	-.388	.307	-.390	.230	-.390	.250	-.394	.150	
.150	.321	-.259	.335	-.441	.339	-.396	.316	-.396	.280	-.387	.279	-.390	.247	-.484	.200	
.175	.343	-.213	.367	-.443	.230	-.437	.316	-.449	.292	-.395	.267	-.390	.246	-.480	.250	
.200	.346	-.220	.344	-.463	.318	-.431	.303	-.449	.281	-.402	.258	-.395	.233	-.492	.300	
.225																
.250																
.300																
.400	.359	-.220	.320	-.402	.319	-.428	.305	-.430	.274		.261	-.406	.228	-.448	.400	
.450																
.500	.331	-.210	.321	-.330	.345	-.458	.245	-.430	.242	-.444	.251	-.405	.241	-.466	.500	
.550																
.600	.286	-.200	.288	-.252	.269	-.452	.259	-.435	.247	-.427	.246	-.449	.281	-.470	.600	
.650																
.700	.265	-.166	.275	-.234	.264	-.422	.263	-.445	.251	-.431	.274	-.453	.289	-.510	.700	
.750																
.800	.297	-.191	.289	-.216	.278	-.379	.300	-.461	.290	-.426	.300	-.423	.294	-.510	.800	
.850																
.900	.365	-.186	.347	-.191	.341	-.316	.279	-.437	.304	-.426	.312	-.426	.531	-.900	.950	
.950																
$\alpha = -18$																
.0125	.307	-.376	.314	-.390	.270	-.399	.214	-.389	.150	-.401	.125	-.385	.209	-.385	.0125	
.025	.330	-.399	.293	-.386	.265	-.394	.277	-.388	.245	-.390	.178	-.387	.209	-.383	.050	
.050	.295	-.447	.341	-.394	.325	-.397	.283	-.388	.255	-.396	.216	-.393	.193	-.385	.075	
.075	.347	-.466	.274	-.418	.232	-.397	.283	-.388	.266	-.396	.207	-.389	.221	-.383	.100	
.100	.372	-.426	.342	-.460	.280	-.389	.283	-.388	.221	-.387	.249	-.383	.215	-.469	.200	
.125	.354	-.287	.325	-.439	.312	-.384	.283	-.391	.249	-.391	.230	-.388	.209	-.467	.250	
.150	.280	-.197	.291	-.438	.299	-.388	.265	-.389	.221	-.387	.249	-.383	.215	-.469	.300	
.200	.318	-.178	.325	-.445	.195	-.435	.262	-.439	.236	-.387	.228	-.388	.209	-.467	.350	
.250																
.300	.303	-.185	.302	-.459	.265	-.430	.242	-.448	.227	-.390	.209	-.395	.188	-.486	.300	
.350																
.400	.315	-.212	.273	-.375	.251	-.430	.246	-.430	.224		.217	-.393	.175	-.441	.400	
.450																
.500	.276	-.200	.258	-.291	.282	-.466	.191	-.432	.193	-.440	.201	-.392	.172	-.457	.500	
.550																
.600	.210	-.196	.225	-.224	.217	-.456	.212	-.436	.191	-.423	.185	-.440	.173	-.458	.600	
.650																
.700	.209	-.163	.216	-.209	.209	-.414	.196	-.447	.191	-.428	.169	-.451	.181	-.491	.700	
.750																
.800	.233	-.187	.224	-.194	.204	-.358	.205	-.470	.175	-.422	.184	-.419	.209	-.508	.800	
.850																
.900	.261	-.180	.245	-.180	.231	-.281	.178	-.448	.209	-.427	.419	-.419				
.950																
$\alpha = -16$																
.0125	.354	-.376	.286	-.371	.268	-.367	.211	-.362	.150	-.387	.120	-.370	.195	-.370	.0125	
.025	.279	-.397	.258	-.364	.248	-.364	.252	-.362	.228	-.370	.162	-.371	.209	-.372	.050	
.050	.246	-.447	.300	-.387	.294	-.368	.253	-.362	.230	-.376	.197	-.376	.166	-.376	.075	
.075	.300	-.453	.228	-.395	.200	-.372	.253	-.378	.244	-.378	.179	-.376	.178	-.368	.100	
.100	.321	-.330	.297	-.433	.245	-.367	.253	-.363	.244	-.378	.179	-.376	.175	-.455	.200	
.125	.304	-.169	.280	-.437	.275	-.393	.253	-.368	.222	-.372	.176	-.376	.171	-.455	.250	
.150	.231	-.147	.246	-.438	.258	-.385	.229	-.364	.196	-.369	.201	-.373	.175	-.455	.300	
.200	.231	-.147	.246	-.438	.258	-.385	.229	-.364	.196	-.369	.176	-.373	.175	-.455	.350	
.250	.270	-.151	.283	-.456	.157	-.417	.224	-.414	.200	-.367	.176	-.379	.171	-.455	.400	
.300	.259	-.157	.260	-.417	.225	-.409	.205	-.425	.183	-.369	.162	-.388	.164	-.476	.350	
.350																
.400	.270	-.180	.229	-.287	.210	-.408	.203	-.408	.170		.160	-.380	.134	-.437	.400	
.450																
.500	.229	-.173	.216	-.208	.229	-.459	.135	-.409	.134	-.422	.151	-.364	.120	-.458	.500	
.550																
.600	.160	-.175	.171	-.183	.159	-.423	.155	-.417	.148	-.398	.132	-.414	.126	-.455	.600	
.650																
.700	.146	-.158	.152	-.181	.150	-.362	.143	-.437	.132	-.403	.122	-.427	.121	-.476	.700	
.750																
.800	.164	-.177	.165	-.172	.144	-.298	.162	-.457	.129	-.399	.114	-.397	.113	-.479	.800	
.850																
.900	.177	-.168	.164	-.160	.150	-.216	.096	-.424	.120	-.401	.118	-.403				
.950																
$\alpha = -14$																
.0125	.315	-.374	.261	-.355	.258	-.351	.208	-.324	.142	-.356	.108	-.339	.190	-.348	.0125	
.025	.240	-.393	.227	-.350	.234	-.346	.227	-.334	.162	-.355	.148	-.344	.209	-.348	.050	
.050	.206	-.420	.257	-.374	.264	-.346	.221	-.344	.216	-.346	.176	-.350	.154	-.348	.075	
.075	.255	-.444	.209	-.371	.193	-.350	.221	-.336	.218	-.343	.155	-.346	.154	-.348	.100	
.100	.227	-.104	.254	-.394	.210	-.359	.221	-.336	.218	-.343	.155	-.346	.154	-.352	.200	
.125	.258	-.122	.241	-.448	.237	-.384	.218	-.354	.190	-.338	.155	-.346	.154	-.352	.250	
.150	.186	-.132	.210	-.462	.220	-.347	.195	-.376	.161	-.341	.172	-.346	.143	-.432	.300	
.200	.224	-.138	.237	-.429	.122	-.399	.186	-.408	.167	-.356	.147	-.356	.131	-.432	.250	
.250	.211	-.134	.221	-.322	.189	-.406	.168	-.395	.150	-.363	.135	-.366	.108	-.453	.300	
.300																
.350	.226	-.163	.189	-.174	.172	-.434	.164	-.393	.134		.123	-.362	.095	-.418	.400	
.400																
.450	.179	-.147	.177	-.132	.192	-.437	.098	-.389	.097	-.						

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING

(a) $M = 1.61$ - Continued

x/c , nominal	C_p at $y/c = 0$														x/c , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -12^\circ$																
.0125	.282	-.370	.298	-.343	.248	-.325	.207	-.318	.137	-.327	.104	-.309	.196	-.319	.0125	
.025	.207	-.387	.197	-.338	.209	-.330	.201	-.315	.192	-.307	.137	-.312	.146	-.319	.050	
.040	.168	-.357	.222	-.343	.236	-.321	.195	-.325	.190	-.315	.161	-.317	.146	-.317	.075	
.060	.212	-.047	.144	-.047	.157	-.325	.195	-.325	.190	-.315	.138	-.316	.317	-.100		
.100	.232	-.066	.218	-.393	.175	-.340	.191	-.314	.188	-.315	.138	-.316	.317	-.100		
.150	.248	-.114	.202	-.393	.196	-.344	.189	-.337	.162	-.308	.135	-.315	.317	-.150		
.200	.150	-.121	.174	-.436	.183	-.322	.162	-.343	.127	-.319	.150	-.312	.125	-.400		
.250	.185	-.123	.195	-.294	.091	-.307	.152	-.354	.137	-.339	.123	-.325	.113	-.398	.250	
.300	.174	-.114	.181	-.145	.155	-.422	.134	-.355	.120	-.338	.111	-.340	.089	-.420	.300	
.350	.189	-.145	.150	-.115	.133	-.432	.134	-.394	.109		.097	-.345	.068	-.395	.400	
.400	.145	-.126	.142	-.118	.154	-.337	.066	-.417	.068	-.341	.076	-.327	.056	-.402	.500	
.500	.145	-.126	.142	-.118	.154	-.337	.066	-.417	.075	-.363	.060	-.337	.052	-.409	.550	
.550	.081	-.131	.101	-.149	.086	-.194	.082	-.419	.075						.600	
.600	.074	-.126	.081	-.130	.069	-.136	.059	-.392	.053	-.394	.040	-.337	.042	-.426	.650	
.700	.074	-.126	.081	-.130	.069	-.136	.059	-.305	.034	-.411	.033	-.328	.038	-.416	.750	
.750	.076	-.136	.074	-.132	.051	-.123	.059	-.305	.034	-.411	.038	-.351			.800	
.800	.076	-.136	.074	-.132	.059	-.108	.012	-.211	.039	-.402					.850	
.850	.074	-.135	.069	-.132											.900	
.900	.074	-.135	.069	-.132											.950	
$\alpha = -10^\circ$																
.0125	.247	-.325	.217	-.323	.244	-.295	.206	-.286	.133	-.289	.096	-.275	.190	-.286	.0125	
.025	.181	-.292	.173	-.317	.190	-.296	.178	-.284	.173	-.268	.128	-.278	.130	-.286	.050	
.050	.142	-.122	.186	-.323	.211	-.293	.104	-.304	.170	-.290	.169	-.279	.142	-.286	.075	
.075	.175	-.083	.120	-.332	.104	-.324	.162	-.281	.163	-.275	.116	-.282	.106	-.286	.100	
.100	.195	-.044	.180	-.360	.142	-.306	.156	-.301	.131	-.277	.124	-.282	.106	-.366	.150	
.150	.182	-.100	.167	-.377	.165	-.324	.134	-.306	.107	-.291	.095	-.291	.086	-.366	.250	
.200	.119	-.099	.143	-.230	.150	-.330	.124	-.306	.107	-.291	.124	-.282	.106	-.393	.300	
.250	.156	-.094	.162	-.067	.070	-.377	.174	-.352	.108	-.298	.095	-.291	.086	-.366	.350	
.300	.138	-.093	.148	-.069	.128	-.382	.195	-.363	.096	-.303	.081	-.309	.060	-.393	.400	
.350	.156	-.119	.121	-.098	.111	-.287	.104	-.377	.086	-.305	.070	-.305	.046	-.365	.450	
.400	.114	-.099	.114	-.097	.128	-.105	.035	-.371	.042	-.352	.048	-.289	.032	-.377	.500	
.500	.055	-.105	.072	-.130	.064	-.087	.055	-.311	.045	-.359	.029	-.305	.027	-.373	.600	
.600	.047	-.105	.055	-.106	.050	-.097	.034	-.197	.029	-.369	.013	-.319	.016	-.401	.700	
.700	.047	-.105	.055	-.106	.050	-.097	.034	-.197	.029	-.369	.013	-.328	.013	-.384	.750	
.750	.052	-.109	.080	-.108	.029	-.105	.030	-.090	.012	-.355	.005	-.328			.800	
.800	.048	-.107	.043	-.105	.040	-.097	.013	-.078	.006	-.319	.013	-.350			.850	
.850	.048	-.107	.043	-.105											.900	
.900	.048	-.107	.043	-.105											.950	
$\alpha = -8^\circ$																
.0125	.202	-.211	.186	-.309	.221	-.286	.207	-.267	.117	-.264	.071	-.241	.174	-.245	.0125	
.025	.157	-.176	.157	-.307	.157	-.285	.150	-.262	.148	-.245	.108	-.245	.050			
.050	.123	-.140	.149	-.223	.121	-.279	.150	-.267	.140	-.255	.115	-.250	.110	-.245	.075	
.075	.164	-.068	.081	-.319	.067	-.284	.135	-.267	.127	-.259	.124	-.248	.088	-.243	.100	
.100	.152	-.043	.140	-.300	.101	-.284	.127	-.235	.124	-.226	.100	-.248	.078	-.245	.150	
.150	.141	-.087	.127	-.112	.121	-.304	.124	-.268	.124	-.248	.124	-.246	.073	-.325	.200	
.200	.079	-.087	.103	-.076	.076	-.294	.099	-.296	.070	-.256	.096	-.246	.073	-.325	.250	
.250	.112	-.082	.105	-.076	.075	-.314	.088	-.328	.075	-.266	.064	-.250	.062	-.326	.300	
.300	.099	-.080	.104	-.076	.068	-.212	.073	-.316	.063	-.287	.052	-.271	.042	-.346	.350	
.350	.113	-.100	.080	-.092	.070	-.060	.073	-.307	.046		.040	-.255	.017	-.326	.400	
.400	.079	-.086	.079	-.091	.085	-.045	.008	-.226	.009	-.311	.019	-.252	.002	-.323	.450	
.500	.079	-.086	.079	-.091	.085	-.045	.008	-.226	.009	-.311	.019	-.252	.001	-.326	.500	
.600	.016	-.095	.037	-.123	.026	-.092	.025	-.064	.016	-.295	.002	-.272	.001	-.326	.650	
.650	.014	-.092	.021	-.097	.012	-.100	.006	-.071	.005	-.284	.014	-.276	.011	-.344	.700	
.700	.014	-.092	.021	-.097	.012	-.100	.006	-.071	.005	-.284	.014	-.276	.011	-.323	.750	
.750	.019	-.092	.014	-.098	.005	-.103	.004	-.082	.019	-.222	.021	-.262	.014	-.331	.800	
.800	.014	-.092	.013	-.079	.030	-.086	.022	-.086	.022	-.086	.042	-.053	.042	-.244	.850	
.850	.014	-.091	.009	-.091	.000	-.089	.038	-.082	.026	-.115	.039	-.149			.900	
.900	.010	-.069	.018	-.073	.025	-.073	.064	-.076	.046	-.060					.950	
$\alpha = -6^\circ$																
.0125	.169	-.111	.159	-.298	.205	-.279	.196	-.243	.105	-.225	.054	-.200	.128	-.209	.0125	
.025	.127	-.105	.129	-.296	.124	-.276	.121	-.263	.118	-.209	.089	-.201	.084	-.209	.050	
.050	.097	-.086	.114	-.115	.134	-.263	.105	-.244	.111	-.217	.085	-.207	.084	-.209	.075	
.075	.100	-.022	.070	-.110	.035	-.256	.090	-.235	.081	-.211	.058	-.199				
.100	.117	-.003	.106	-.095	.069	-.236	.084	-.238	.063	-.206	.192	-.190	.040	-.190		
.150	.104	-.061	.094	-.105	.082	-.182	.088	-.194	.064	-.226	.039	-.213	.039	-.273	.200	
.200	.053	-.063	.071	-.084	.064	-.086	.064	-.056	.043	-.221	.062	-.194	.039	-.273	.250	
.250	.082	-.063	.075	-.071	.007	-.056	.053	-.223	.044	-.221	.059	-.194	.027	-.272	.300	
.300	.070	-.056	.066	-.066	.050	-.061	.043	-.123	.033	-.221	.027	-.213	.011	-.295	.350	
.350	.067	-.068	.048	-.068	.037	-.086	.043	-.070	.019		.011	-.190	.012	-.270	.400	
.400	.048	-.063	.042	-.073	.056	-.066	.027	-.076	.020	-.124	.012	-.163	.031	-.256	.450	
.500	.007	-.077	.008	-.108	.000	-.083	.002	-.086	.013	-.084	.025	-.166	.032	-.257	.500	
.600	-.004	-.077	-.007	-.081	-.011	-.092	-.019	-.086	-.029	-.081	.042	-.161	.039	-.273	.600	
.700	-.004	-.077	-.013	-.079	-.030	-.086	-.022	-.086	-.042	-.053	.042	-.112	.040	-.239	.700	
.750	-.004	-.073	-.013	-.073	-.025	-.073	-.064	-.076	-.046	-.060					.800	
.800	-.010	-.069	-.018	-.073	-.025	-.073	-.064	-.076	-.046	-.060					.850	
.850	-.010															

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING

(a) M = 1.61 - Continued

x/c, nominal	C _p at $y/c = \frac{1}{4}$ of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -4^\circ$																
.0125	.130	-.039	.127	-.156	.174	-.221	.127	-.214	.087	-.208	.049	-.157	.064	-.064	.0125	
.025	.090	-.062	.098	-.110	.100	-.171	.073	-.112	.085	-.142	.056	-.141	.025	-.098	.025	
.050	.063	-.048	.078	-.099	.088	-.074	.063	-.097	.078	-.078	.048	-.112	.025	-.075	.050	
.075	.069	-.034	.043	-.063	.038	-.097	.063	-.078	.078	-.078	.048	-.112	.025	-.075	.075	
.100	.085	.028	.066	-.053	.049	-.075	.046	-.058	.058	-.039	.059	-.031	.064	-.072	.100	
.150	.071	-.036	.065	-.066	.042	-.050	.043	-.077	.050	-.046	.066	-.004	.072	-.150	.150	
.200	.032	-.043	.043	-.056	.030	-.044	.018	-.061	.007	-.043	.030	-.071	.000	-.147	.200	
.250	.051	-.043	.042	-.045	.035	-.048	.026	-.092	.013	-.055	.007	-.073	-.007	-.147	.250	
.300	.048	-.033	.036	-.038	.028	-.048	.012	-.080	.005	-.038	-.003	-.105	-.023	-.153	.300	
.350	.055	-.035	.021	-.043	.009	-.061	.011	-.057	-.008	-.020	-.090	-.037	-.177	.400	.350	
.400	.020	-.038	.015	-.049	.028	-.043	-.049	-.061	-.045	-.034	-.038	-.071	-.070	-.165	.450	
.500	.050	-.033	-.016	-.085	-.026	-.059	-.029	-.067	-.036	-.057	-.052	-.064	-.004	-.151	.500	
.600	.033	-.048	-.016	-.085	-.026	-.059	-.029	-.067	-.036	-.057	-.052	-.064	-.004	-.151	.600	
.650	.029	-.054	-.028	-.057	-.035	-.068	-.040	-.066	-.051	-.070	-.067	-.078	-.073	-.196	.700	
.700	.030	-.047	-.035	-.052	-.053	-.064	-.043	-.061	-.061	-.043	-.066	-.017	-.067	-.170	.750	
.750	.033	-.039	-.039	-.045	-.046	-.050	-.082	-.051	-.067	-.048	-.069	-.017	-.067	-.170	.800	
.800	.030	-.047	-.035	-.052	-.053	-.064	-.043	-.061	-.061	-.043	-.066	-.017	-.067	-.170	.850	
.850	.033	-.039	-.039	-.045	-.046	-.050	-.082	-.051	-.067	-.048	-.069	-.017	-.067	-.170	.900	
.900	.033	-.039	-.039	-.045	-.046	-.050	-.082	-.051	-.067	-.048	-.069	-.017	-.067	-.170	.950	
$\alpha = -2^\circ$																
.0125	.000	.019	.081	-.064	.114	-.080	.069	-.028	.067	-.038	.021	-.062	-.039	.015	-.094	.0125
.025	.049	-.020	.047	-.035	.049	-.028	.067	-.038	.021	-.062	-.030	.018	-.028	.028	.050	
.050	.032	-.015	.039	-.027	.044	-.027	.037	-.017	.016	-.012	-.030	.009	-.100	.075	.075	
.075	.036	-.010	.010	-.027	.030	-.032	.013	-.013	.019	-.009	-.028	.003	-.100	.100	.100	
.100	.056	-.001	.037	-.021	.002	-.022	.007	-.005	.003	-.010	-.017	.007	-.007	.043	.150	
.150	.044	-.017	.032	-.031	.017	-.008	.009	-.017	.007	-.027	-.027	.003	-.076	.075	.150	
.200	.000	-.026	.015	-.030	-.004	-.005	.004	-.005	.001	-.019	.014	-.019	.003	-.066	-.052	.200
.250	.027	-.025	.017	-.016	-.009	-.010	-.008	-.015	-.014	-.004	-.028	.002	-.059	-.068	.250	
.300	.022	-.015	.008	-.015	-.006	-.007	-.016	-.009	-.021	-.019	-.041	-.065	-.056	-.098	.300	
.350	.011	-.015	-.000	-.010	-.021	-.028	-.014	-.022	-.024	-.050	-.043	-.075	-.129	.400	.450	
.400	.005	-.020	-.006	-.024	-.003	-.008	-.069	-.020	-.070	-.003	-.070	-.028	-.113	-.091	.500	
.500	.054	-.035	-.035	-.063	-.047	-.030	-.049	-.028	-.053	-.021	-.077	-.016	-.098	-.106	.600	
.600	.045	-.039	-.042	-.032	-.054	-.045	-.058	-.035	-.066	-.030	-.091	-.033	-.106	-.153	.700	
.700	.047	-.030	-.053	-.030	-.072	-.037	-.052	-.031	-.078	-.010	-.091	-.045	-.102	-.117	.750	
.800	.047	-.030	-.053	-.030	-.072	-.037	-.052	-.031	-.078	-.010	-.091	-.045	-.102	-.117	.800	
.850	.052	-.026	-.056	-.026	-.069	-.021	-.100	-.020	-.079	-.021	-.035	-.035	-.122	-.090	.850	
.900	.052	-.026	-.056	-.026	-.069	-.021	-.100	-.020	-.079	-.021	-.035	-.035	-.122	-.090	.900	
.950	.052	-.026	-.056	-.026	-.069	-.021	-.100	-.020	-.079	-.021	-.035	-.035	-.122	-.090	.950	
$\alpha = 0^\circ$																
.0125	.044	.065	.018	.015	.014	.066	.055	.055	.109	.036	-.230	.127	-.292	-.292	.0125	
.025	.013	-.021	-.001	-.031	-.015	-.015	.041	-.055	-.055	.036	-.226	.105	-.278	.110	.050	
.050	.005	.016	.001	.022	-.010	.036	.042	.052	-.087	.094	-.199	.070	-.278	-.075	.075	
.075	.009	-.020	-.052	.014	-.080	.011	-.042	.043	-.076	.065	-.199	.075	-.278	-.132	.100	
.100	.031	.078	.011	.016	-.035	.025	-.046	.042	-.072	.066	-.119	.075	-.278	-.132	.100	
.150	.022	.006	-.003	-.014	.036	-.040	.021	-.070	.071	.064	-.239	.064	-.300	-.080	.200	
.200	.023	-.003	-.003	-.003	-.038	.029	-.036	.036	-.071	.064	-.274	.064	-.233	-.093	.200	
.250	.006	-.002	-.007	-.007	-.008	.021	-.048	-.018	-.061	.046	-.278	.036	-.212	-.012	.250	
.300	.006	.006	-.020	.013	-.033	.025	-.053	-.021	-.061	.034	-.282	.017	-.186	-.071	.300	
.350	.012	.014	-.026	.016	-.043	.004	-.044	-.009	-.065	-.008	-.088	-.008	-.140	-.085	.400	
.400	.012	.014	-.026	.016	-.043	.004	-.044	-.009	-.065	-.008	-.088	-.008	-.140	-.085	.450	
.500	.025	.005	-.029	.001	-.022	.024	-.096	-.014	-.101	.035	-.108	.018	-.140	-.045	.500	
.600	.071	-.003	-.054	-.034	-.066	-.006	-.077	-.003	-.091	.022	-.108	.021	-.125	-.054	.600	
.650	.058	-.020	-.062	-.016	-.072	-.022	-.084	-.010	-.097	-.006	-.118	.011	-.132	-.106	.700	
.700	.058	-.018	-.068	-.006	-.080	-.013	-.087	-.010	-.109	.021	-.117	.087	-.126	-.080	.800	
.750	.063	-.008	-.068	-.006	-.080	-.013	-.087	-.010	-.109	.021	-.117	.087	-.126	-.080	.850	
.800	.065	-.001	-.071	.002	-.080	-.004	-.110	-.005	-.103	.012	-.004	-.004	-.004	-.043	.900	
.850	.078	-.028	-.090	-.029	-.102	.036	-.136	-.044	-.125	.056	-.054	-.043	-.043	-.050	.950	
$\alpha = 2^\circ$																
.0125	-.008	.106	-.070	.076	-.154	.120	-.238	.126	-.256	.102	-.276	.207	-.290	-.156	.025	
.025	-.023	.016	-.060	.085	-.147	.094	-.233	.107	-.226	.164	-.276	.171	-.292	-.156	.050	
.050	-.018	.048	-.050	.063	-.070	.094	-.233	.107	-.226	.122	-.276	.180	-.292	-.075	.075	
.075	-.007	.049	-.077	.053	-.120	.068	-.212	.097	-.212	.112	-.280	.120	-.292	-.172	.100	
.100	-.003	.124	-.024	.048	-.083	.068	-.295	.094	-.204	.127	-.283	.129	-.292	-.136	.150	
.150	.001	.032	-.028	.024	-.056	.068	-.080	.078	-.196	.138	-.295	.106	-.303	-.042	.200	
.200	-.038	.026	-.036	.026	-.072	.073	-.076	.082	-.165	.111	-.295	.106	-.303	-.042	.200	
.250	-.014	.023	-.033	.036	-.120	.063	-.081	.016	-.113	.091	-.295	.106	-.303	-.035	.250	
.300	-.016	.031	-.043	.046	-.065	.065	-.089	.065	-.090	.096	-.291	.032	-.323	-.021	.300	
.350	-.032	.041	-.053	.046	-.075	.045	-.075	.049	-.083	-.024	-.074	.047	-.331	-.047	.400	
.400	-.043	.035	-.053	.031	-.052	.059	-.121	.058	-.127	.084	-.123	.075	-.346	-.007	.450	
.500	-.043	.035	-.053	.031	-.052	.059	-.121	.058	-.127	.084	-.123	.075	-.346	-.007	.500	
.550	-.060	.026	-.080	-.008	-.093	.029	-.100	.033	-.112	.066	-.092	.070	-.354	-.005	.550	
.600	-.095	.026	-.080	-.014	-.096	.009										

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING
(a) $M = 1.61$ - Continued

x/c , nominal	C_p at y/c of:														x/c , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 4$																
.0125	-.065	.147	-.213	.118	-.253	.157	-.263	.161	-.266	.106	-.295	.234	-.314		.0125	
.025	-.063	.097	-.218	.123	-.252	.146	-.263	.161	-.266	.106	-.295	.200	-.319	.183	.025	
.050	-.076	.077	-.108	.106	-.226	.164	-.268	.155	-.259	.220	-.295	.150	-.319		.050	
.075	-.037	.080	-.121	.095	-.248	.104	-.268	.137	-.262	.150	-.295	.150	-.319		.075	
.100	-.018	.146	-.065	.084	-.227	.114	-.273	.148	-.266	.166	-.301	.170	-.319		.100	
.150	-.025	.055	-.065	.061	-.195	.111	-.272	.125	-.277	.173	-.301	.160	-.324	.169	.150	
.200	-.059	.062	-.063	.060	-.172	.106	-.267	.121	-.307	.152	-.306	.148	-.327	.089	.200	
.250	-.036	.048	-.050	.072	-.131	.100	-.244	.067	-.310	.127	-.319	.131	-.331	.089	.250	
.300	-.042	.059	-.066	.084	-.076	.100	-.144	.104	-.320	.133	-.334	.055	-.342	.014	.300	
.350																
.400	-.058	.070	-.076	.081	-.099	.079	-.071	.087	-.326		-.359	.077	-.348	.011	.400	
.450																
.500	-.067	.060	-.074	.066	-.071	.090	-.145	.102	-.228	.120	-.367	.105	-.366	.059	.500	
.550																
.600	-.115	.056	-.098	.027	-.108	.060	-.120	.076	-.091	.093	-.367	.100	-.375	.040	.600	
.650																
.700	-.085	.032	-.096	.047	-.110	.038	-.123	.067	-.102	.072	-.367	.096	-.387	.028	.700	
.750																
.800	-.095	.047	-.103	.054	-.121	.050	-.123	.067	-.129	.105	-.337	.177	-.397	.004	.800	
.850																
.900	-.095	.059	-.105	.064	-.115	.069	-.153	.081	-.134	.091	-.291	.084		.900	.950	
.950																
$\alpha = 6$																
.0125	-.204	.184	-.255	.153	-.264	.182	-.276	.199	-.287	.132	-.318	.256	-.326		.0125	
.025	-.122	.131	-.263	.160	-.264	.186	-.283	.175	-.291	.211	-.316	.211	-.322	.214	.025	
.050	-.113	.111	-.260	.139	-.258	.175	-.290	.190	-.283	.211	-.316	.211	-.322	.214	.050	
.075	-.086	.115	-.264	.153	-.261	.152	-.283	.177	-.287	.188	-.316	.194	-.322	.175		
.100	-.060	.182	-.212	.120	-.283	.160	-.288	.195	-.287	.204	-.327	.205	-.332	.230	.100	
.150	-.051	.086	-.059	.098	-.293	.155	-.302	.165	-.300	.211	-.327	.202	-.336	.214	.150	
.200	-.080	.078	-.066	.093	-.296	.150	-.327	.160	-.320	.183	-.324	.198	-.341	.137	.200	
.250	-.059	.079	-.067	.109	-.275	.142	-.351	.101	-.337	.159	-.332	.177	-.345	.126	.250	
.300	-.064	.091	-.086	.118	-.173	.139	-.353	.139	-.356	.160	-.340	.096	-.354	.054	.300	
.350																
.400	-.084	.099	-.097	.115	-.077	.121	-.337	.133	-.372		-.363	.137	-.361	.031	.400	
.450																
.500	-.091	.093	-.094	.099	-.072	.128	-.387	.139	-.376	.162	-.387	.150	-.369	.105	.500	
.550																
.600	-.132	.088	-.115	.061	-.118	.095	-.115	.112	-.371	.132	-.401	.148	-.376	.081	.600	
.650																
.700	-.101	.063	-.112	.079	-.124	.076	-.119	.108	-.339	.104	-.407	.136	-.386	.019	.700	
.750																
.800	-.114	.081	-.119	.087	-.134	.088	-.177	.108	-.286	.137	-.409	.234	-.396	.051	.800	
.850																
.900	-.112	.096	-.121	.098	-.129	.106	-.164	.117	-.202	.126						
.950																
$\alpha = 8$																
.0125	-.310	.229	-.288	.169	-.288	.204	-.295	.219	-.310	.151	-.333	.283	-.342		.0125	
.025	-.182	.170	-.292	.194	-.292	.218	-.295	.219	-.307	.281	-.330	.256	-.347	.050	.025	
.050	-.139	.157	-.296	.178	-.289	.236	-.300	.237	-.307	.226	-.331	.219	-.350	.075		
.075	-.089	.161	-.321	.162	-.317	.188	-.306	.218	-.307	.226	-.331	.233	-.369	.100		
.100	-.069	.205	-.325	.155	-.315	.199	-.312	.241	-.310	.249	-.336	.233	-.350	.150		
.150	-.069	.119	-.314	.135	-.347	.198	-.320	.212	-.331	.255	-.347	.247	-.350	.150		
.200	-.099	.110	-.077	.126	-.363	.190	-.340	.205	-.343	.247	-.343	.242	-.354	.169	.200	
.250	-.074	.112	-.059	.142	-.372	.184	-.378	.140	-.350	.215	-.358	.214	-.358	.169	.250	
.300	-.081	.124	-.088	.155	-.373	.178	-.400	.176	-.360	.199	-.363	.174	-.367	.091	.300	
.350																
.400	-.102	.133	-.111	.147	-.267	.165	-.404	.171	-.389		-.369	.174	-.373	.071	.400	
.450																
.500	-.106	.125	-.116	.133	-.068	.166	-.388	.177	-.416	.219	-.385	.193	-.385	.145	.500	
.550																
.600	-.145	.121	-.133	.092	-.133	.136	-.329	.146	-.416	.180	-.404	.191	-.384	.122	.600	
.650																
.700	-.114	.096	-.129	.113	-.139	.113	-.428	.139	-.425	.155	-.420	.181	-.385	.059	.700	
.750																
.800	-.131	.114	-.136	.121	-.154	.129	-.474	.139	-.420	.193	-.427	.285	-.389	.122	.800	
.850																
.900	-.128	.131	-.138	.130	-.149	.147	-.406	.158	-.390	.176	-.423	.174	-.392	.092	.900	
.950																
$\alpha = 10$																
.0125	-.346	.295	-.316	.189	-.309	.219	-.323	.239	-.340	.176	-.361	.299	-.363		.0125	
.025	-.345	.209	-.318	.229	-.314	.251	-.323	.274	-.333	.205	-.359	.272	-.376	.050	.025	
.050	-.345	.209	-.318	.229	-.314	.251	-.323	.274	-.333	.205	-.359	.272	-.376	.075		
.075	-.083	.202	-.350	.212	-.343	.232	-.327	.257	-.334	.257	-.356	.245	-.369	.100		
.100	-.078	.239	-.372	.199	-.340	.241	-.335	.281	-.336	.236	-.363	.276	-.369	.279		
.150	-.081	.156	-.294	.178	-.346	.234	-.347	.256	-.350	.289	-.374	.292	-.369	.279	.150	
.200	-.116	.150	-.372	.163	-.377	.234	-.355	.266	-.373	.265	-.360	.282	-.373	.212	.200	
.250	-.094	.149	-.164	.185	-.403	.225	-.370	.179	-.368	.254	-.374	.260	-.374	.206	.250	
.300	-.100	.160	-.109	.199	-.419	.219	-.397	.216	-.372	.228	-.383	.168	-.380	.127	.300	
.350																
.400	-.120	.171	-.123	.189	-.419	.209	-.421	.210	-.387		-.390	.209	-.388	.113	.400	
.450																
.500	-.128	.162	-.126	.177	-.297	.203	-.447	.213	-.411	.259	-.390	.233	-.402	.189	.500	
.550																
.600	-.160	.159	-.150	.133	-.221	.174	-.441	.184	-.425	.215	-.399	.233	-.402	.175	.600	
.650																
.700	-.127	.134	-.141	.154	-.177	.150	-.399	.179	-.456	.190	-.413	.2				

TABLE VI - Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING
(a) $M = 1.61$ - Continued

x/c, nominal	Cp at y/b of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 12$																
.0125	-.347	.311	-.329	.211	-.333	.234	-.342	.259	-.357	.200	-.378	.311	-.381	.296	.0125	
.025	-.365	.250	-.334	.261	-.336	.275	-.348	.307	-.352	.338	-.375	.303	-.386	.296	.025	
.050	-.376	.238	-.341	.257	-.335	.287	-.348	.307	-.352	.338	-.375	.303	-.386	.296	.050	
.075	-.291	.245	-.358	.259	-.359	.261	-.370	.290	-.355	.291	-.374	.312	-.385	.315	.075	
.100	-.294	.283	-.368	.260	-.362	.282	-.355	.317	-.357	.327	-.378	.331	-.385	.313	.100	
.125	-.100	.199	-.400	.218	-.362	.277	-.371	.295	-.365	.330	-.384	.308	-.388	.322	.200	
.200	-.124	.193	-.436	.201	-.378	.273	-.378	.284	-.384	.308	-.378	.322	-.388	.250	.200	
.300	-.110	.193	-.348	.228	-.409	.267	-.384	.219	-.394	.274	-.396	.289	-.388	.246	.250	
.350	-.115	.203	-.261	.241	-.433	.263	-.396	.257	-.394	.274	-.396	.209	-.395	.164	.300	
.400	-.133	.211	-.163	.231	-.455	.253	-.422	.245	-.406		-.411	.259	-.402	.161	.400	
.450															.450	
.500	-.146	.203	-.152	.218	-.402	.245	-.454	.258	-.421	.295	-.420	.281	-.414	.237	.500	
.550															.550	
.600	-.176	.199	-.172	.175	-.356	.218	-.466	.227	-.430	.263	-.423	.276	-.419	.220	.600	
.650															.650	
.700	-.138	.179	-.162	.192	-.286	.194	-.449	.227	-.446	.237	-.427	.272	-.425	.148	.700	
.750															.750	
.800	-.162	.195	-.173	.206	-.276	.214	-.423	.235	-.461	.283	-.437	.376	-.428	.192	.800	
.850															.850	
.900	-.160	.218	-.174	.218	-.229	.232	-.407	.249	-.467	.265		.271			.900	
.950															.950	
$a = 14$																
.0125	-.356	.325	-.349	.245	-.356	.249	-.368	.285	-.391	.226	-.400	.334	-.402	.327	.0125	
.025	-.380	.298	-.356	.300	-.359	.302	-.368	.302	-.372	.349	-.380	.372	-.398	.349	.025	
.050	-.420	.281	-.361	.300	-.361	.330	-.372	.349	-.384	.327	-.400	.329	-.404	.327	.050	
.075	-.409	.299	-.392	.299	-.380	.310	-.375	.336	-.384	.327	-.400	.359	-.349	.100		
.100	-.272	.337	-.399	.287	-.375	.332	-.378	.366	-.384	.367	-.402	.359	-.376	.357	.150	
.125	-.131	.246	-.417	.268	-.393	.330	-.378	.344	-.391	.374	-.400	.372	-.407	.302	.200	
.200	-.131	.242	-.432	.251	-.422	.326	-.388	.336	-.395	.362	-.400	.353	-.407	.297	.250	
.250	-.131	.236	-.428	.278	-.426	.321	-.419	.275	-.395	.336	-.405	.353	-.412	.214	.300	
.300	-.136	.254	-.382	.293	-.424	.316	-.431	.310	-.405	.320	-.404	.255	-.412		.350	
.350															.350	
.400	-.154	.261	-.277	.281	-.437	.310	-.424	.305	-.435		-.412	.308	-.417	.233	.400	
.450															.450	
.500	-.172	.256	-.222	.271	-.449	.305	-.436	.313	-.440	.349	-.432	.334	-.418	.308	.500	
.550															.550	
.600	-.196	.252	-.212	.225	-.433	.269	-.443	.276	-.435	.313	-.447	.347	-.421	.299	.600	
.650															.650	
.700	-.155	.231	-.203	.241	-.391	.248	-.455	.280	-.460	.306	-.451	.348	-.425	.324	.750	
.750															.750	
.800	-.184	.246	-.202	.258	-.358	.281	-.460	.301	-.446	.360	-.447	.467	-.433	.292	.800	
.850															.850	
.900	-.174	.292	-.206	.294	-.330	.312	-.447	.330	-.454	.355	-.364				.900	
.950															.950	
$a = 16$																
.0125	-.365	.354	-.365	.270	-.377	.256	-.387	.307	-.392	.254	-.409	.355	-.416	.358	.0125	
.025	-.384	.338	-.371	.332	-.379	.325	-.387	.307	-.392	.326	-.409	.382	-.418	.358	.025	
.050	-.431	.323	-.371	.343	-.381	.363	-.391	.381	-.390	.401	-.406	.361	-.418	.375	.050	
.075	-.444	.359	-.415	.342	-.397	.349	-.392	.371	-.395	.358	-.406	.407	-.419	.392	.100	
.100	-.373	.379	-.429	.342	-.375	.376	-.397	.405	-.395	.404	-.407	.407	-.419	.406	.150	
.125	-.215	.286	-.420	.313	-.435	.393	-.415	.380	-.409	.415	-.416	.414	-.418	.380	.200	
.200	-.155	.285	-.427	.294	-.436	.367	-.407	.313	-.404	.405	-.410	.419	-.418	.360	.250	
.250	-.159	.280	-.424	.314	-.434	.363	-.432	.315	-.400	.392	-.413	.396	-.418	.360	.250	
.300	-.163	.297	-.430	.337	-.430	.362	-.448	.358	-.416	.363	-.413	.303	-.423	.281	.300	
.350															.350	
.400	-.176	.303	-.360	.327	-.437	.359	-.435	.359	-.445		-.417	.375	-.425	.291	.400	
.450															.450	
.500	-.196	.302	-.297	.318	-.458	.355	-.446	.369	-.450	.411	-.437	.407	-.425	.379	.500	
.550															.550	
.600	-.214	.295	-.271	.268	-.459	.314	-.447	.340	-.444	.389	-.458	.423	-.425	.379	.600	
.650															.650	
.700	-.169	.276	-.246	.297	-.428	.317	-.457	.364	-.444	.386	-.466	.430	-.430	.339	.700	
.750															.750	
.800	-.197	.310	-.238	.332	-.402	.352	-.473	.382	-.449	.442	-.460	.555	-.440	.470	.800	
.850															.850	
.900	-.185	.387	-.232	.396	-.377	.425	-.468	.485	-.453	.590		.661		.581	.900	
.950															.950	
$a = 18$																
.0125	-.370	.387	-.384	.366	-.392	.356	-.400	.326	-.407	.272	-.423	.383	-.425	.397	.0125	
.025	-.400	.384	-.388	.384	-.396	.398	-.403	.411	-.407	.424	-.423	.419		.397	.050	
.050	-.444	.369	-.384	.386	-.394	.398	-.387	.404	-.407	.387	-.423	.405	-.427	.475	.075	
.075	-.445	.398	-.416	.381	-.409	.407	-.405	.404	-.407						.100	
.100	-.423	.423	-.439	.371	-.400	.418	-.409	.441	-.410	.436	-.423	.446			.150	
.125	-.308	.329	-.430	.354	-.412	.412	-.404	.430	-.414	.458	-.427	.461			.200	
.200	-.207	.327	-.434	.343	-.455	.409	-.414	.425	-.417	.442	-.418	.476	-.427	.416	.200	
.250	-.197	.323	-.448	.370	-.455	.405	-.446	.356	-.414	.457	-.421	.459	-.427	.416	.250	
.300	-.191	.347	-.459	.381	-.446	.410	-.463	.406	-.428	.420	-.418	.367	-.433	.348	.300	
.350															.350	
.400	-.209	.346	-.409	.373	-.450	.405	-.450	.422	-.459		-.421	.442	-.433	.395	.400	
.450															.450	
.500	-.214	.352	-.352	.366	-.466	.416	-.454	.431	-.466	.477	-.446	.484	-.430	.573	.500	
.600	-.228	.347	-.320	.329	-.480	.386	-.454	.410	-.454	.468	-.467	.564	-.430	.628	.600	
.700	-.182	.349	-.280	.376	-.454	.398	-.465	.442	-.454	.490	-.475	.687	-.430	.572	.700	
.750															.750	
.800	-.206	.394	-.276	.416	-.											

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING
(a) M = 1.61 - Concluded

x/c, nominal	Cp at $y/c = \frac{1}{2}$ of :														x/c, nominal	
	.10		.25		.40		.50		.60		.70		.80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 20^\circ$																
.0125	-.378	.422	-.393	.317	-.399	.292	-.406	.350	-.414	.320	-.426	.426	-.428	.476	.0125	
.025	-.406	.450	-.399	.406	-.397	.397	-.406	.436	-.409	.450	-.411	.482	-.423	.472	.025	
.050	-.442	.419	-.393	.431	-.400	.436	-.409	.436	-.412	.447	-.411	.452	-.423	.463	.050	
.075	-.440	.443	-.423	.424	-.413	.433	-.412	.447	-.411	.452	-.423	.463	-.428	.476	.075	
.100	-.447	.474	-.448	.423	-.403	.468	-.416	.489	-.412	.512	-.423	.500	-.423	.546	.100	
.150	-.371	.379	-.442	.404	-.424	.459	-.408	.495	-.417	.543	-.419	.539	-.429	.622	.150	
.200	-.271	.379	-.444	.403	-.465	.458	-.424	.499	-.418	.535	-.424	.546	-.429	.602	.200	
.250	-.246	.373	-.452	.426	-.462	.466	-.457	.435	-.419	.538	-.424	.540	-.429	.608	.250	
.300	-.238	.396	-.473	.433	-.452	.486	-.473	.493	-.438	.503	-.424	.459	-.433	.588	.300	
.350	-.235	.398	-.437	.436	-.455	.507	-.457	.502	-.463	-.428	-.452	-.433	-.433	.585	.350	
.400	-.231	.419	-.388	.459	-.466	.501	-.461	.515	-.468	.582	-.454	.720	-.434	.657	.400	
.450	-.231	.419	-.388	.459	-.466	.501	-.461	.515	-.468	.582	-.454	.720	-.434	.657	.450	
.500	-.231	.419	-.388	.459	-.466	.501	-.461	.515	-.468	.582	-.454	.720	-.434	.657	.500	
.550	-.236	.440	-.361	.417	-.486	.472	-.459	.515	-.457	.702	-.473	.740	-.434	.662	.600	
.600	-.236	.440	-.361	.417	-.466	.468	-.466	.549	-.468	.695	-.457	.722	-.478	.735	.650	
.650	-.197	.432	-.315	.468	-.466	.549	-.468	.595	-.457	.722	-.478	.735	-.439	.588	.700	
.700	-.220	.565	-.305	.666	-.450	.691	-.485	.706	-.461	.739	-.470	.810	-.457	.665	.750	
.750	-.220	.565	-.305	.666	-.450	.691	-.485	.706	-.461	.739	-.470	.810	-.457	.599	.800	
.800	-.220	.565	-.305	.666	-.450	.691	-.485	.706	-.461	.739	-.470	.810	-.457	.599	.850	
.850	-.209	.653	-.297	.666	-.430	.679	-.490	.686	-.459	.695	-.459	.679	-.457	.599	.900	
.900	-.209	.653	-.297	.666	-.430	.679	-.490	.686	-.459	.695	-.459	.679	-.457	.599	.950	
.950	-.209	.653	-.297	.666	-.430	.679	-.490	.686	-.459	.695	-.459	.679	-.457	.599	.950	

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING
(b) $M_\infty = 2.01$

$\frac{x}{c}$, nominal	C_p at $\sqrt{\frac{V}{2}}$ of:														$\frac{x}{c}$, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -20^\circ$																
.0125	.425	-.270	.357	-.281	.299	-.285	.251	-.287	.212	-.276	.169	-.282	.228	-.280	.0125	
.025	.381	-.280	.334	-.281	.317	-.285	.301	-.287	.281	-.275	.213	-.281	.206	-.280	.025	
.050	.343	-.288	.369	-.281	.352	-.285	.307	-.275	.276	-.276	.241	-.282	.206	-.280	.050	
.075	.385	-.294	.324	-.284	.270	-.285	.311	-.275	.307	-.276	.241	-.282	.206	-.280	.075	
.100	.392	-.295	.326	-.285	.313	-.289	.293	-.284	.295	-.278	.234	-.278	.206	-.284	.100	
.150	.378	-.295	.361	-.297	.325	-.284	.317	-.285	.279	-.277	.278	-.278	.227	-.287	.150	
.200	.308	-.248	.332	-.302	.321	-.296	.292	-.266	.247	-.276	.254	-.281	.216	-.287	.200	
.250	.349	-.198	.342	-.300	.245	-.306	.281	-.297	.261	-.277	.233	-.280	.209	-.282	.250	
.300	.331	-.196	.319	-.300	.296	-.302	.266	-.302	.249	-.277	.222	-.285	.188	-.288	.300	
.350	.400	-.197	.308	-.294	.279	-.299	.240	-.302	.234	-.277	.207	-.285	.173	-.281	.400	
.400	.370	-.197	.308	-.294	.279	-.299	.240	-.302	.234	-.277	.188	-.269	.155	-.281	.450	
.450	.323	-.148	.289	-.284	.281	-.303	.206	-.294	.191	-.297	.188	-.269	.155	-.281	.500	
.500	.355	-.148	.289	-.284	.281	-.303	.206	-.294	.191	-.297	.188	-.269	.155	-.281	.550	
.600	.189	-.193	.240	-.262	.215	-.303	.195	-.301	.196	-.287	.169	-.283	.150	-.283	.600	
.650	.190	-.175	.204	-.247	.183	-.307	.172	-.290	.161	-.292	.149	-.294	.139	-.287	.700	
.700	.221	-.190	.195	-.233	.160	-.302	.159	-.293	.146	-.291	.134	-.293	.138	-.280	.750	
.800	.221	-.190	.195	-.233	.160	-.302	.159	-.293	.146	-.291	.134	-.293	.138	-.281	.850	
.900	.209	-.194	.181	-.218	.164	-.296	.114	-.293	.137	-.290	.140	-.294		-.281	.900	
.950	.209	-.194	.181	-.218	.164	-.296	.114	-.293	.137	-.290	.140	-.294		-.281	.950	
$\alpha = -18^\circ$																
.0125	.401	-.266	.327	-.285	.280	-.281	.243	-.281	.198	-.272	.158	-.279	.221	-.276	.0125	
.025	.347	-.276	.299	-.285	.280	-.281	.276	-.280	.260	-.276	.198	-.280	.221	-.276	.050	
.050	.300	-.286	.329	-.285	.314	-.281	.276	-.280	.260	-.276	.220	-.282	.193	-.276	.075	
.075	.343	-.287	.287	-.285	.284	-.281	.281	-.280	.270	-.272	.211	-.280	.221	-.282	.100	
.100	.351	-.300	.270	-.286	.269	-.281	.266	-.280	.267	-.272	.227	-.285	.205	-.285	.150	
.150	.326	-.285	.315	-.305	.281	-.283	.279	-.279	.250	-.272	.231	-.282	.196	-.288	.200	
.200	.267	-.205	.291	-.310	.270	-.294	.261	-.262	.221	-.272	.231	-.282	.196	-.288	.250	
.250	.309	-.158	.297	-.310	.199	-.307	.248	-.293	.232	-.278	.206	-.283	.187	-.285	.300	
.300	.294	-.162	.272	-.309	.255	-.307	.235	-.299	.219	-.275	.198	-.287	.165	-.290	.350	
.350	.400	-.152	.267	-.290	.238	-.303	.223	-.303	.205	-.275	.183	-.284	.151	-.288	.400	
.400	.326	-.152	.267	-.290	.238	-.303	.223	-.303	.205	-.275	.164	-.269	.129	-.288	.450	
.450	.291	-.127	.250	-.278	.242	-.306	.178	-.296	.161	-.298	.164	-.269	.129	-.288	.500	
.500	.355	-.127	.250	-.278	.242	-.306	.178	-.296	.161	-.298	.164	-.269	.129	-.288	.550	
.600	.156	-.168	.201	-.239	.172	-.307	.171	-.298	.164	-.289	.147	-.282	.128	-.287	.600	
.650	.153	-.156	.164	-.220	.148	-.312	.144	-.293	.130	-.299	.123	-.295	.116	-.291	.700	
.700	.153	-.156	.164	-.220	.148	-.312	.144	-.293	.130	-.299	.123	-.295	.116	-.282	.750	
.800	.184	-.171	.157	-.205	.122	-.301	.137	-.296	.116	-.296	.110	-.299	.112	-.285	.800	
.850	.173	-.174	.145	-.190	.127	-.291	.092	-.298	.107	-.295	.115	-.302		-.285	.850	
.900	.125	-.165	.107	-.161	.095	-.274	.054	-.304	.077	-.296	.083	-.301		-.280	.900	
.950	.097	-.146	.083	-.136	.065	-.199	.029	-.309	.043	-.299	.053	-.284		-.281	.950	
$\alpha = -16^\circ$																
.0125	.364	-.264	.292	-.283	.267	-.281	.227	-.282	.184	-.267	.140	-.274	.212	-.265	.0125	
.025	.310	-.274	.281	-.283	.253	-.280	.227	-.282	.184	-.261	.178	-.275	.212	-.265	.050	
.050	.275	-.291	.285	-.286	.289	-.280	.246	-.283	.238	-.264	.201	-.278	.178	-.275	.075	
.075	.286	-.299	.235	-.283	.200	-.280	.240	-.283	.254	-.264	.227	-.277	.186	-.275	.100	
.100	.296	-.305	.240	-.290	.234	-.282	.223	-.283	.237	-.265	.185	-.277	.186	-.275	.150	
.150	.288	-.247	.269	-.309	.243	-.281	.235	-.283	.217	-.266	.227	-.277	.186	-.275	.200	
.200	.220	-.141	.242	-.314	.234	-.290	.222	-.265	.187	-.269	.200	-.273	.170	-.276	.250	
.250	.255	-.127	.252	-.315	.161	-.302	.208	-.299	.197	-.273	.178	-.278	.163	-.276	.300	
.300	.246	-.138	.225	-.319	.218	-.310	.193	-.303	.196	-.272	.168	-.280	.143	-.283	.350	
.350	.400	-.132	.221	-.294	.200	-.310	.184	-.310	.171	-.275	.150	-.278	.125	-.288	.400	
.400	.272	-.132	.221	-.294	.200	-.310	.184	-.310	.171	-.275	.150	-.278	.125	-.288	.450	
.450	.234	-.120	.207	-.239	.202	-.314	.142	-.296	.131	-.292	.130	-.267	.107	-.280	.500	
.500	.234	-.120	.207	-.239	.202	-.314	.142	-.296	.131	-.292	.130	-.267	.107	-.280	.550	
.600	.112	-.154	.162	-.187	.140	-.317	.132	-.302	.134	-.286	.115	-.281	.102	-.277	.600	
.650	.112	-.154	.162	-.187	.140	-.317	.132	-.302	.134	-.286	.115	-.281	.102	-.277	.650	
.700	.099	-.147	.125	-.178	.114	-.316	.107	-.298	.097	-.290	.093	-.292	.092	-.288	.700	
.750	.138	-.161	.118	-.168	.091	-.300	.096	-.300	.084	-.296	.080	-.298	.089	-.280	.750	
.800	.125	-.165	.107	-.161	.095	-.274	.054	-.304	.077	-.296	.083	-.301		-.280	.800	
.850	.097	-.146	.083	-.136	.065	-.199	.029	-.309	.043	-.299	.053	-.284		-.281	.850	
.900	.097	-.146	.083	-.136	.065	-.199	.029	-.309	.043	-.299	.053	-.284		-.281	.900	
.950	.097	-.146	.083	-.136	.065	-.199	.029	-.309	.043	-.299	.053	-.284		-.281	.950	
$\alpha = -14^\circ$																
.0125	.308	-.249	.266	-.270	.249	-.276	.213	-.275	.163	-.264	.125	-.267	.200	-.265	.0125	
.025	.266	-.265	.256	-.268	.259	-.276	.213	-.275	.163	-.256	.160	-.264	.239	-.265	.050	
.050	.239	-.275	.247	-.270	.262	-.272	.219	-.275	.163	-.256	.160	-.264	.239	-.265	.075	
.075	.250	-.277	.204	-.271	.201	-.275	.189	-.275	.163	-.256	.160	-.264	.239	-.265	.100	
.100	.264	-.272	.228	-.278	.201	-.275	.189	-.275	.163	-.256	.160	-.264	.239	-.265	.150	
.150	.238	-.161	.231	-.299	.210	-.279	.195	-.274	.178	-.257	.167	-.267	.159	-.271	.200	
.200	.184	-.104	.209	-.304	.193	-.283	.183	-.260	.151	-.264	.170	-.262	.145	-.266	.250	
.250	.216	-.114	.212	-.312	.128	-.292	.174	-.287	.157	-.271	.145	-.267	.135	-.263	.300	
.300	.204	-.124	.189	-.312	.183	-.306	.159	-.292	.150	-.273	.139	-.271	.116	-.273	.350	
.400	.231	-.115	.186	-.247	.167	-.314	.155	-.298	.134	-.275	.118	-.271	.097	-.288	.400</td	

TABLE VI - Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING

(b) $M = 2.01$ - Continued

x/c, nominal	C_p at $y/\frac{c}{2}$ of:														x/c, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = -12^\circ$																
.0125	.269	-.224	.236	-.260	.223	-.255	.195	-.257	.152	-.246	.114	-.245	.188	-.240	.0125	
.025	.220	-.245	.200	-.260	.216	-.245	.188	-.257	.183	-.240	.146	-.242	.200	-.240	.025	
.050	.189	-.218	.209	-.263	.216	-.245	.177	-.257	.190	-.242	.154	-.248	.145	-.240	.050	
.075	.198	-.173	.166	-.263	.126	-.255	.152	-.257	.166	-.241	.134	-.244	.145	-.240	.075	
.100	.204	-.168	.174	-.274	.159	-.255	.144	-.257	.144	-.246	.120	-.244	.141	-.242	.100	
.150	.193	-.122	.183	-.290	.161	-.261	.153	-.258	.146	-.241	.127	-.248	.131	-.247	.150	
.200	.149	-.101	.160	-.290	.144	-.270	.120	-.280	.117	-.253	.113	-.248	.113	-.243	.200	
.250	.170	-.100	.166	-.295	.084	-.280	.133	-.273	.127	-.248	.117	-.253	.096	-.255	.300	
.300	.159	-.113	.142	-.281	.131	-.289	.120	-.280	.117	-.253	.113	-.253	.096	-.255	.350	
.350	.173	-.104	.139	-.127	.118	-.300	.104	-.292	.106	-.261	.093	-.250	.074	-.240	.400	
.400	.165	-.090	.137	-.102	.131	-.304	.081	-.287	.068	-.261	.076	-.244	.055	-.246	.450	
.450	.159	-.127	.087	-.130	.075	-.283	.069	-.284	.071	-.264	.063	-.253	.054	-.249	.500	
.500	.051	-.127	.067	-.130	.047	-.214	.050	-.293	.042	-.282	.043	-.257	.042	-.255	.600	
.550	.028	-.126	.057	-.130	.047	-.214	.050	-.293	.042	-.282	.043	-.257	.042	-.255	.650	
.600	.065	-.130	.050	-.130	.025	-.143	.035	-.298	.029	-.287	.031	-.260	.038	-.258	.800	
.650	.051	-.129	.038	-.129	.028	-.100	.003	-.283	.021	-.284	.031	-.265		-.258	.850	
.700	.051	-.129	.038	-.129	.028	-.100	.003	-.283	.021	-.284	.031	-.265		-.258	.900	
.750	.050	-.129	.038	-.129	.028	-.100	.003	-.283	.021	-.284	.031	-.265		-.258	.950	
$\alpha = -10^\circ$																
.0125	.232	-.205	.210	-.252	.213	-.251	.181	-.252	.136	-.237	.101	-.230	.165	-.222	.0125	
.025	.183	-.226	.187	-.253	.182	-.250	.160	-.250	.154	-.232	.128	-.231		-.232	.025	
.050	.145	-.197	.175	-.256	.184	-.250	.148	-.254	.160	-.233	.127	-.236	.120	-.237	.050	
.075	.159	-.077	.141	-.257	.115	-.250	.125	-.250	.114	-.231	.111	-.233	.114	-.237	.100	
.100	.164	-.068	.166	-.265	.134	-.254	.125	-.250	.113	-.229	.104	-.227	.104	-.237	.150	
.150	.152	-.096	.153	-.266	.136	-.254	.125	-.250	.113	-.229	.104	-.227	.099	-.233	.200	
.200	.107	-.090	.131	-.262	.117	-.253	.108	-.251	.098	-.232	.094	-.230	.092	-.233	.250	
.250	.138	-.090	.134	-.224	.054	-.271	.098	-.268	.098	-.237	.093	-.236	.084	-.233	.300	
.300	.126	-.094	.112	-.099	.104	-.278	.087	-.268	.087	-.240	.084	-.240	.066	-.244	.350	
.350	.133	-.099	.108	-.085	.088	-.284	.081	-.281	.076	-.274	.066	-.274	.052	-.274	.400	
.400	.109	-.079	.099	-.098	.103	-.270	.051	-.272	.045	-.254	.048	-.229	.030	-.231	.450	
.450	.035	-.107	.065	-.121	.054	-.155	.042	-.269	.047	-.258	.039	-.238	.028	-.233	.500	
.500	.035	-.107	.037	-.114	.030	-.098	.022	-.271	.019	-.272	.020	-.242	.017	-.245	.600	
.550	.003	-.112	.037	-.114	.030	-.098	.022	-.271	.019	-.272	.020	-.242	.017	-.245	.700	
.600	.031	-.113	.027	-.112	.007	-.110	.009	-.244	.006	-.276	.008	-.247	.014	-.235	.750	
.650	.030	-.106	.023	-.110	.012	-.100	.020	-.158	.003	-.272	.009	-.254		-.239	.800	
.700	.030	-.106	.023	-.110	.012	-.100	.020	-.158	.003	-.272	.009	-.254		-.239	.850	
.750	.030	-.106	.023	-.110	.012	-.100	.020	-.158	.003	-.272	.009	-.254		-.239	.900	
.800	.030	-.106	.023	-.110	.012	-.100	.020	-.158	.003	-.272	.009	-.254		-.239	.950	
$\alpha = -8^\circ$																
.0125	.201	-.145	.180	-.237	.204	-.232	.185	-.236	.120	-.215	.075	-.210	.150	-.215	.0125	
.025	.153	-.153	.163	-.238	.162	-.233	.137	-.237	.131	-.212	.103	-.210		-.210	.025	
.050	.125	-.164	.150	-.238	.153	-.233	.119	-.241	.139	-.212	.100	-.213	.102	-.216	.050	
.075	.129	-.052	.140	-.237	.107	-.235	.098	-.237	.114	-.212	.078	-.212		-.216	.075	
.100	.130	-.042	.140	-.236	.112	-.232	.098	-.237	.114	-.212	.078	-.212		-.216	.100	
.150	.123	-.072	.119	-.195	.107	-.235	.096	-.236	.089	-.209	.071	-.211	.076	-.216	.150	
.200	.081	-.074	.102	-.120	.088	-.235	.078	-.224	.063	-.211	.083	-.207	.070	-.211	.200	
.250	.108	-.075	.107	-.081	.029	-.235	.066	-.243	.072	-.212	.066	-.212	.060	-.210	.250	
.300	.093	-.077	.089	-.078	.078	-.230	.058	-.236	.066	-.218	.056	-.217	.042	-.220	.300	
.350	.101	-.082	.080	-.077	.064	-.200	.052	-.236	.054	-.223	.038	-.209	.027	-.207	.350	
.400	.080	-.062	.074	-.086	.079	-.105	.025	-.216	.022	-.223	.022	-.203	.003	-.206	.400	
.450	.021	-.091	.042	-.102	.034	-.085	.024	-.190	.028	-.217	.015	-.205	.005	-.204	.450	
.500	.019	-.097	.015	-.098	.015	-.092	.005	-.159	.000	-.227	.001	-.209	.001	-.213	.500	
.550	.002	-.090	.006	-.096	.011	-.101	.009	-.084	.007	-.220	.011	-.210	.004	-.202	.550	
.600	.009	-.092	.002	-.090	.006	-.094	.038	-.064	.017	-.203	.011	-.214		-.202	.600	
.650	.019	-.085	.002	-.088	.003	-.101	.017	-.091	.027	-.113	.029	-.120	.030	-.166	.650	
.700	.010	-.072	.018	-.084	.029	-.098	.031	-.086	.034	-.096	.034	-.116	.027	-.142	.700	
.750	.010	-.072	.018	-.084	.029	-.098	.031	-.086	.034	-.096	.034	-.116		-.134	.800	
.800	.011	-.072	.018	-.084	.029	-.098	.031	-.086	.034	-.096	.034	-.116		-.134	.850	
.850	.011	-.072	.018	-.084	.029	-.098	.031	-.086	.034	-.096	.034	-.116		-.134	.900	
.900	.011	-.072	.018	-.084	.029	-.098	.031	-.086	.034	-.096	.034	-.116		-.134	.950	
$\alpha = -6^\circ$																
.0125	.156	-.104	.160	-.192	.178	-.244	.140	-.241	.097	-.231	.059	-.213	.099	-.213	.0125	
.025	.115	-.096	.132	-.185	.128	-.236	.106	-.238	.097	-.226	.072	-.215		-.213	.050	
.050	.092	-.104	.115	-.196	.117	-.236	.074	-.244	.083	-.237	.111	-.227	.063	-.214	.075	
.075	.097	-.029	.086	-.198	.074	-.222	.065	-.230	.069	-.226	.046	-.206		-.214	.100	
.100	.101	-.016	.133	-.082	.122	-.222	.056	-.226	.050	-.206	.016	-.191	.033	-.214	.100	
.150	.096	-.056	.089	-.107	.054	-.117	.045	-.117	.045	-.109	.024	-.109	.033	-.173	.150	
.200	.050	-.056	.073	-.080	.056	-.114	.049	-.117	.049	-.109	.024	-.109	.032	-.166	.200	
.250	.077	-.056	.072	-.070	.000	-.105	.019	-.117	.033	-.113	.024	-.113	.020	-.169	.250	
.300	.167	-.060	.056	-.073	.044	-.096	.025	-.132	.026	-.163	.020	-.169	.011	-.215	.300	
.350	.059	-.070	.039	-.070	.028	-.103	.018	-.112	.017	-.122	.002	-.152	.001	-.201	.350	
.400	.036	-.048	.033	-.075	.038	-.089	.014	-.089	.019	-.116	.021	-.135	.033	-.168	.400	
.450	.013	-.075	.023	-.091	.012	-.094	.005	-.096	.012	-.107	.019	-.124	.026	-.156	.450	
.500	.013	-.075	.023</td													

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING

TABLE VI.- Continued
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING

(b) $M = 2.01$ - Continued

$\frac{c}{c}$, nominal	Cp at $y/\frac{c}{2}$ of:														$\frac{c}{c}$, nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$\alpha = 4^\circ$																
.0125	-.057	.150	-.151	.112	-.175	.151	-.211	.144	-.214	.140	-.227	.215	-.247	.179	.0125	
.025	-.057	.084	-.146	.112	-.169	.131	-.201	.136	-.210	.189	-.224	.185	-.250	.050	.025	
.050	-.062	.073	-.072	.088	-.169	.121	-.204	.125	-.213	.154	-.228	.150	-.250	.075	.050	
.075	-.034	.086	-.080	.080	-.172	.096	-.204	.125	-.213	.154	-.227	.152	-.252	.100	.075	
.100	-.023	.110	-.065	.052	-.160	.096	-.204	.117	-.216	.145	-.227	.152	-.250	.125	.100	
.150	-.030	.050	-.060	.052	-.110	.082	-.164	.091	-.215	.132	-.232	.136	-.252	.165	.125	
.200	-.053	.043	-.060	.052	-.117	.072	-.153	.042	-.206	.116	-.233	.121	-.252	.195	.150	
.250	-.036	.031	-.052	.056	-.106	.072	-.133	.073	-.200	.135	-.232	.078	-.256	.200	.175	
.300	-.047	.030	-.066	.053	-.106	.072	-.133	.073	-.200	.135	-.232	.078	-.256	.200	.175	
.400	-.061	.047	-.072	.057	-.092	.054	-.084	.068	-.194	.078	-.200	.078	-.257	.400	.350	
.450	-.068	.054	-.080	.053	-.085	.062	-.116	.070	-.185	.099	-.225	.089	-.260	.500	.450	
.500	-.090	.053	-.097	.026	-.102	.053	-.111	.054	-.145	.085	-.224	.091	-.261	.600	.550	
.550	-.090	.053	-.097	.026	-.102	.053	-.111	.054	-.145	.085	-.224	.091	-.261	.600	.550	
.600	-.090	.053	-.097	.026	-.102	.053	-.111	.054	-.145	.085	-.224	.091	-.261	.600	.550	
.650	-.079	.006	-.093	.032	-.104	.041	-.111	.061	-.108	.072	-.226	.088	-.259	.057	.700	
.700	-.089	.023	-.095	.024	-.114	.029	-.113	.057	-.104	.090	-.226	.081	-.261	.085	.750	
.750	-.089	.023	-.095	.024	-.114	.029	-.113	.057	-.104	.090	-.226	.081	-.261	.078	.800	
.800	-.088	.024	-.096	.030	-.109	.037	-.124	.053	-.108	.078	-.218	.082	-.259	.900	.850	
.850	-.088	.024	-.096	.030	-.109	.037	-.124	.053	-.108	.078	-.218	.082	-.259	.900	.950	
$\alpha = 6^\circ$																
.0125	-.107	.174	-.194	.138	-.212	.176	-.220	.178	-.221	.145	-.236	.250	-.243	.0125	.025	
.025	-.098	.124	-.194	.117	-.200	.126	-.220	.174	-.220	.134	-.233	.226	-.217	.050	.025	
.050	-.092	.105	-.170	.133	-.209	.173	-.222	.174	-.222	.159	-.236	.191	-.246	.075	.050	
.075	-.054	.103	-.156	.116	-.209	.127	-.222	.159	-.221	.186	-.236	.195	-.246	.100	.075	
.100	-.046	.156	-.107	.204	-.134	.154	-.227	.154	-.223	.186	-.236	.195	-.246	.125	.100	
.150	-.046	.083	-.089	.085	-.196	.134	-.219	.128	-.226	.192	-.242	.203	-.246	.150	.125	
.200	-.065	.076	-.073	.082	-.190	.114	-.219	.121	-.234	.173	-.236	.186	-.249	.207	.200	
.250	-.050	.071	-.062	.090	-.185	.104	-.219	.081	-.232	.157	-.242	.165	-.250	.198	.250	
.300	-.057	.057	-.076	.083	-.166	.104	-.215	.107	-.235	.171	-.243	.119	-.253	.145	.350	
.400	-.069	.079	-.083	.088	-.121	.091	-.210	.100	-.240	.122	-.222	.122	-.254	.400	.450	
.450	-.081	.083	-.093	.088	-.101	.095	-.204	.111	-.239	.146	-.247	.136	-.257	.500	.550	
.500	-.103	.085	-.098	.058	-.110	.087	-.175	.098	-.236	.128	-.251	.140	-.260	.600	.650	
.550	-.087	.028	-.103	.054	-.108	.068	-.102	.100	-.231	.116	-.254	.140	-.259	.098	.700	
.600	-.095	.053	-.106	.049	-.124	.054	-.111	.093	-.234	.129	-.257	.132	-.262	.131	.750	
.650	-.101	.058	-.110	.059	-.116	.062	-.134	.083	-.223	.112	-.253	.122	-.262	.128	.800	
.700	-.090	.053	-.110	.059	-.116	.062	-.134	.083	-.223	.112	-.253	.122	-.262	.128	.850	
.750	-.090	.053	-.110	.059	-.116	.062	-.134	.083	-.223	.112	-.253	.122	-.262	.128	.900	
.800	-.090	.053	-.110	.059	-.116	.062	-.134	.083	-.223	.112	-.253	.122	-.262	.128	.950	
$\alpha = 8^\circ$																
.025							-.230	.214	-.233	.167	-.246	.276	-.261	.250	.025	
.075							-.231	.223	-.231	.266	-.243	.254	-.262	.075	.050	
.100							-.231	.203	-.231	.221	-.246	.224	-.262	.100	.075	
.150							-.233	.209	-.232	.221	-.246	.233	-.261	.150	.125	
.200							-.233	.173	-.234	.232	-.245	.235	-.261	.238	.200	
.250							-.236	.179	-.244	.216	-.251	.227	-.263	.248	.220	
.300							-.243	.129	-.245	.199	-.255	.202	-.263	.241	.250	
.350							-.246	.162	-.249	.208	-.256	.153	-.267	.182	.300	
.400							-.249	.155	-.256	.246	-.268	.160	-.268	.400	.350	
.450							-.248	.151	-.261	.186	-.266	.174	-.271	.202	.500	
.500							-.244	.133	-.266	.168	-.271	.179	-.274	.195	.600	
.550							-.226	.128	-.266	.143	-.274	.174	-.274	.143	.650	
.600							-.206	.123	-.268	.167	-.279	.166	-.277	.170	.800	
.650							-.179	.118	-.271	.143	-.277	.152	-.277	.170	.850	
.700							-.206	.123	-.268	.167	-.279	.166	-.277	.170	.900	
.750							-.179	.118	-.271	.143	-.277	.152	-.277	.170	.950	
$\alpha = 10^\circ$																
.0125	-.211	.238	-.232	.201	-.244	.217	-.243	.243	-.249	.211	-.264	.298	-.268	.278	.0125	
.025	-.217	.200	-.232	.213	-.235	.237	-.243	.253	-.248	.282	-.261	.294	-.271	.075	.050	
.050	-.167	.176	-.232	.203	-.235	.236	-.243	.253	-.248	.282	-.261	.298	-.271	.100	.075	
.075	-.122	.171	-.232	.190	-.249	.207	-.247	.243	-.247	.255	-.261	.269	-.271	.125	.100	
.100	-.092	.171	-.225	.177	-.249	.219	-.251	.247	-.245	.256	-.263	.268	-.271	.150	.125	
.150	-.078	.153	-.232	.158	-.255	.216	-.251	.216	-.248	.271	-.255	.275	-.269	.273	.200	
.200	-.091	.142	-.237	.149	-.260	.194	-.254	.215	-.255	.254	-.262	.269	-.270	.285	.220	
.250	-.080	.142	-.204	.160	-.265	.188	-.261	.162	-.258	.232	-.265	.241	-.270	.281	.250	
.300	-.085	.119	-.145	.151	-.266	.188	-.268	.202	-.262	.246	-.269	.185	-.273	.223	.300	
.400	-.094	.157	-.099	.168	-.267	.180	-.273	.191	-.268	.257	-.273	.194	-.273	.400	.450	
.500	-.104	.156	-.111	.161	-.255	.180	-.273	.193	-.276	.222	-.276	.217	-.276	.500	.550	
.600	-.134	.125	-.119	.109	-.235	.156	-.270	.168	-.280	.200	-.282	.217	-.279	.240	.600	
.650	-.107	.080	-.128	.115	-.266	.126	-.262	.158	-.287	.197	-.287	.214	-.278	.185	.650	
.700	-.107	.104	-.128	.115	-.266	.126	-.262	.158	-.287	.197	-.291	.200	-.281	.229	.750	
.750	-.121	.104	-.128	.115	-.266	.126	-.262	.158	-.287	.197	-.291	.200	-.281	.229	.800	
.800	-.121	.104	-.128	.115	-.266	.126	-.262	.158	-.287	.197	-.291	.200	-.281	.229	.850	
.850	-.123	.116	-.134	.126	-.263	.142	-.263	.155	-.292	.173	-.291	.190	-.281	.208	.900	
.900	-.123	.116	-.134	.126	-.263	.142	-.263	.155	-.292	.173	-.291	.190	-.281	.208	.950	

TABLE VI - Continued

TABLE VI.- Concluded
PRESSURE COEFFICIENTS FOR CAMBERED AND TWISTED DELTA WING
(b) $M = 2.01$ - Concluded

x/c , nominal	C_p at $y/b = 0.5$														x/c , nominal	
	+10		+25		+40		+50		+60		+70		+80			
	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower	Upper	Lower		
$a = 20$																
.0125	-272	.415	-283	.337	-287	.326	-292	.371	-289	.344	-295	.425	-302	.415	.0125	
.025	-280	.409	-283	.401	-277	.408	-287	.428	-292	.442	-286	.491	-292	.454	.025	
.040	-290	.393	-283	.420	-287	.426	-292	.442	-295	.445	-296	.448	-295	.434	.050	
.075	-297	.426	-289	.416	-292	.436	-292	.435	-298	.448	-296	.440	-305	.475	.075	
.110	-313	.404	-301	.401	-296	.436	-292	.435	-292	.472	-298	.480	-295	.454	.100	
.150	-272	.372	-306	.377	-294	.435	-292	.458	-291	.500	-294	.486	-303	.471	.150	
.200	-217	.364	-311	.374	-302	.427	-292	.460	-294	.486	-304	.491	-303	.490	.200	
.250	-198	.347	-306	.372	-312	.437	-300	.383	-294	.468	-305	.464	-303	.490	.250	
.300	-193	.339	-311	.391	-316	.432	-308	.426	-293	.460	-303	.385	-304	.416	.300	
.350	-193	.339	-311	.391	-316	.432	-308	.426	-293	.460	-303	.385	-304	.416	.350	
.400	-193	.389	-302	.400	-318	.425	-316	-	-295	-	-289	.424	-304	.400	.400	
.450	-180	.382	-296	.394	-319	.416	-315	.426	-305	.473	-304	.457	-304	.492	.450	
.500	-180	.382	-296	.394	-319	.416	-315	.426	-305	.473	-304	.457	-304	.492	.500	
.550	-199	.337	-278	.331	-325	.377	-316	.386	-311	.438	-307	.451	-305	.483	.550	
.600	-199	.337	-278	.331	-325	.377	-316	.386	-311	.438	-307	.451	-305	.483	.600	
.650	-155	.280	-265	.333	-320	.344	-318	.394	-311	.408	-312	.459	-305	.407	.650	
.700	-155	.280	-265	.333	-320	.344	-318	.394	-311	.408	-312	.459	-305	.407	.700	
.750	-196	.329	-259	.343	-314	.353	-319	.383	-305	.444	-307	.447	-302	.486	.750	
.800	-196	.329	-259	.343	-314	.353	-319	.383	-305	.444	-307	.447	-302	.486	.800	
.850	-206	.346	-253	.356	-303	.379	-314	.399	-292	.426	-297	.434	-302	.455	.850	
.900	-206	.346	-253	.356	-303	.379	-314	.399	-292	.426	-297	.434	-302	.455	.900	
.950	-206	.346	-253	.356	-303	.379	-314	.399	-292	.426	-297	.434	-302	.455	.950	

TABLE VII
SECTION AERODYNAMIC CHARACTERISTICS FOR FLAT TRAPEZOIDAL WING
(a) $M = 1.61$

α , deg	$y/b/2$						
	.10	.35	.55	.67	.77	.87	.97
	C_n						
-20	-1.003	-0.951	-0.884	-0.844	-0.796	-0.740	-0.650
-18	-0.896	-0.853	-0.796	-0.755	-0.712	-0.658	-0.583
-16	-0.794	-0.758	-0.705	-0.666	-0.624	-0.570	-0.513
-14	-0.695	-0.665	-0.617	-0.580	-0.541	-0.494	-0.448
-12	-0.595	-0.569	-0.528	-0.496	-0.461	-0.417	-0.379
-10	-0.502	-0.479	-0.439	-0.407	-0.378	-0.337	-0.308
-08	-0.402	-0.382	-0.352	-0.329	-0.300	-0.260	-0.235
-06	-0.297	-0.282	-0.256	-0.233	-0.210	-0.179	-0.155
-04	-0.199	-0.190	-0.170	-0.152	-0.134	-0.116	-0.089
-02	-0.093	-0.088	-0.080	-0.072	-0.064	-0.052	-0.032
00	0.000	0.000	0.000	0.000	0.000	0.000	0.000
02	0.093	0.088	0.080	0.072	0.064	0.052	0.032
04	0.199	0.190	0.170	0.152	0.134	0.116	0.089
06	0.297	0.282	0.256	0.233	0.210	0.179	0.155
08	0.402	0.382	0.352	0.329	0.300	0.260	0.235
10	0.502	0.479	0.439	0.407	0.378	0.337	0.308
12	0.595	0.569	0.528	0.496	0.461	0.417	0.379
14	0.695	0.665	0.617	0.580	0.541	0.494	0.448
16	0.794	0.758	0.705	0.666	0.624	0.570	0.513
18	0.896	0.853	0.796	0.755	0.712	0.658	0.583
20	1.003	0.951	0.884	0.844	0.796	0.740	0.650
	C_m						
-20	0.138	0.152	0.165	0.179	0.194	0.194	0.185
-18	0.124	0.134	0.146	0.156	0.169	0.170	0.165
-16	0.110	0.118	0.126	0.133	0.143	0.143	0.144
-14	0.098	0.103	0.108	0.113	0.120	0.122	0.126
-12	0.085	0.088	0.091	0.094	0.100	0.100	0.108
-10	0.074	0.075	0.074	0.074	0.078	0.078	0.089
-08	0.060	0.058	0.058	0.058	0.059	0.057	0.069
-06	0.045	0.042	0.040	0.038	0.038	0.036	0.046
-04	0.030	0.028	0.025	0.023	0.022	0.022	0.025
-02	0.013	0.012	0.012	0.010	0.009	0.008	0.006
00	0.000	0.000	0.000	0.000	0.000	0.000	0.000
02	-0.013	-0.012	-0.012	-0.010	-0.009	-0.008	-0.006
04	-0.030	-0.028	-0.025	-0.023	-0.022	-0.022	-0.025
06	-0.045	-0.042	-0.040	-0.038	-0.038	-0.036	-0.046
08	-0.060	-0.058	-0.058	-0.058	-0.059	-0.057	-0.069
10	-0.074	-0.075	-0.074	-0.074	-0.078	-0.078	-0.089
12	-0.085	-0.088	-0.091	-0.094	-0.100	-0.100	-0.108
14	-0.098	-0.103	-0.108	-0.113	-0.120	-0.122	-0.126
16	-0.110	-0.118	-0.126	-0.133	-0.143	-0.143	-0.144
18	-0.124	-0.134	-0.146	-0.156	-0.169	-0.170	-0.165
20	-0.138	-0.152	-0.165	-0.179	-0.194	-0.194	-0.185

TABLE VII.- Concluded
 SECTION AERODYNAMIC CHARACTERISTICS FOR FLAT TRAPEZOIDAL WING
 (b) $M = 2.01$

α , deg	$y/\frac{b}{2}$						
	.10	.35	.55	.67	.77	.87	.97
C_n							
-20	-.826	-.788	-.742	-.701	-.659	-.600	-.509
-18	-.756	-.702	-.660	-.619	-.581	-.540	-.463
-16	-.650	-.627	-.589	-.553	-.518	-.474	-.407
-14	-.563	-.546	-.511	-.480	-.446	-.401	-.344
-12	-.478	-.465	-.436	-.407	-.378	-.341	-.296
-10	-.389	-.381	-.357	-.332	-.305	-.273	-.228
-08	-.318	-.314	-.290	-.269	-.247	-.213	-.177
-06	-.229	-.228	-.213	-.196	-.178	-.149	-.114
-04	-.150	-.148	-.141	-.128	-.113	-.092	-.062
-02	-.072	-.073	-.070	-.063	-.055	-.045	-.025
00	.000	.000	.000	.000	.000	.000	.000
02	.072	.073	.070	.063	.055	.045	.025
04	.150	.148	.141	.128	.113	.092	.062
06	.229	.228	.213	.196	.178	.149	.114
08	.318	.314	.290	.269	.247	.213	.177
10	.389	.381	.357	.332	.305	.273	.228
12	.478	.465	.436	.407	.378	.341	.296
14	.563	.546	.511	.480	.446	.401	.344
16	.650	.627	.589	.553	.518	.474	.407
18	.756	.702	.660	.619	.581	.540	.463
20	.826	.788	.742	.701	.659	.600	.509
C_m							
-20	.126	.139	.149	.152	.157	.151	.139
-18	.107	.124	.130	.132	.136	.135	.126
-16	.101	.111	.115	.117	.120	.116	.111
-14	.088	.096	.099	.100	.101	.096	.093
-12	.075	.082	.085	.083	.085	.081	.082
-10	.061	.067	.068	.067	.067	.063	.064
-08	.049	.056	.055	.054	.053	.048	.051
-06	.035	.040	.040	.038	.036	.031	.032
-04	.022	.026	.027	.024	.022	.018	.016
-02	.010	.013	.013	.012	.010	.008	.004
00	.000	.000	.000	.000	.000	.000	.000
02	-.010	-.013	-.013	-.012	-.010	-.008	-.004
04	-.022	-.026	-.027	-.024	-.022	-.018	-.016
06	-.035	-.040	-.040	-.038	-.036	-.031	-.032
08	-.049	-.056	-.055	-.054	-.053	-.048	-.051
10	-.061	-.067	-.068	-.067	-.067	-.063	-.064
12	-.075	-.082	-.085	-.083	-.085	-.081	-.082
14	-.088	-.096	-.099	-.100	-.101	-.096	-.093
16	-.101	-.111	-.115	-.117	-.120	-.116	-.111
18	-.107	-.124	-.130	-.132	-.136	-.135	-.126
20	-.126	-.139	-.149	-.152	-.157	-.151	-.139

TABLE VIII
SECTION AERODYNAMIC CHARACTERISTICS FOR WARPED TRAPEZOIDAL WING

(a) M = 1.61

α , deg	$y/\frac{b}{2}$						
	.10	.35	.55	.67	.77	.87	.97
	C_n						
-20	-.953	-.915	-.893	-.869	-.855	-.826	-.712
-18	-.818	-.787	-.768	-.751	-.745	-.725	-.638
-16	-.713	-.685	-.673	-.654	-.653	-.641	-.576
-14	-.616	-.595	-.586	-.569	-.572	-.560	-.512
-12	-.518	-.501	-.492	-.485	-.483	-.477	-.444
-10	-.421	-.407	-.405	-.387	-.399	-.396	-.372
-08	-.319	-.308	-.315	-.305	-.313	-.309	-.299
-06	-.200	-.197	-.204	-.208	-.211	-.206	-.203
-04	-.111	-.113	-.129	-.131	-.141	-.136	-.135
-02	-.009	-.015	-.037	-.051	-.058	-.060	-.054
00	.093	.081	.056	.028	.014	.004	.000
02	.204	.186	.147	.109	.086	.063	.034
04	.302	.280	.227	.189	.149	.112	.071
06	.397	.370	.310	.257	.214	.167	.114
08	.508	.475	.406	.349	.296	.242	.177
10	.599	.558	.485	.424	.364	.303	.239
12	.701	.649	.573	.512	.446	.376	.314
14	.785	.734	.648	.577	.511	.436	.379
16	.885	.826	.734	.663	.592	.512	.449
18	.993	.922	.824	.757	.672	.591	.520
20	1.107	1.027	.925	.844	.763	.678	.595
	C_m						
-20	.153	.185	.198	.201	.218	.218	.196
-18	.129	.159	.167	.168	.183	.187	.173
-16	.109	.136	.141	.139	.152	.159	.152
-14	.093	.118	.121	.117	.127	.134	.132
-12	.077	.102	.100	.096	.102	.109	.113
-10	.064	.084	.081	.067	.079	.085	.092
-08	.049	.066	.063	.054	.056	.059	.072
-06	.031	.048	.042	.034	.031	.030	.045
-04	.017	.034	.028	.017	.014	.011	.025
-02	.003	.018	.012	.004	-.004	-.007	.000
00	-.012	.004	-.002	-.010	-.019	-.021	-.015
02	-.028	-.011	-.016	-.022	-.030	-.030	-.021
04	-.042	-.025	-.027	-.035	-.040	-.037	-.032
06	-.054	-.038	-.039	-.044	-.050	-.048	-.045
08	-.067	-.054	-.054	-.060	-.066	-.064	-.062
10	-.078	-.065	-.069	-.075	-.081	-.077	-.080
12	-.088	-.077	-.083	-.092	-.099	-.094	-.099
14	-.096	-.088	-.096	-.104	-.114	-.108	-.115
16	-.105	-.100	-.111	-.123	-.134	-.127	-.132
18	-.119	-.117	-.130	-.148	-.156	-.150	-.151
20	-.133	-.137	-.155	-.169	-.182	-.176	-.173

TABLE VIII.- Concluded
 SECTION AERODYNAMIC CHARACTERISTICS FOR WARPED TRAPEZOIDAL WING
 (b) M = 2.01

α , deg	$y/b/2$						
	.10	.35	.55	.67	.77	.87	.97
C_n							
-20	-.746	-.712	-.709	-.697	-.697	-.667	-.568
-18	-.666	-.646	-.640	-.632	-.639	-.619	-.532
-16	-.574	-.562	-.561	-.559	-.567	-.552	-.482
-14	-.494	-.481	-.485	-.485	-.490	-.485	-.433
-12	-.407	-.403	-.415	-.409	-.421	-.418	-.379
-10	-.324	-.324	-.332	-.349	-.346	-.344	-.321
-08	-.241	-.243	-.260	-.263	-.274	-.270	-.260
-06	-.160	-.165	-.185	-.191	-.201	-.198	-.190
-04	-.079	-.088	-.113	-.124	-.132	-.135	-.133
-02	-.002	-.011	-.044	-.047	-.068	-.070	-.069
00	.072	.064	.030	.016	-.004	-.016	-.018
02	.158	.149	.102	.077	.052	.027	.013
04	.236	.224	.183	.141	.106	.070	.041
06	.316	.304	.262	.205	.157	.119	.067
08	.399	.384	.324	.273	.218	.166	.106
10	.481	.463	.391	.345	.275	.220	.156
12	.565	.543	.471	.410	.346	.281	.215
14	.649	.625	.541	.481	.415	.343	.275
16	.741	.710	.625	.553	.480	.406	.342
18	.826	.786	.696	.616	.543	.468	.396
20	.911	.864	.763	.685	.602	.531	.448
C_m							
-20	.110	.145	.157	.158	.169	.165	.147
-18	.097	.133	.141	.141	.154	.153	.138
-16	.084	.117	.124	.123	.135	.135	.124
-14	.071	.102	.107	.106	.113	.116	.111
-12	.058	.088	.093	.088	.095	.098	.097
-10	.044	.073	.075	.073	.074	.077	.082
-08	.030	.057	.061	.054	.056	.056	.066
-06	.018	.043	.045	.037	.036	.035	.046
-04	.005	.029	.030	.022	.018	.017	.029
-02	-.005	.015	.017	.004	.002	.000	.008
00	-.016	.002	.002	-.007	-.011	-.012	-.007
02	-.030	-.014	-.011	-.018	-.022	-.020	-.014
04	-.042	-.027	-.027	-.030	-.032	-.027	-.020
06	-.055	-.041	-.038	-.042	-.041	-.036	-.026
08	-.068	-.055	-.052	-.055	-.053	-.046	-.038
10	-.080	-.069	-.065	-.071	-.064	-.057	-.051
12	-.094	-.082	-.080	-.082	-.081	-.071	-.067
14	-.106	-.096	-.093	-.096	-.095	-.085	-.082
16	-.119	-.110	-.109	-.112	-.110	-.100	-.099
18	-.128	-.122	-.123	-.124	-.124	-.115	-.113
20	-.136	-.133	-.135	-.139	-.138	-.132	-.127

TABLE IX
SECTION AERODYNAMIC CHARACTERISTICS FOR FLAT DELTA WING
(a) $M = 1.61$

α , deg	$y/b/2$						
	.10	.25	.40	.50	.60	.70	.80
C_n							
-20	-.863	-.840	-.753	-.642	-.518	-.390	-.263
-18	-.737	-.726	-.665	-.576	-.462	-.349	-.234
-16	-.614	-.614	-.586	-.506	-.407	-.306	-.204
-14	-.517	-.522	-.512	-.461	-.376	-.270	-.181
-12	-.425	-.434	-.428	-.403	-.340	-.244	-.162
-10	-.353	-.339	-.340	-.338	-.311	-.222	-.147
-08	-.267	-.261	-.257	-.255	-.244	-.200	-.131
-06	-.192	-.190	-.179	-.176	-.171	-.166	-.114
-04	-.125	-.125	-.111	-.106	-.102	-.099	-.089
-02	-.054	-.054	-.053	-.051	-.049	-.043	-.036
00	.000	.000	.000	.000	.000	.000	.000
02	.054	.054	.053	.051	.049	.043	.036
04	.125	.125	.111	.106	.102	.099	.089
06	.192	.190	.179	.176	.171	.166	.114
08	.267	.261	.257	.255	.244	.200	.131
10	.353	.339	.340	.338	.311	.222	.147
12	.425	.434	.428	.403	.340	.244	.162
14	.517	.522	.512	.461	.376	.270	.181
16	.614	.614	.586	.506	.407	.306	.204
18	.737	.726	.665	.576	.462	.349	.234
20	.863	.840	.753	.642	.518	.390	.263
C_m							
-20	.042	.136	.232	.250	.239	.208	.159
-18	.027	.104	.197	.222	.212	.185	.142
-16	.013	.074	.164	.190	.184	.160	.123
-14	.010	.054	.134	.172	.169	.141	.108
-12	.007	.037	.101	.146	.152	.127	.097
-10	.007	.031	.071	.112	.139	.115	.088
-08	.010	.025	.050	.076	.102	.104	.078
-06	.007	.019	.035	.049	.066	.085	.068
-04	.003	.012	.022	.030	.037	.047	.052
-02	-.000	.004	.011	.015	.019	.020	.020
00	.000	.000	.000	.000	.000	.000	.000
02	.000	-.004	-.011	-.015	-.019	-.020	-.020
04	-.003	-.012	-.022	-.030	-.037	-.047	-.052
06	-.007	-.019	-.035	-.049	-.066	-.085	-.068
08	-.010	-.025	-.050	-.076	-.102	-.104	-.078
10	-.007	-.031	-.071	-.112	-.139	-.115	-.088
12	-.007	-.037	-.101	-.146	-.152	-.127	-.097
14	-.010	-.054	-.134	-.172	-.169	-.141	-.108
16	-.013	-.074	-.164	-.190	-.184	-.160	-.123
18	-.027	-.104	-.197	-.222	-.212	-.185	-.142
20	-.042	-.136	-.232	-.250	-.239	-.208	-.159

TABLE IX.- Concluded
SECTION AERODYNAMIC CHARACTERISTICS FOR FLAT DELTA WING
(b) $M = 2.01$

α , deg	$y/b/2$						
	.10	.25	.40	.50	.60	.70	.80
	C_n						
-20	-.689	-.650	-.544	-.444	-.356	-.268	-.179
-18	-.597	-.579	-.508	-.419	-.338	-.250	-.169
-16	-.516	-.509	-.460	-.384	-.310	-.233	-.155
-14	-.428	-.426	-.403	-.347	-.281	-.211	-.143
-12	-.359	-.360	-.349	-.314	-.260	-.191	-.131
-10	-.286	-.292	-.285	-.267	-.228	-.168	-.117
-08	-.218	-.218	-.212	-.200	-.190	-.153	-.106
-06	-.160	-.164	-.155	-.145	-.142	-.124	-.090
-04	-.106	-.104	-.102	-.096	-.088	-.077	-.065
-02	-.053	-.049	-.046	-.043	-.040	-.051	-.038
00	.000	.000	.000	.000	.000	.000	.000
02	.053	.049	.046	.043	.040	.051	.038
04	.106	.104	.102	.096	.088	.077	.065
06	.160	.164	.155	.145	.142	.124	.090
08	.218	.218	.212	.200	.190	.153	.106
10	.286	.292	.285	.267	.228	.168	.117
12	.359	.360	.349	.314	.260	.191	.131
14	.428	.426	.403	.347	.281	.211	.143
16	.516	.509	.460	.384	.310	.233	.155
18	.597	.579	.508	.419	.338	.250	.169
20	.689	.650	.544	.444	.356	.268	.179
	C_m						
-20	.013	.096	.153	.161	.158	.139	.107
-18	.007	.079	.142	.152	.149	.129	.101
-16	.004	.063	.126	.138	.137	.120	.092
-14	.002	.046	.106	.124	.124	.109	.085
-12	.002	.035	.086	.111	.114	.098	.077
-10	.001	.027	.064	.091	.100	.086	.069
-08	.001	.020	.043	.062	.082	.078	.063
-06	.001	.015	.032	.042	.059	.062	.053
-04	.000	.010	.022	.029	.035	.037	.038
-02	.000	.004	.010	.013	.015	.025	.022
00	.000	.000	.000	.000	.000	.000	.000
02	-.000	-.004	-.010	-.013	-.015	-.025	-.022
04	-.000	-.010	-.022	-.029	-.035	-.037	-.038
06	-.001	-.015	-.032	-.042	-.059	-.062	-.053
08	-.001	-.020	-.043	-.062	-.082	-.078	-.063
10	-.001	-.027	-.064	-.091	-.100	-.086	-.069
12	-.002	-.035	-.086	-.111	-.114	-.098	-.077
14	-.002	-.046	-.106	-.124	-.124	-.109	-.085
16	-.004	-.063	-.126	-.138	-.137	-.120	-.092
18	-.007	-.079	-.142	-.152	-.149	-.129	-.101
20	-.013	-.096	-.153	-.161	-.158	-.139	-.107

TABLE X
SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED DELTA WING

(a) $M = 1.61$

α , deg	$y/b/2$						
	.10	.25	.40	.50	.60	.70	.80
C_n							
-20	-.823	-.780	-.695	-.580	-.457	-.344	-.223
-16	-.616	-.592	-.561	-.474	-.365	-.270	-.169
-18	-.701	-.674	-.616	-.519	-.404	-.299	-.188
-14	-.514	-.492	-.479	-.421	-.328	-.239	-.152
-12	-.414	-.391	-.393	-.375	-.292	-.209	-.134
-10	-.325	-.302	-.304	-.307	-.259	-.181	-.117
-08	-.237	-.218	-.226	-.219	-.215	-.153	-.098
-06	-.163	-.146	-.144	-.139	-.142	-.120	-.077
-04	-.095	-.075	-.072	-.064	-.066	-.065	-.047
-02	-.025	-.013	-.007	-.002	-.004	-.007	-.005
00	.041	.049	.052	.050	.043	.035	.031
02	.106	.112	.107	.102	.095	.081	.068
04	.174	.174	.168	.161	.147	.132	.122
06	.238	.240	.233	.230	.214	.204	.144
08	.320	.313	.310	.297	.297	.239	.158
10	.399	.391	.389	.384	.352	.262	.172
12	.480	.475	.472	.461	.388	.283	.186
14	.573	.568	.561	.519	.421	.309	.203
16	.679	.678	.648	.574	.460	.343	.226
18	.797	.797	.736	.646	.520	.393	.262
20	.927	.904	.820	.711	.574	.431	.291
C_m							
-20	-.006	.107	.204	.220	.208	.182	.135
-16	-.031	.051	.151	.174	.161	.141	.101
-18	-.023	.074	.172	.193	.181	.157	.113
-14	-.035	.027	.119	.153	.143	.124	.090
-12	-.035	.009	.084	.133	.128	.108	.079
-10	-.034	-.001	.051	.101	.114	.093	.070
-08	-.033	-.007	.029	.059	.093	.079	.058
-06	-.031	-.011	.011	.028	.054	.061	.045
-04	-.031	-.013	-.001	.006	.018	.029	.027
-02	-.030	-.019	-.012	-.009	-.003	.001	.001
00	-.031	-.024	-.025	-.024	-.022	-.019	-.019
02	-.032	-.032	-.037	-.040	-.042	-.040	-.039
04	-.035	-.039	-.051	-.057	-.060	-.064	-.072
06	-.038	-.047	-.063	-.075	-.086	-.106	-.086
08	-.042	-.054	-.077	-.096	-.128	-.126	-.095
10	-.043	-.061	-.095	-.131	-.158	-.138	-.103
12	-.048	-.069	-.120	-.167	-.176	-.149	-.111
14	-.053	-.082	-.154	-.195	-.192	-.163	-.122
16	-.059	-.104	-.189	-.219	-.210	-.181	-.136
18	-.073	-.139	-.229	-.254	-.242	-.211	-.159
20	-.087	-.168	-.262	-.281	-.267	-.231	-.176

TABLE X.- Concluded
 SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED DELTA WING
 (b) M = 2.01

α , deg	y/b						
	.10	.25	.40	.50	.60	.70	.80
	C_n						
-20	-0.671	-0.588	-0.476	-0.387	-0.305	-0.226	-0.147
-18	-0.612	-0.557	-0.466	-0.378	-0.299	-0.222	-0.144
-16	-0.530	-0.503	-0.440	-0.354	-0.279	-0.208	-0.135
-14	-0.449	-0.427	-0.399	-0.328	-0.255	-0.191	-0.126
-12	-0.367	-0.363	-0.347	-0.299	-0.226	-0.171	-0.113
-10	-0.287	-0.276	-0.272	-0.260	-0.200	-0.149	-0.098
-08	-0.217	-0.205	-0.202	-0.201	-0.171	-0.128	-0.083
-06	-0.146	-0.135	-0.140	-0.124	-0.126	-0.097	-0.065
-04	-0.087	-0.079	-0.073	-0.064	-0.063	-0.057	-0.042
-02	-0.028	-0.016	-0.012	-0.005	-0.007	-0.009	-0.008
00	0.029	0.036	0.039	0.041	0.033	0.028	0.022
02	0.090	0.090	0.089	0.086	0.078	0.065	0.055
04	0.152	0.148	0.143	0.138	0.126	0.109	0.088
06	0.218	0.206	0.203	0.193	0.180	0.153	0.109
08	0.274	0.267	0.254	0.250	0.231	0.179	0.124
10	0.344	0.331	0.321	0.314	0.266	0.200	0.138
12	0.410	0.400	0.392	0.364	0.295	0.220	0.148
14	0.486	0.479	0.455	0.403	0.324	0.239	0.160
16	0.536	0.562	0.515	0.441	0.350	0.260	0.172
18	0.665	0.636	0.554	0.477	0.379	0.279	0.185
20	0.763	0.718	0.605	0.519	0.408	0.298	0.196
	C_m						
-20	-0.016	0.077	0.127	0.135	0.131	0.116	0.087
-18	-0.023	0.067	0.124	0.132	0.129	0.114	0.086
-16	-0.029	0.051	0.117	0.123	0.120	0.107	0.080
-14	-0.031	0.031	0.104	0.114	0.110	0.098	0.075
-12	-0.033	0.016	0.084	0.105	0.097	0.088	0.067
-10	-0.032	0.002	0.055	0.090	0.086	0.076	0.058
-08	-0.032	-0.004	0.030	0.062	0.073	0.065	0.049
-06	-0.030	-0.008	0.016	0.029	0.051	0.049	0.038
-04	-0.027	-0.010	0.002	0.010	0.020	0.026	0.023
-02	-0.025	-0.015	-0.008	-0.006	-0.001	0.002	0.004
00	-0.026	-0.021	-0.019	-0.019	-0.017	-0.015	-0.013
02	-0.027	-0.026	-0.030	-0.033	-0.034	-0.032	-0.032
04	-0.027	-0.034	-0.042	-0.049	-0.052	-0.054	-0.052
06	-0.029	-0.039	-0.055	-0.065	-0.075	-0.078	-0.064
08	-0.029	-0.045	-0.064	-0.083	-0.100	-0.092	-0.074
10	-0.031	-0.051	-0.081	-0.110	-0.117	-0.104	-0.082
12	-0.032	-0.059	-0.104	-0.132	-0.131	-0.114	-0.088
14	-0.037	-0.071	-0.127	-0.148	-0.144	-0.124	-0.096
16	-0.014	-0.087	-0.148	-0.162	-0.156	-0.135	-0.103
18	-0.043	-0.103	-0.160	-0.177	-0.169	-0.146	-0.111
20	-0.047	-0.122	-0.178	-0.193	-0.182	-0.155	-0.118

TABLE XI

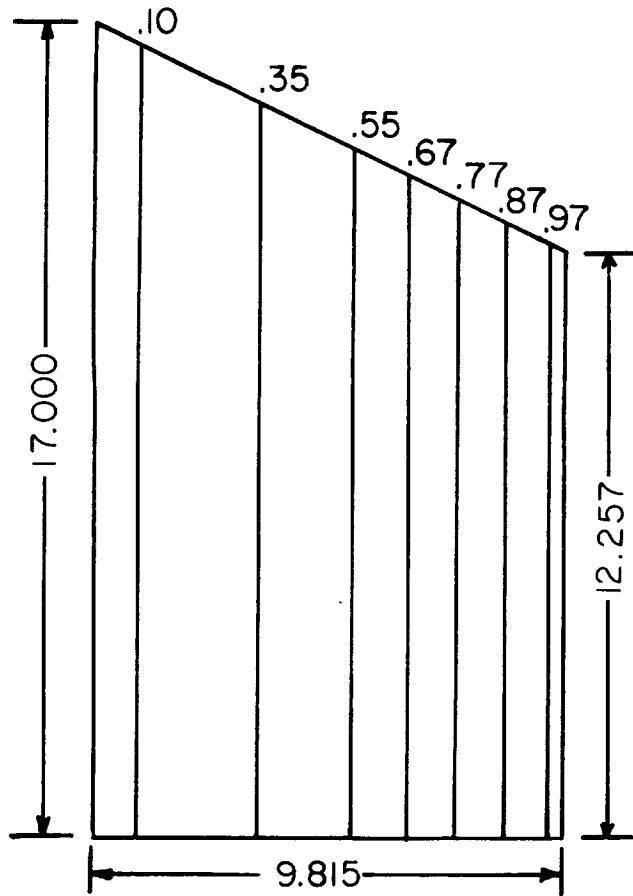
SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED AND TWISTED DELTA WING

(a) $M = 1.61$

α , deg	$y/b/2$						
	.10	.25	.40	.50	.60	.70	.80
C_n							
-20	-.776	-.732	-.640	-.536	-.415	-.305	-.272
-18	-.675	-.648	-.577	-.490	-.374	-.273	-.248
-16	-.562	-.549	-.502	-.439	-.331	-.241	-.225
-14	-.458	-.456	-.424	-.390	-.293	-.213	-.204
-12	-.370	-.374	-.343	-.333	-.270	-.186	-.184
-10	-.291	-.287	-.265	-.249	-.235	-.163	-.162
-08	-.217	-.212	-.186	-.176	-.170	-.131	-.135
-06	-.144	-.142	-.117	-.108	-.085	-.078	-.044
-04	-.076	-.076	-.055	-.044	-.022	-.023	-.044
-02	-.017	-.018	.005	.011	.022	.020	.007
00	.047	.041	.058	.062	.072	.065	.056
02	.109	.104	.122	.123	.130	.124	.126
04	.180	.176	.182	.194	.196	.202	.150
06	.256	.245	.254	.268	.279	.240	.169
08	.327	.320	.340	.355	.348	.265	.188
10	.410	.411	.434	.442	.385	.288	.208
12	.496	.504	.525	.498	.417	.315	.230
14	.606	.622	.625	.552	.457	.350	.259
16	.711	.732	.698	.615	.511	.397	.302
18	.847	.864	.798	.692	.571	.444	.341
20	.977	.980	.880	.761	.631	.487	.386
C_m							
-20	.014	.097	.186	.202	.188	.163	.133
-18	.001	.072	.161	.182	.168	.144	.118
-16	-.005	.049	.130	.161	.147	.126	.105
-14	-.007	.031	.098	.138	.128	.111	.094
-12	-.007	.020	.068	.111	.120	.096	.084
-10	-.008	.013	.046	.070	.102	.084	.074
-08	-.010	.008	.028	.042	.068	.067	.061
-06	-.009	.003	.016	.022	.027	.037	.043
-04	-.014	-.003	.005	.006	.004	.008	.020
-02	-.014	-.008	-.007	-.010	-.012	-.011	-.000
00	-.013	-.014	-.018	-.025	-.031	-.032	-.021
02	-.015	-.022	-.032	-.042	-.053	-.060	-.060
04	-.017	-.029	-.044	-.062	-.078	-.105	-.073
06	-.019	-.036	-.058	-.085	-.121	-.127	-.082
08	-.022	-.042	-.078	-.120	-.158	-.140	-.090
10	-.023	-.050	-.106	-.162	-.176	-.152	-.101
12	-.025	-.063	-.140	-.188	-.190	-.166	-.112
14	-.030	-.087	-.181	-.210	-.208	-.186	-.127
16	-.039	-.115	-.211	-.239	-.238	-.215	-.151
18	-.065	-.164	-.257	-.277	-.269	-.240	-.171
20	-.081	-.195	-.285	-.304	-.295	-.262	-.188

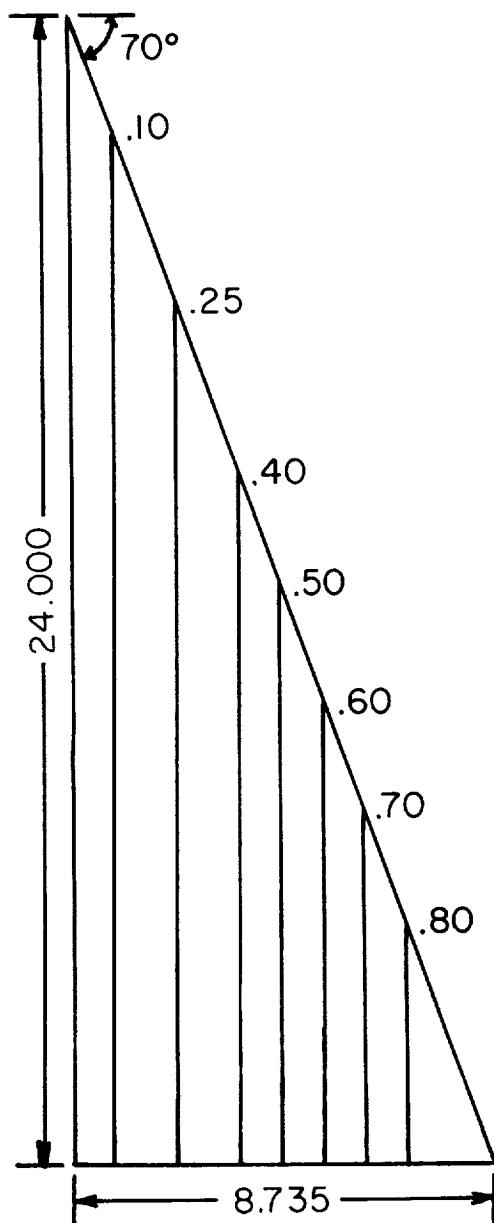
TABLE XI.- Concluded
SECTION AERODYNAMIC CHARACTERISTICS FOR CAMBERED AND TWISTED DELTA WING
(b) $M = 2.01$

α , deg	$y/b/2$						
	.10	.25	.40	.50	.60	.70	.80
C_n							
-20	-.664	-.602	-.486	-.380	-.294	-.211	-.135
-18	-.583	-.545	-.450	-.360	-.277	-.202	-.130
-16	-.494	-.476	-.419	-.335	-.257	-.187	-.121
-14	-.415	-.403	-.378	-.312	-.234	-.169	-.111
-12	-.332	-.327	-.296	-.275	-.209	-.149	-.098
-10	-.257	-.262	-.244	-.230	-.187	-.132	-.086
-08	-.196	-.199	-.185	-.163	-.152	-.108	-.071
-06	-.136	-.142	-.125	-.101	-.087	-.066	-.052
-04	-.081	-.080	-.067	-.045	-.033	-.023	-.015
-02	-.026	-.026	-.004	-.003	.012	.016	.016
00	.029	.026	.049	.047	.056	.051	.049
02	.086	.085	.094	.102	.107	.097	.089
04	.148	.145	.155	.153	.162	.146	.111
06	.208	.201	.208	.213	.226	.176	.123
08				.282	.262	.200	.140
10	.334	.332	.354	.334	.292	.223	.154
12	.409	.415	.419	.381	.325	.245	.169
14	.483	.493	.482	.421	.358	.263	.183
16	.572	.580	.535	.458	.386	.291	.199
18	.670	.658	.581	.501	.415	.313	.214
20	.761	.731	.635	.510	.448	.335	.229
C_m							
-20	-.000	.081	.133	.133	.128	.109	.080
-18	-.006	.066	.123	.126	.120	.104	.077
-16	-.010	.048	.113	.117	.111	.096	.071
-14	-.011	.032	.096	.109	.101	.086	.065
-12	-.011	.021	.064	.095	.090	.076	.058
-10	-.010	.013	.046	.073	.081	.067	.051
-08	-.010	.008	.031	.043	.064	.054	.041
-06	-.009	.005	.020	.022	.030	.031	.029
-04	-.009	.000	.009	.007	.010	.010	.008
-02	-.009	-.004	-.004	-.006	-.007	-.009	-.009
00	-.010	-.010	-.015	-.019	-.024	-.025	-.028
02	-.009	-.017	-.025	-.036	-.044	-.047	-.052
04	-.010	-.023	-.038	-.050	-.065	-.075	-.065
06	-.012	-.029	-.048	-.070	-.099	-.091	-.073
08				-.098	-.116	-.104	-.083
10	-.012	-.042	-.090	-.121	-.130	-.116	-.092
12	-.014	-.054	-.115	-.140	-.145	-.128	-.101
14	-.016	-.068	-.137	-.156	-.160	-.137	-.109
16	-.020	-.089	-.154	-.170	-.173	-.152	-.119
18	-.024	-.109	-.169	-.187	-.186	-.164	-.128
20	-.031	-.125	-.186	-.192	-.200	-.176	-.137



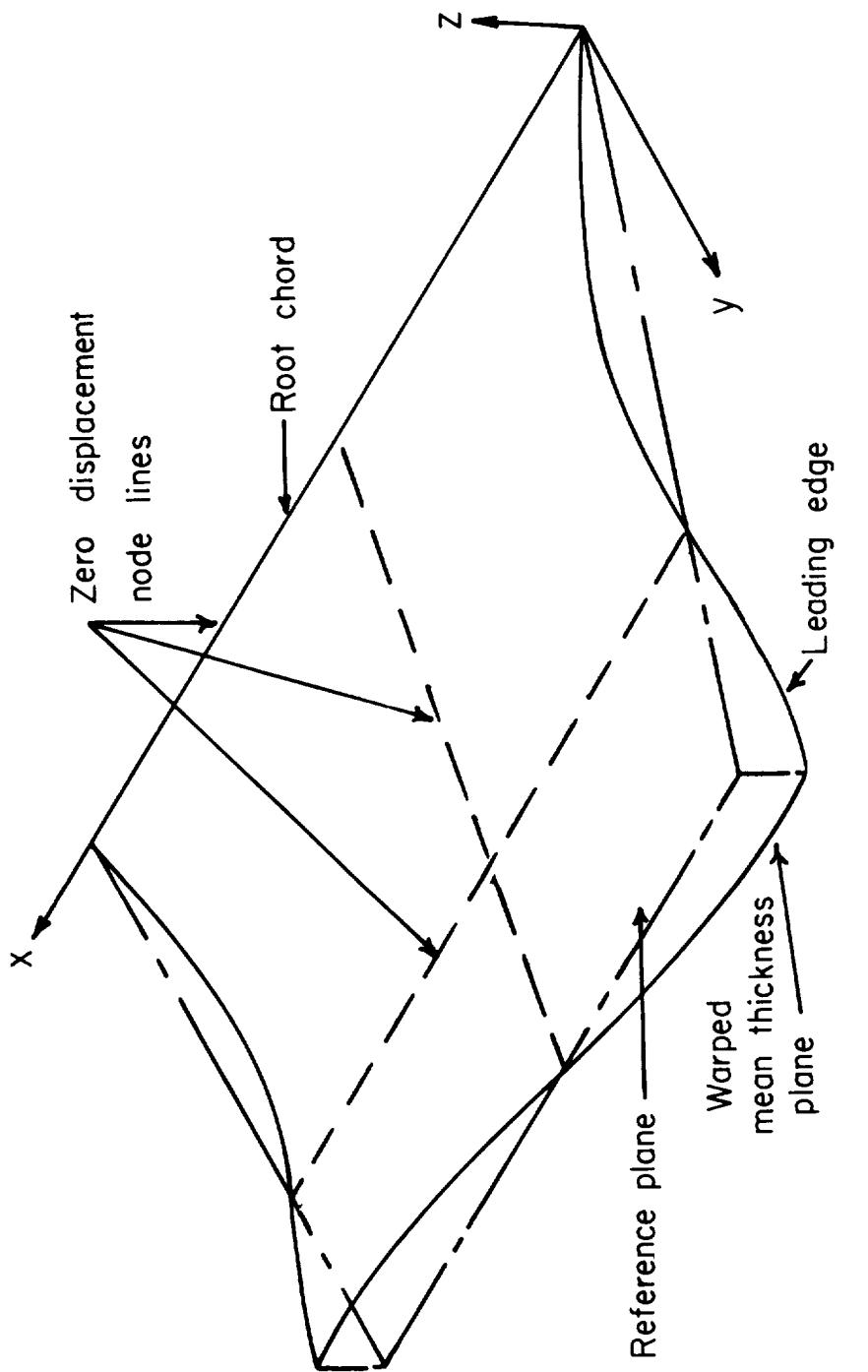
(a) Trapezoidal wings.

Figure 1.- Plan view of the wings. (Lengths are in inches; stations are in fractions of semispan.)



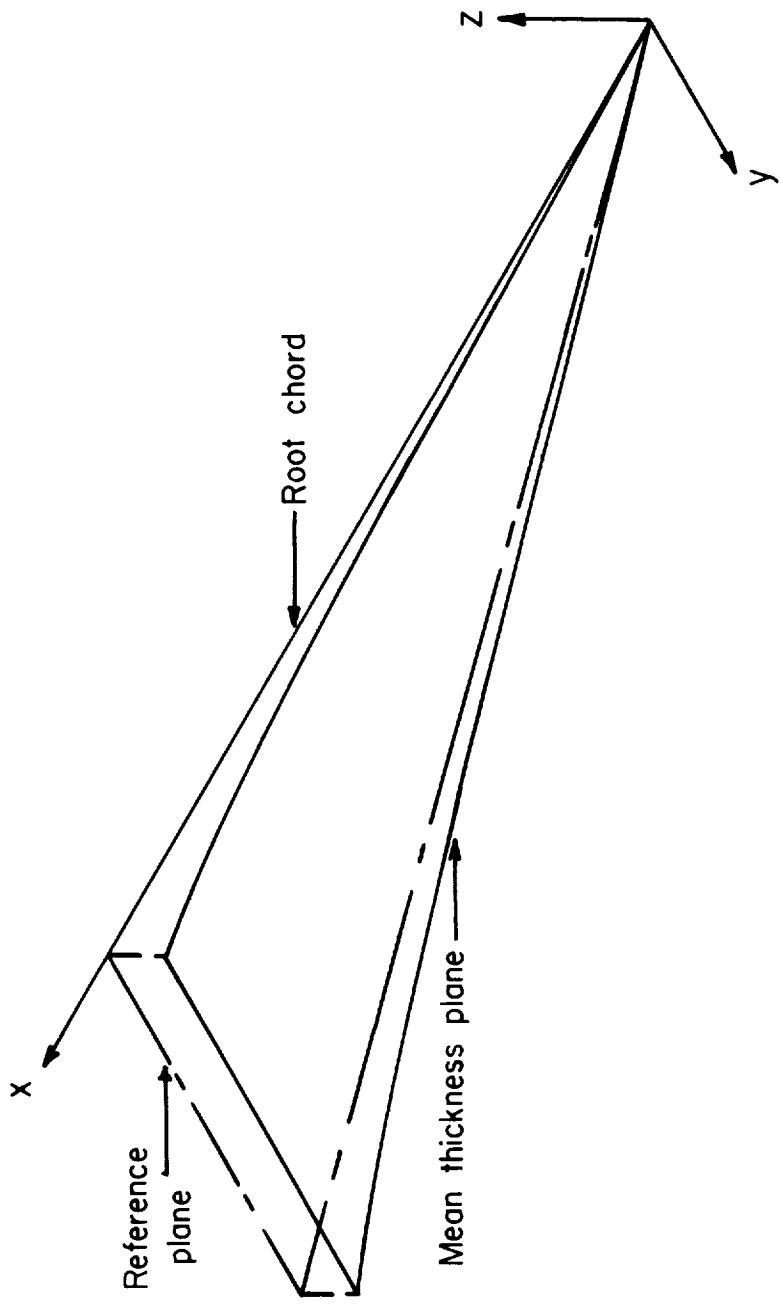
(b) Delta wings.

Figure 1.- Concluded.



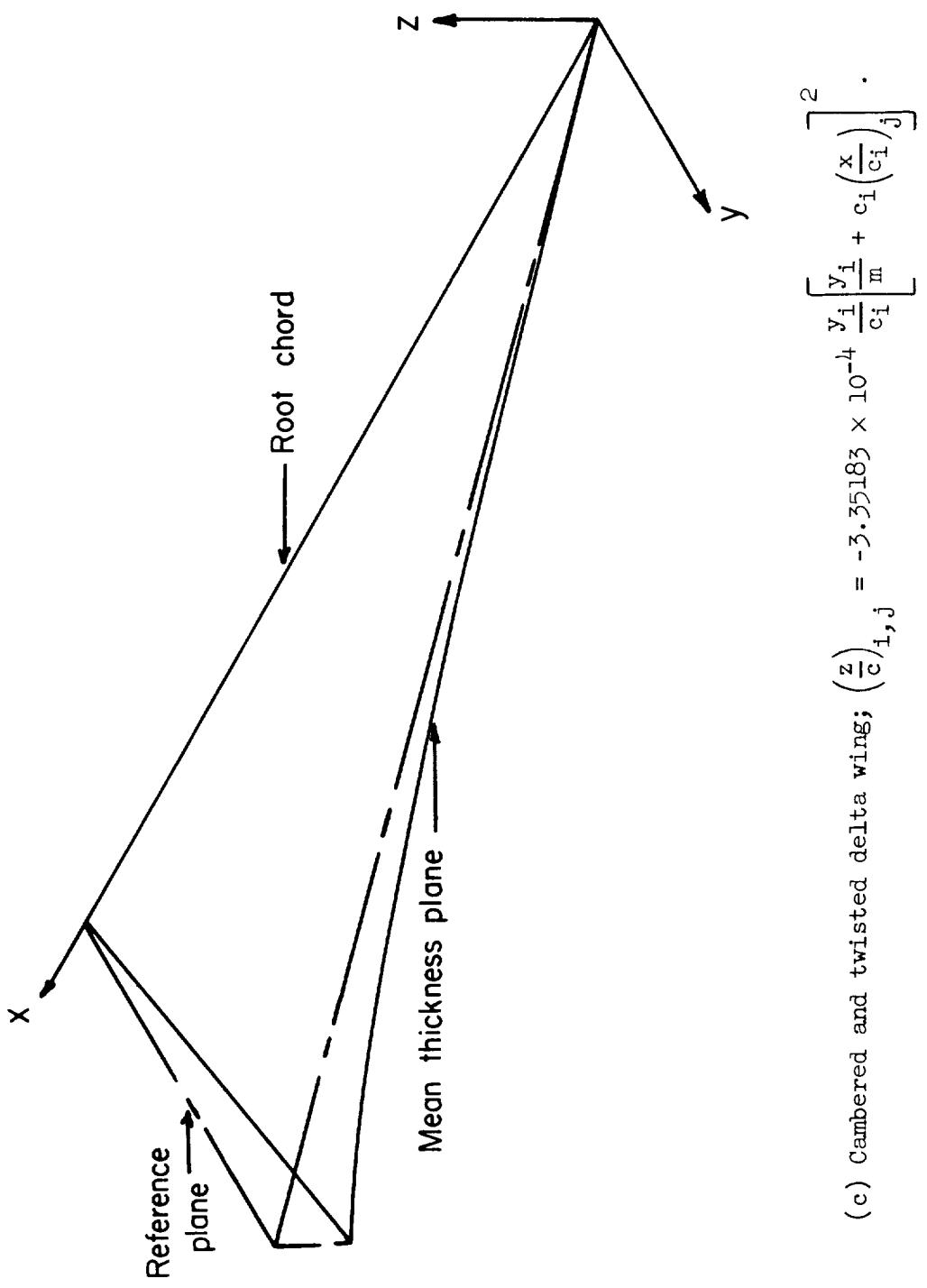
$$(a) \text{ Warped trapezoidal wing; } \left(\frac{z}{c}\right)_{1,j} = 0.033454 \sin\left(\frac{3\pi}{2} \frac{y_1}{b/2}\right) \cos\left[\pi \left(\frac{x}{c_1}\right)_j\right].$$

Figure 2.- Sketches of mean thickness plane.



$$(b) \text{ Cambered delta wing; } \left(\frac{z}{c}\right)_{i,j} = \frac{-2.18958 \times 10^{-3}}{c_i} \left[\frac{y_i}{m} + c_i \left(\frac{x}{c_i} \right)^j \right]^2.$$

Figure 2.- Continued.



$$(c) \text{ Cambered and twisted delta wing; } \left(\frac{z}{c}\right)_{1,j} = -3.35183 \times 10^{-4} \left[\frac{y_1}{c_i} \left(\frac{y_1}{m} + c_i \left(\frac{x}{c_i} \right)^j \right)^2 \right].$$

Figure 2.- Concluded.

L-61-8496

Figure 3.- Semispan wing model mounted in tunnel on boundary-layer bypass plate. (Cambered and twisted delta wing shown at positive angle of attack.)





